

Figure 15-5. Cessna Nav/Omni 800 (190 Channel) (Sheet 2 of 2)

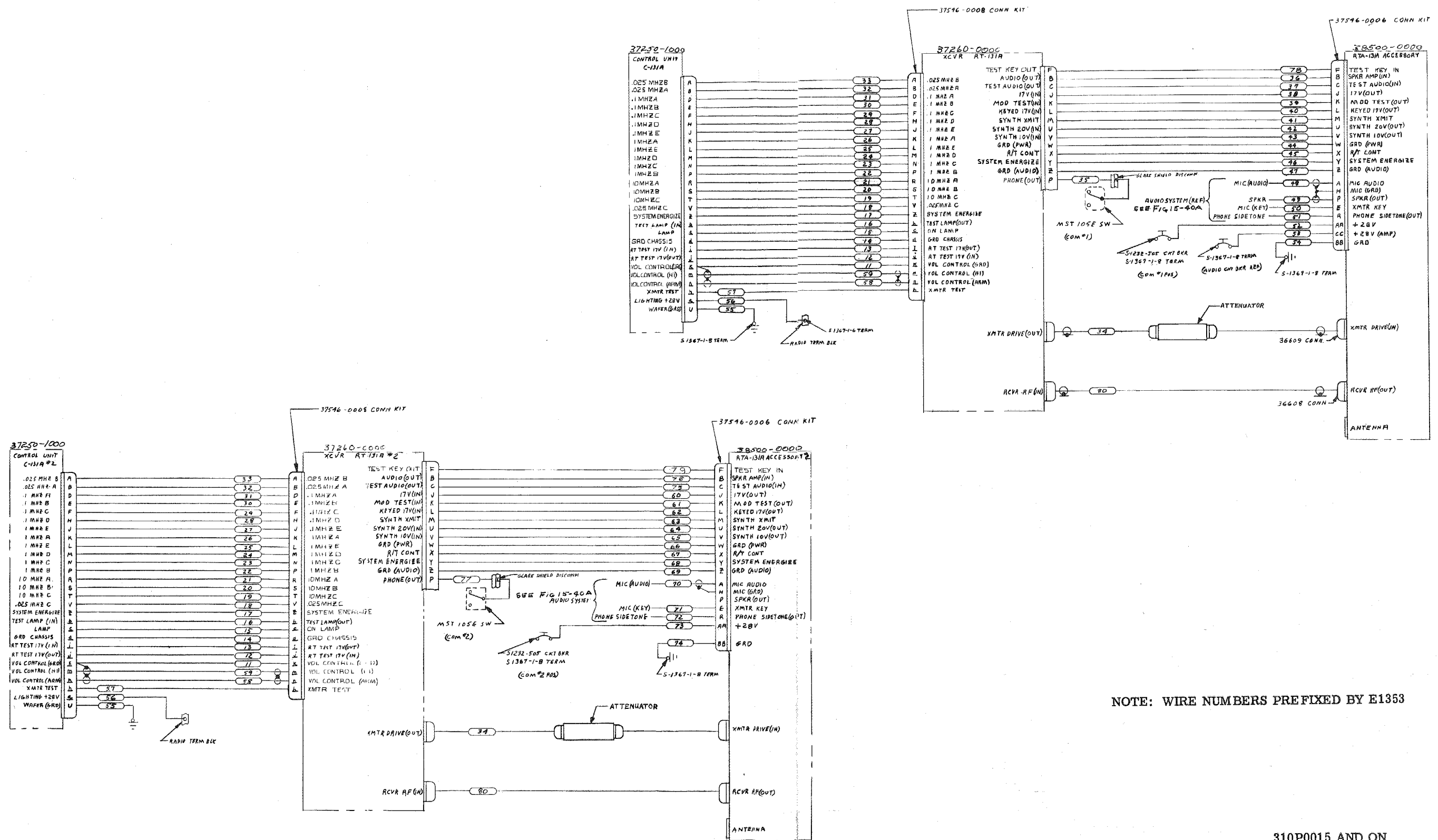


Figure 15-6. Cessna Com 800

310P0015 AND ON
TURBO-SYSTEM 310P0015 AND ON

3100 0601 AND ON

REFERENCE DESIGNATOR CODE:

Reference designators are used in these diagrams to more quickly identify and locate a part or assembly used in any electrical system. A part is identified by reference number on a diagram. The reference number called out on the following page identifies by part number, the description and diagram sheet location.

EXAMPLE -

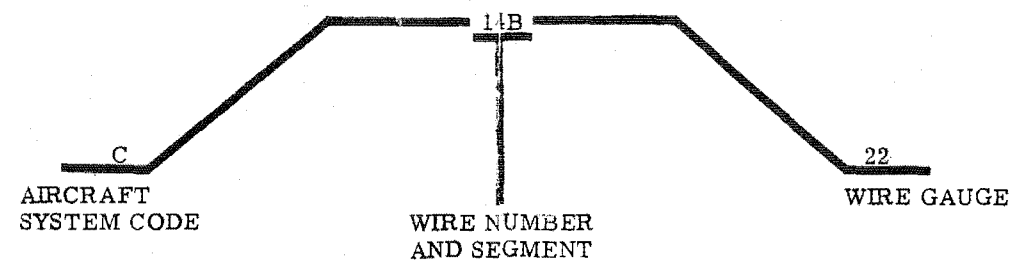
REF. DESIGNATOR	DIA. SHT. NO.	PART NO.	DESCRIPTION
BB20	6	0820001-1	Battery Bus

NOTE

Part numbers referred to on Wiring Diagrams are for reference only. When ordering spare or replacement parts, refer to applicable Parts Catalog.

The wiring diagrams contained in this section clearly show the complete wiring on each item of electrical components listed in the Wiring Diagram Index. The first portion of wire number indicates the Aircraft System, the center portion indicates Wire Number, and the last portion indicates Wire Gauge Size. Each wiring diagram contains part number and nomenclature for each component.

EXAMPLE : C14B22



NOTE

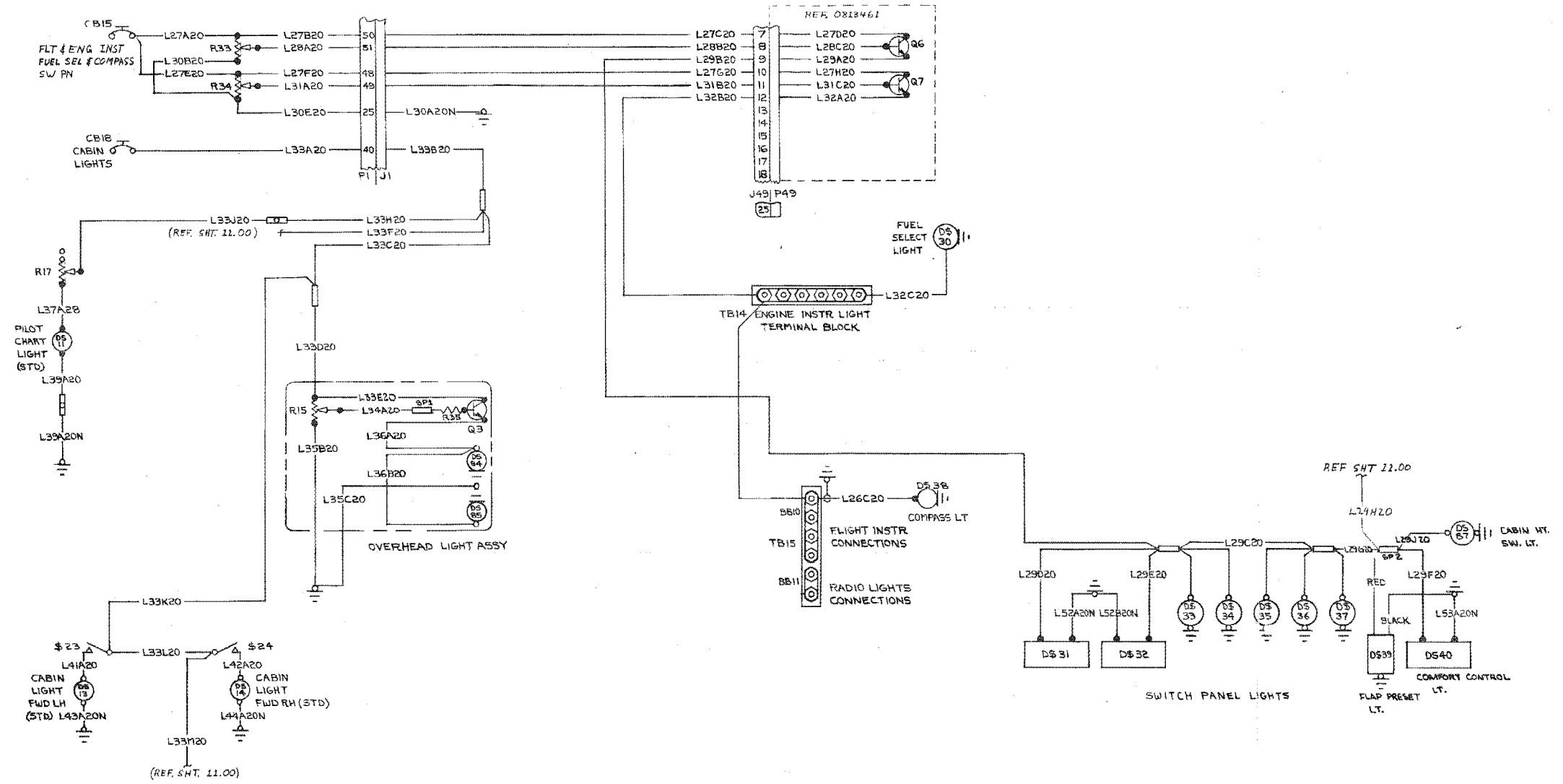
Optional wires (300 and 500 Series) may be determined by dropping the hundreds digit.

EXAMPLE: E327B18 becomes E27B18
E527C18 becomes E27C18

MODEL 310 WIRING REFERENCE DESIGNATORS DESIGNATOR REV F

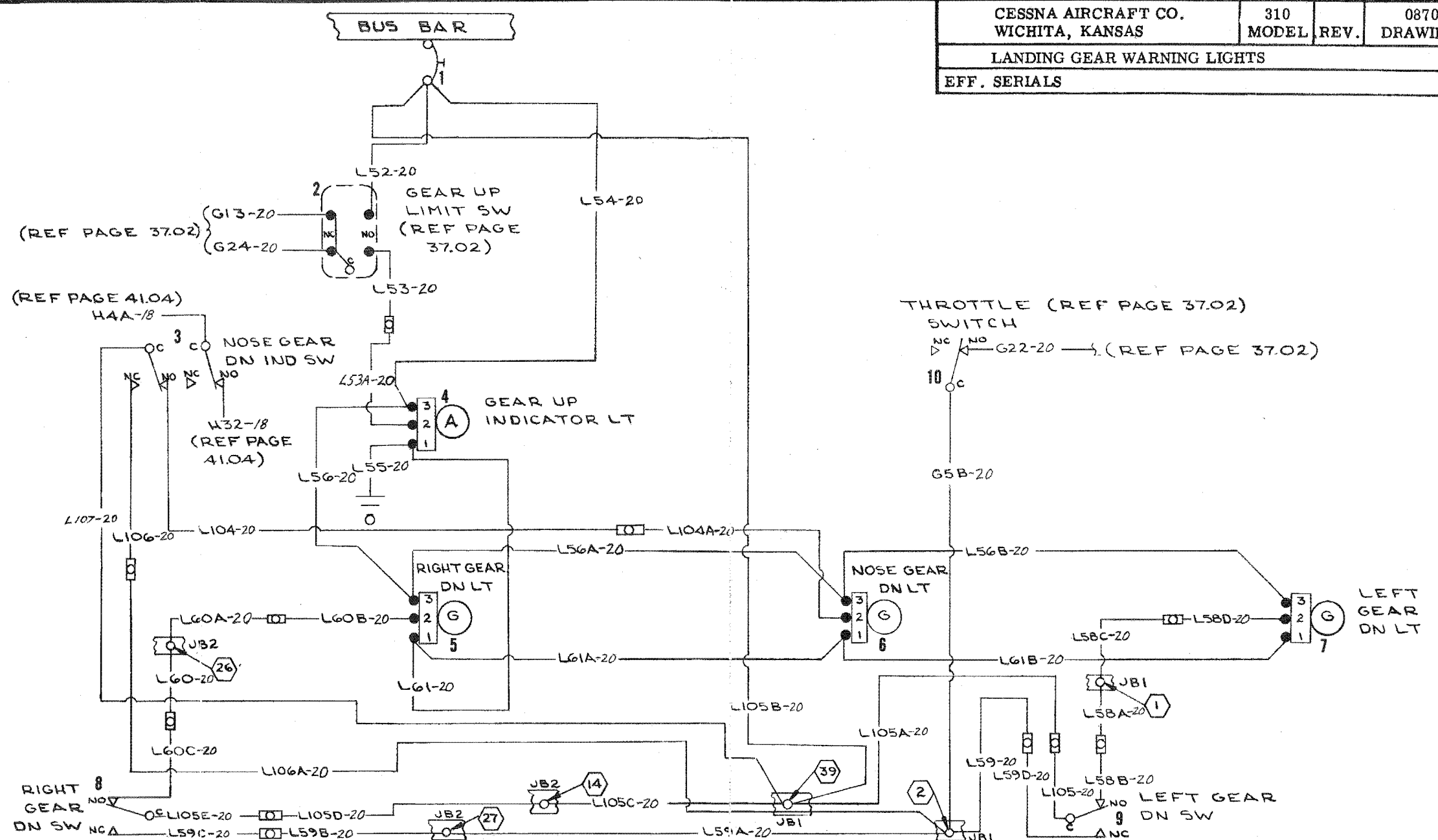
Table with 10 columns: REFERENCE SHEET, PART NUMBER, DESCRIPTION, REFERENCE SHEET, PART NUMBER, DESCRIPTION, REFERENCE SHEET, PART NUMBER, DESCRIPTION, REFERENCE SHEET, PART NUMBER, DESCRIPTION, REFERENCE SHEET, PART NUMBER, DESCRIPTION, REFERENCE SHEET, PART NUMBER, DESCRIPTION. It lists various electrical components and their reference designators.

Cessna AIRCRAFT CO. P.O. BOX 1897 WICHITA, KANSAS 67201 REFERENCE DESIGNATORS FILE CODE (IDENT NO) DRAWING NO D 71379 0808080 SCALE NONE SHEET 2 OF 16



Cessna AIRCRAFT CO.		P. O. BOX 1877 MILWAUKEE, WISCONSIN 53218
TITLE: INTERIOR LIGHTING - STD		
SIZE: D	CODE IDENT. NO: 71379	DRAWING NO: 0808080
SCALE: NONE	REV: F	SHEET 11 OF 16.00

LANDING GEAR WARNING LIGHTS
EFF. SERIALS



ITEM	PART NUMBER	NOMENCLATURE
1	S:232-505	Circuit Breaker
2	OT2RA7	Switch (Gear Up Limit)
3	27B1	Switch (Nose Gear Down Indicator)
4	VM911M4	Light (Gear Up Indicator)
5	VM911M3	Light (Right Gear Down)
6	VM911M3	Light (Nose Gear Down)
7	VM911M3	Light (Left Gear Down)
8	18E3-3	Switch (Left Gear Down Indicator)
9	18E3-3	Switch (Right Gear Down Indicator)
10	8ZR31	Switch Throttle

CESSNA AIRCRAFT CO.
WICHITA, KANSAS

310
MODEL

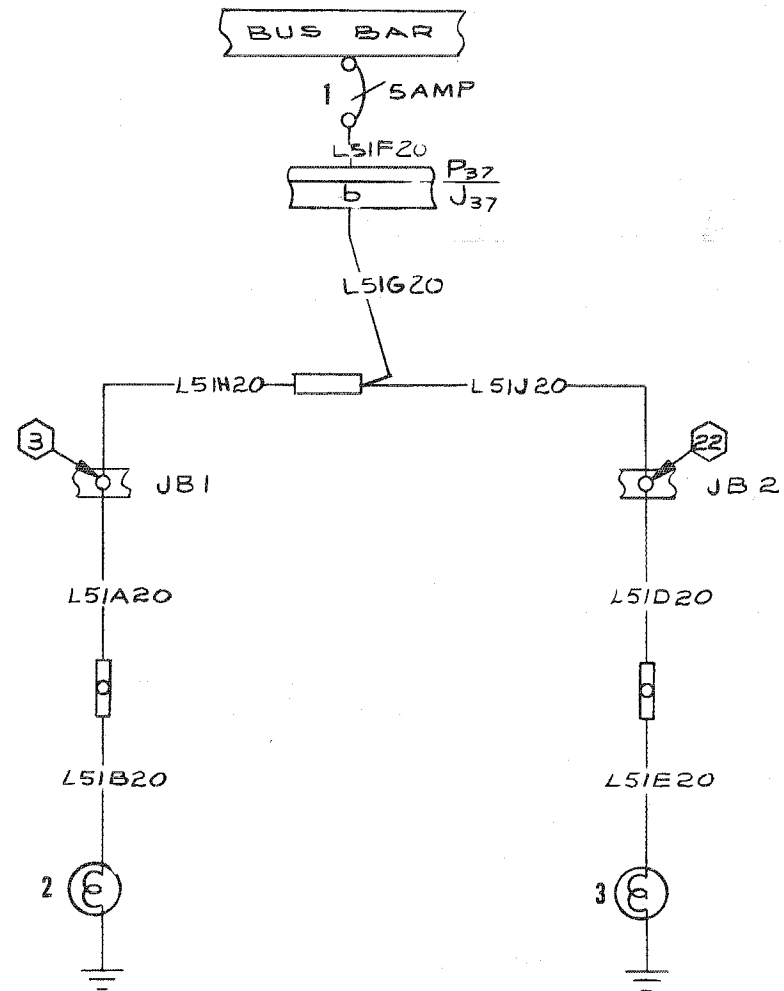
W
REV.

0870090
DRAWING NO.

34.02
PAGE

WING DE-ICE LIGHT (OPT)

EFF. SERIALS



ITEM

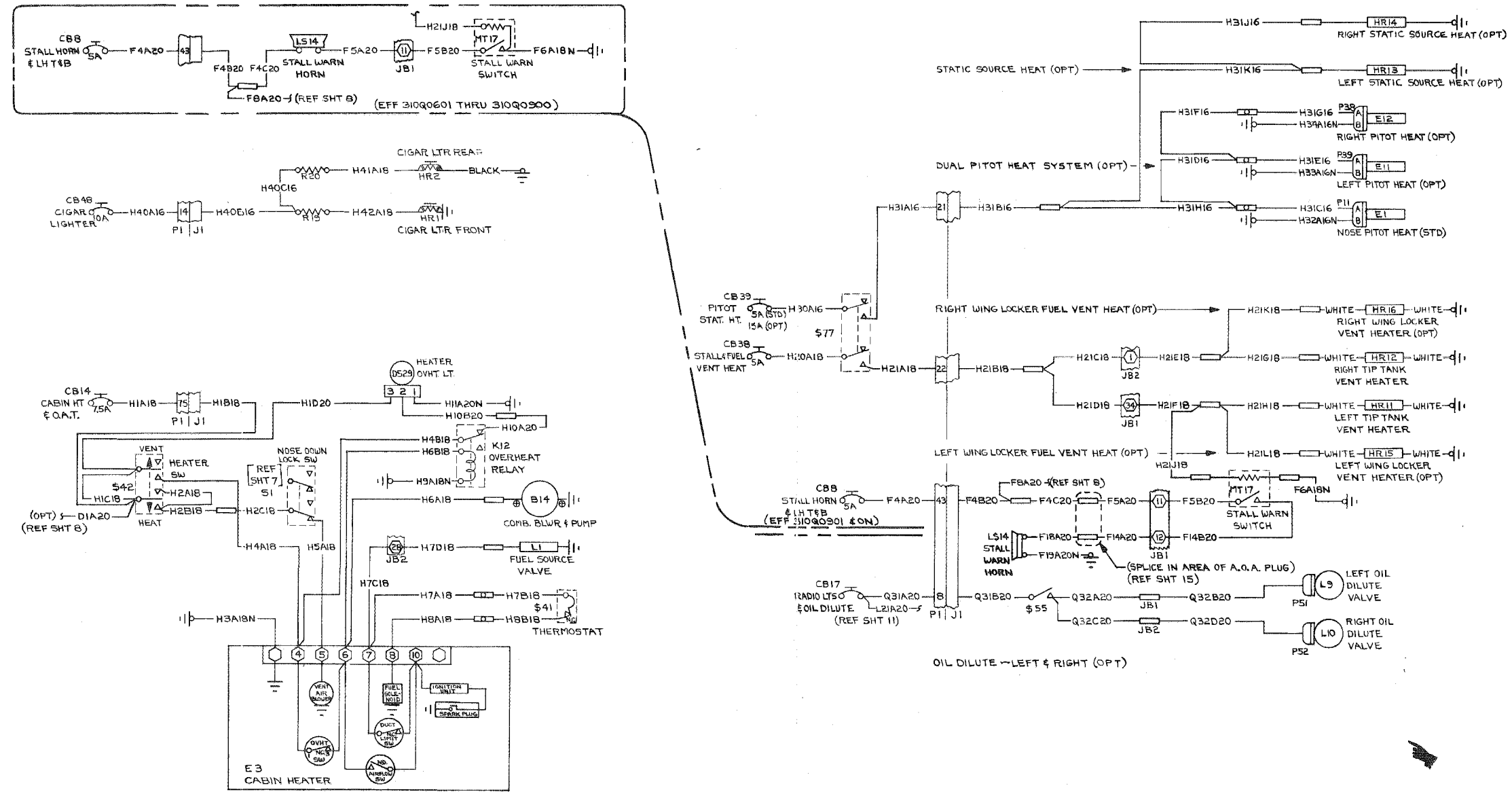
PART NUMBER

NOMENCLATURE

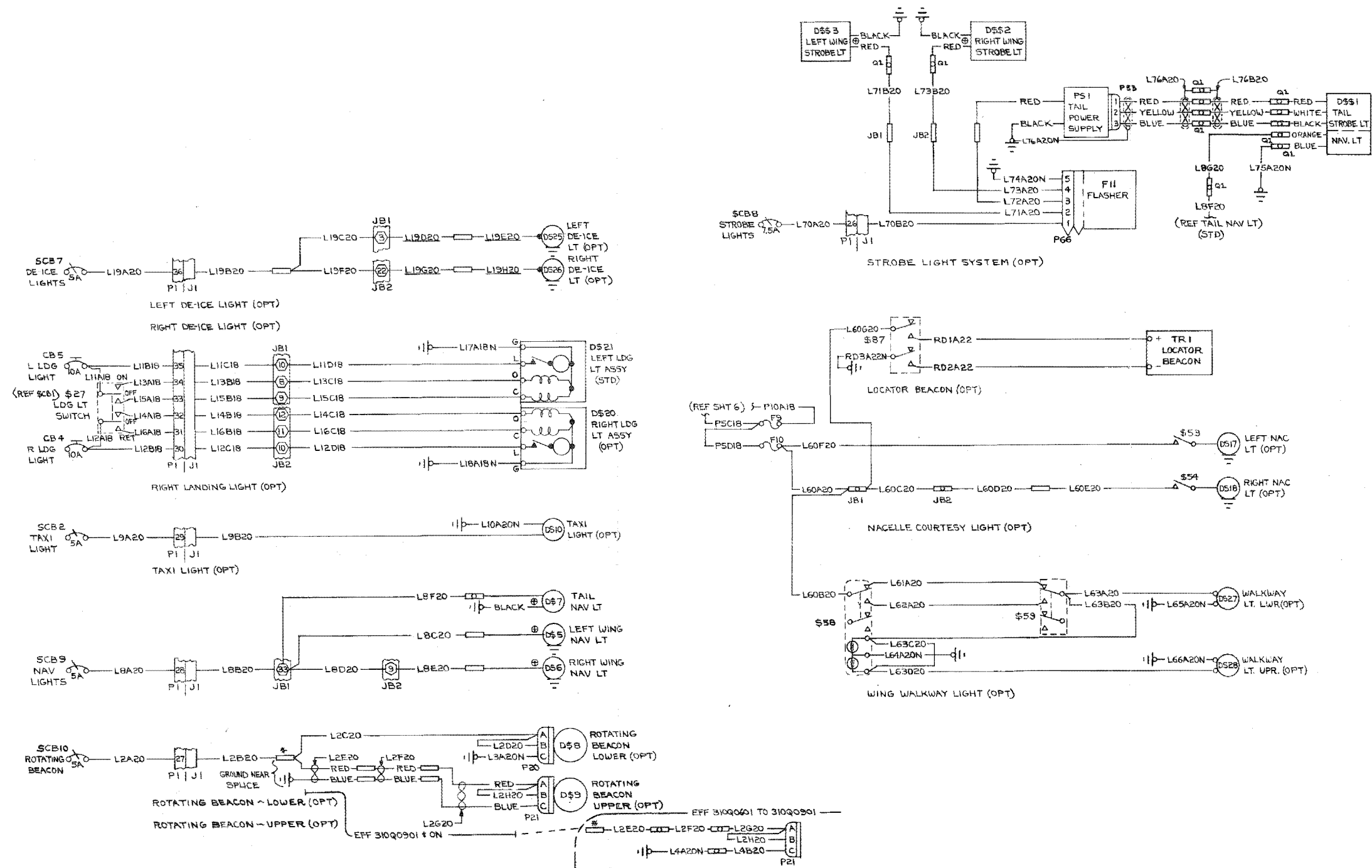
1
2
3

112-205-101
0820501-5
0820501-5

Circuit Breaker
LH Deice Light
RH Deice Light

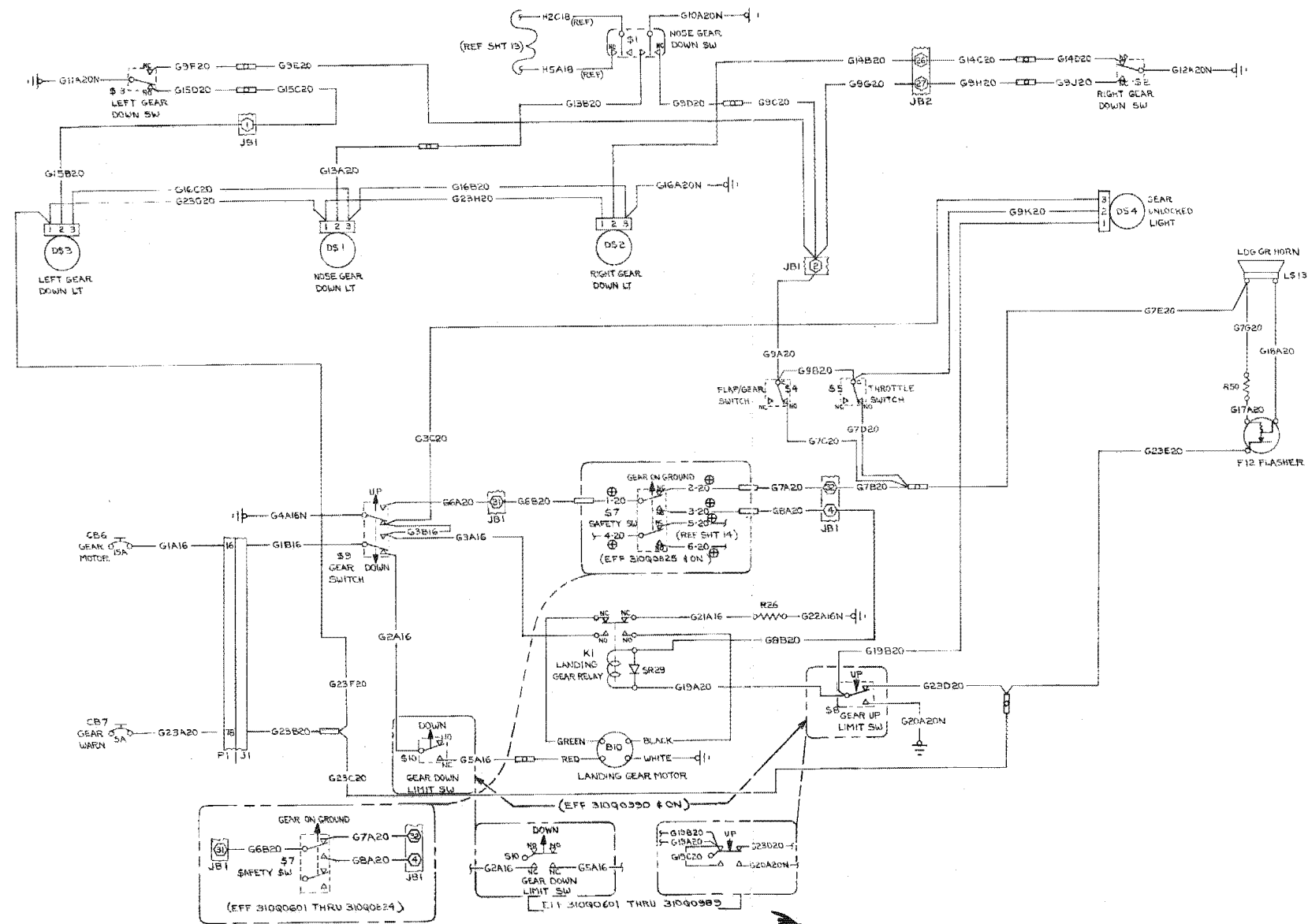


Cessna		AIRCRAFT CO.	P. O. BOX 1877 MILITARY & TRAVEL DIVISION WICHITA, KANSAS 67201
TITLE: CABIN HEAT, CIGAR LTRS, PITOT STATIC HT, STALL HORN, FUEL VENT HT. & OIL DILUTE			
SIZE: D	CODE IDENT NO: 71379	DRAWING NO: 0808080	
SCALE: NONE	REF: F	SHEET 13 OF 16	

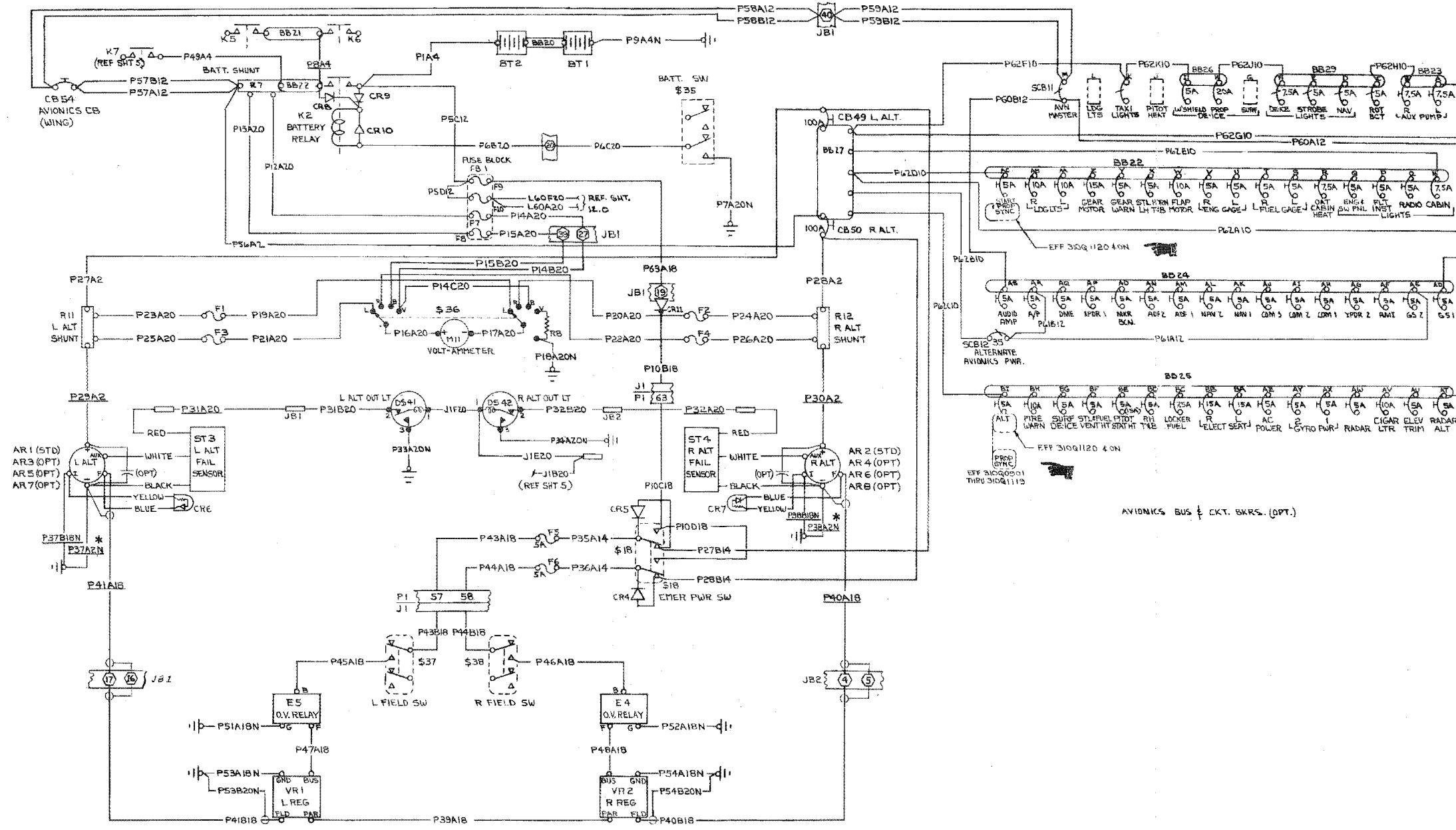


NOTE: 300 OR 500 NUMBERS PRECEDING AIRCRAFT WIRE NUMBERS ARE OPTIONAL (SEE NOTE ON PAGE 14-101.)

Cessna AIRCRAFT CO. P. O. BOX 1877 MILWAUKEE 8, WISCONSIN	
TITLE: EXTERIOR LIGHTING ~ STD & OPT	
SIZE: D	CODE IDENT. NO. 71379
SCALE: NONE	DRAWING NO. 0808080
REF: G	SHEET 12 OF 16



Cessna		P. O. BOX 1677 MILWAUKEE, WISCONSIN 53211
TITLE:		LANDING GEAR
SIZE:	CODE IDENT. NO.:	DRAWING NO.:
D	71379	0808080
SCALE: NONE	REV: 8	SHEET 7 OF 16

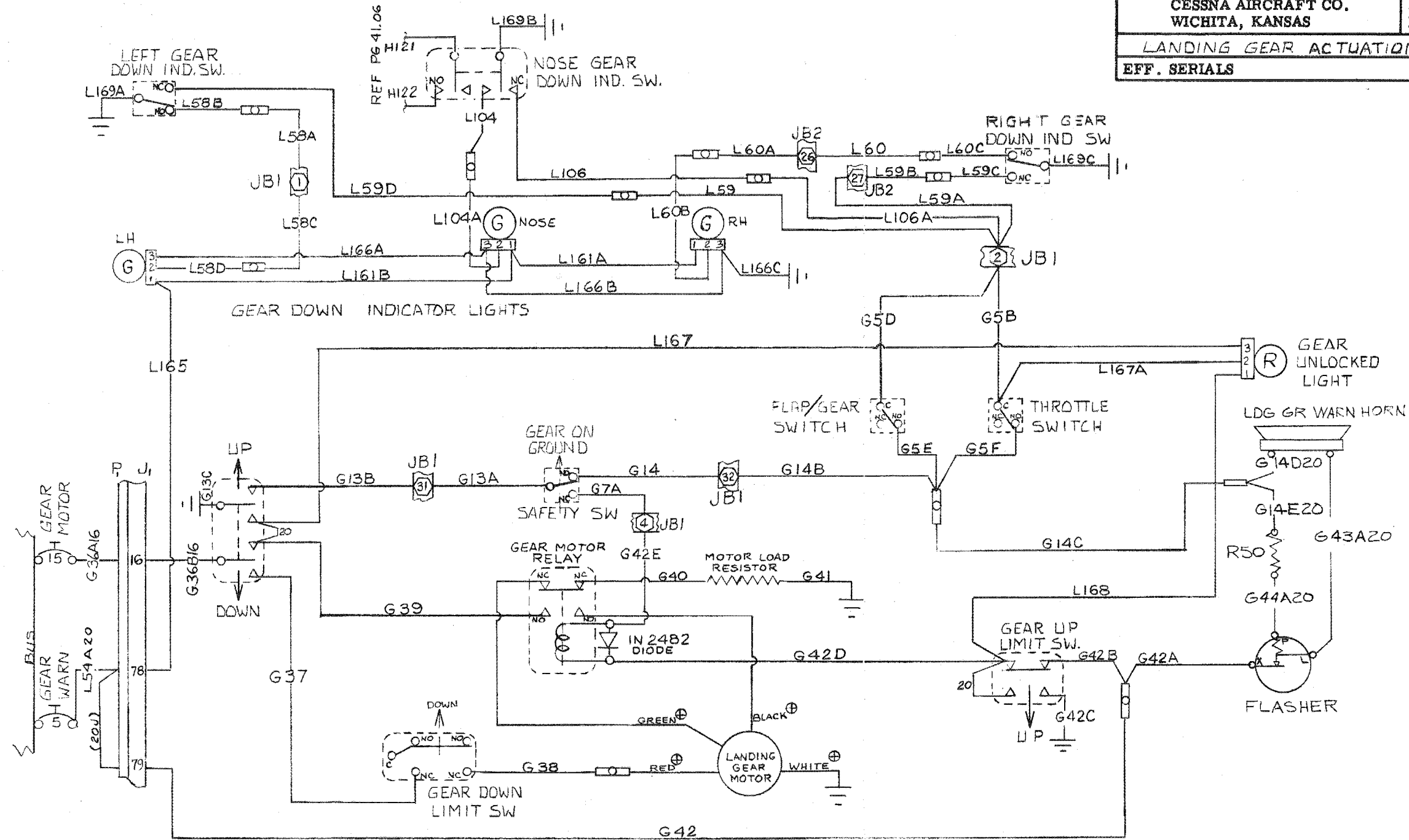


EFF. 310Q0901 4 ON

*P37A2N AND P38A2N USED ON 100 AMP OPTION ONLY

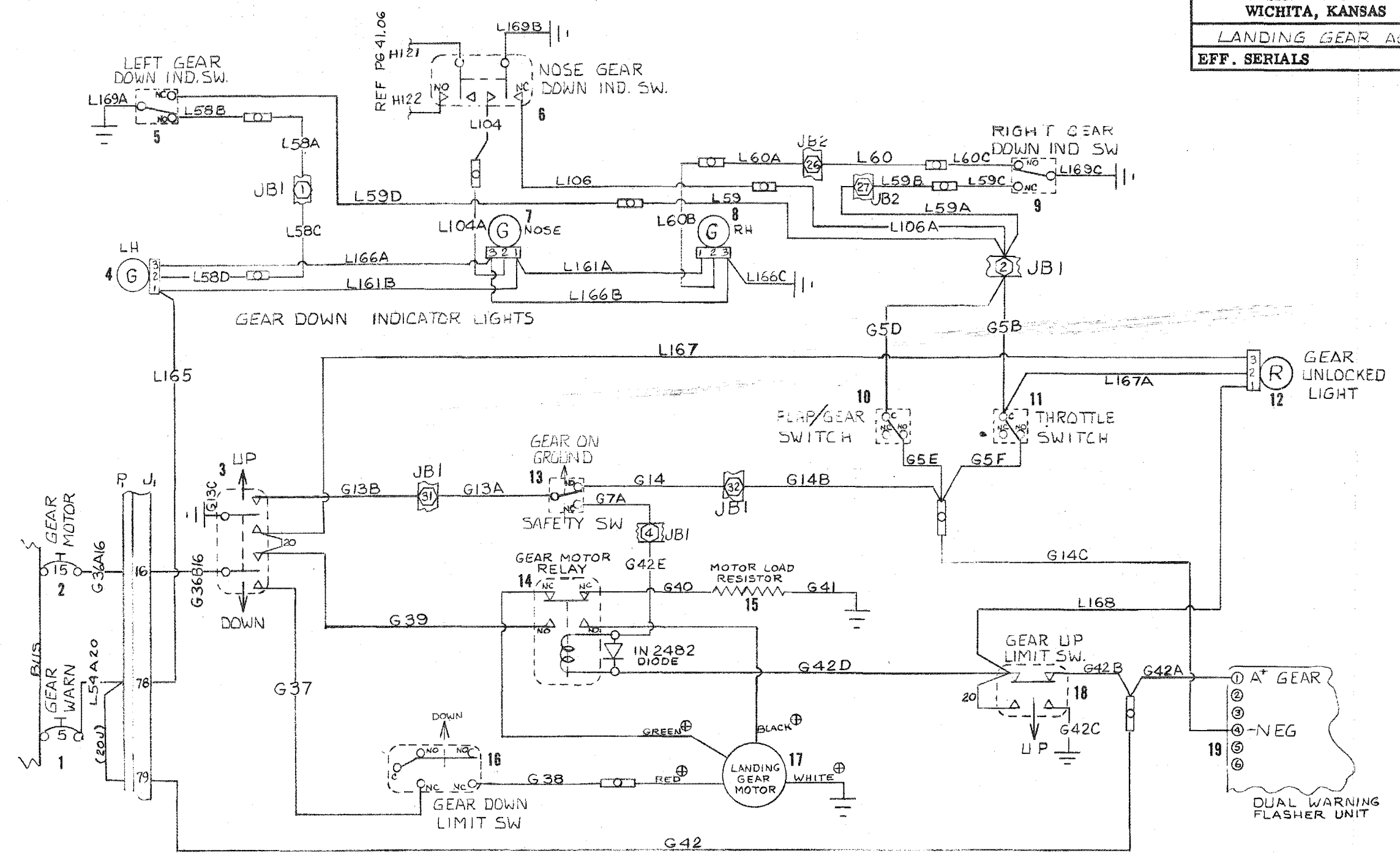
Cessna AIRCRAFT CO. P. O. BOX 1807 WICHITA, KANSAS 67201	
POWER DISTRIBUTION (OPT. C.B.)	
SIZE: D	DRAWING NO.: 0808080
CODE IDENT NO.: 71379	REF: 6
SCALE: NONE	SHEET: 6 OF 16.0

LANDING GEAR ACTUATION & WARNING
EFF. SERIALS

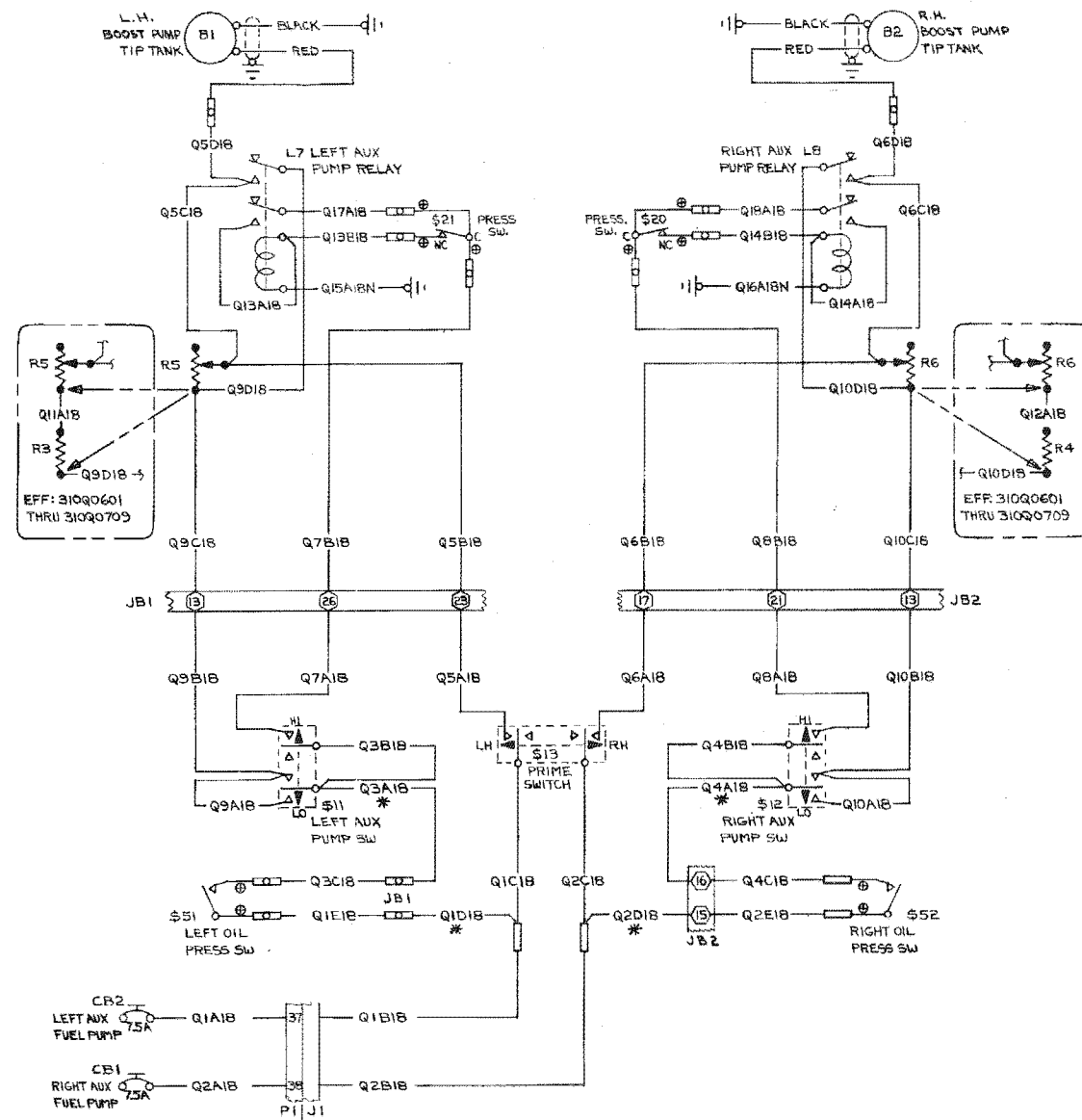


ITEM	PART NUMBER	NOMENCLATURE	ITEM	PART NUMBER	NOMENCLATURE
1	S1232-505	Circuit Breaker	11	BZR31	Switch
2	S1232-515	Circuit Breaker	12	01-911806-6	Light
3	0813525-4	Switch	13	IVA20	Switch
4	VM911M3	Light	14	6D41H220	Relay
5	1SE1-3	Switch	15	FR100-5	Resistor
6	2VB1	Switch	16	BZ7RT04	Switch
7	VM911M3	Light	17	9910002-3	Motor
8	VM911M3	Light	18	BZ3YT	Switch
9	1SE1-3	Switch	19	R102-12V	Warning Horn Flasher
10	MS25253-1	Switch	20	A2948	Gear Warning Horn
			21	1-3/4D48F40	Resistor

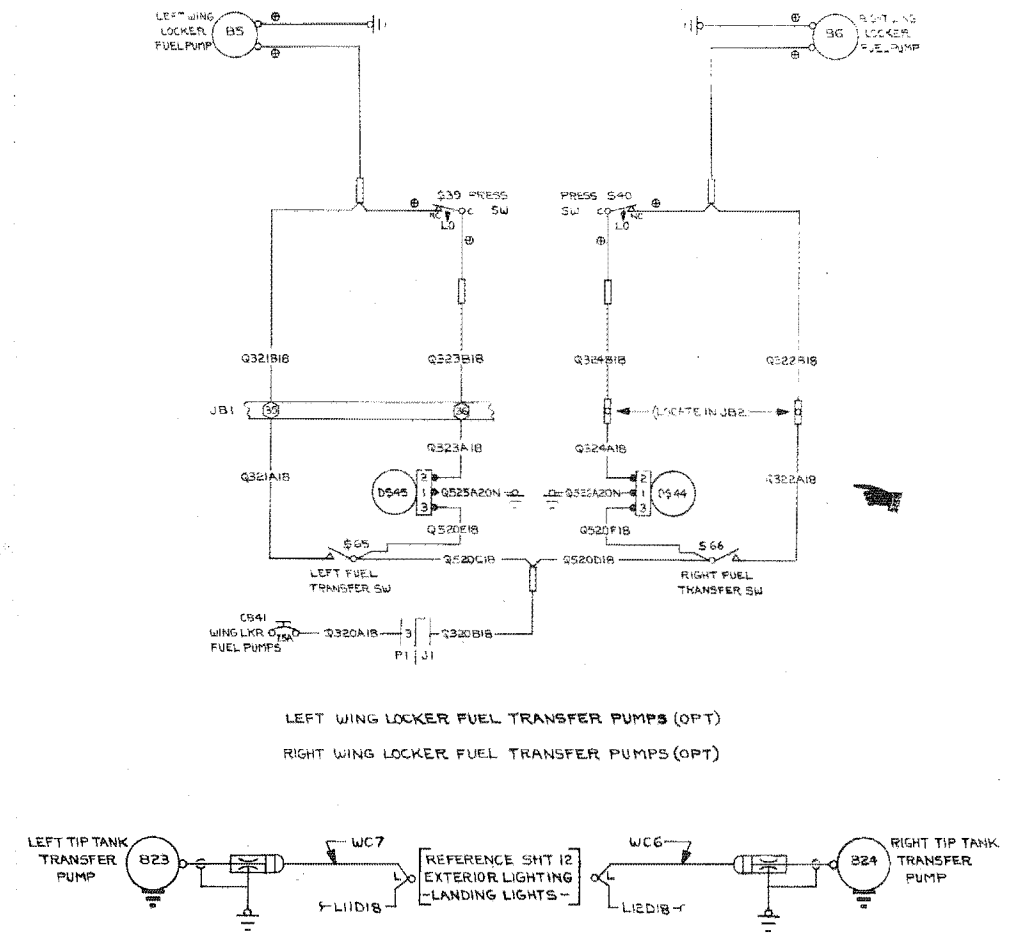
LANDING GEAR ACTUATION & WARNING
EFF. SERIALS



ITEM	PART NUMBER	NOMENCLATURE	ITEM	PART NUMBER	NOMENCLATURE
1	S1232-505	Circuit Breaker	10	MS25253-1	Switch
2	S1232-515	Circuit Breaker	11	BZR31	Switch
3	0813525-4	Switch	12	01-911806-6	Light
4	VM911M3	Light	13	IVA20	Switch
5	1SE1-3	Switch	14	6D41H220	Relay
6	2VB1	Switch	15	FR100-5	Resistor
7	VM911M3	Light	16	BZ7RT04	Switch
8	VM911M3	Light	17	9910002-3	Motor
9	1SE1-3	Switch	18	BZ3YT	Switch
			19	No. 285	Dual Warning Flasher Unit

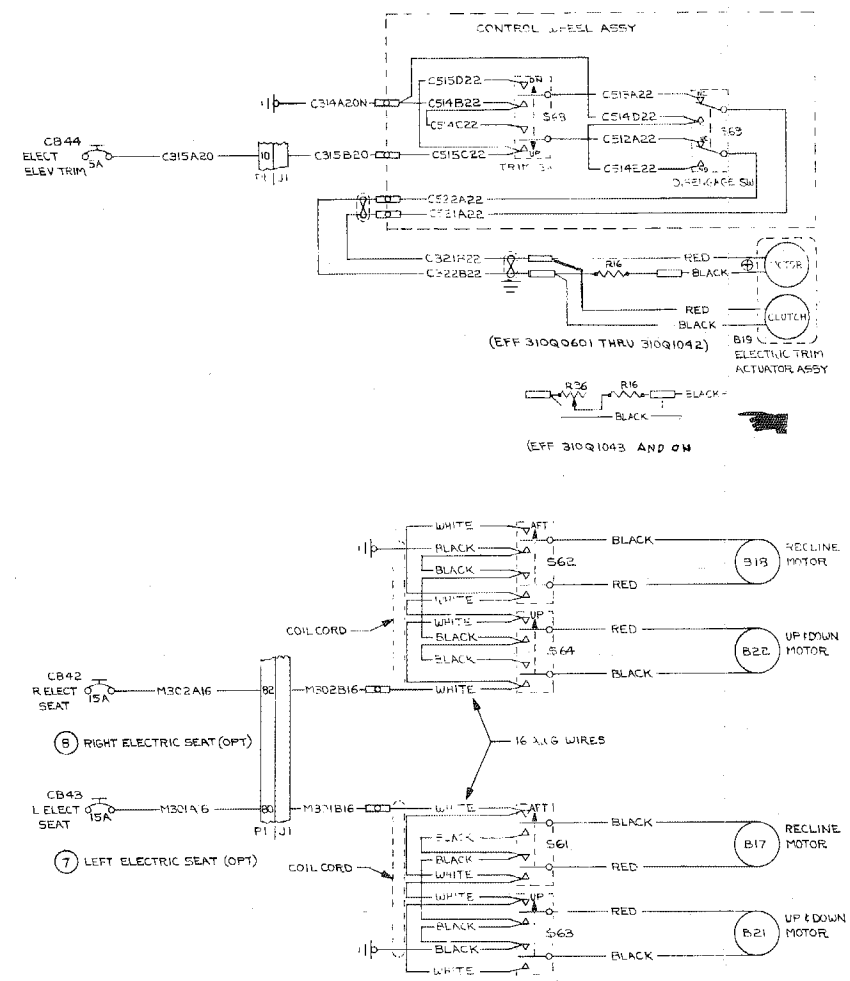
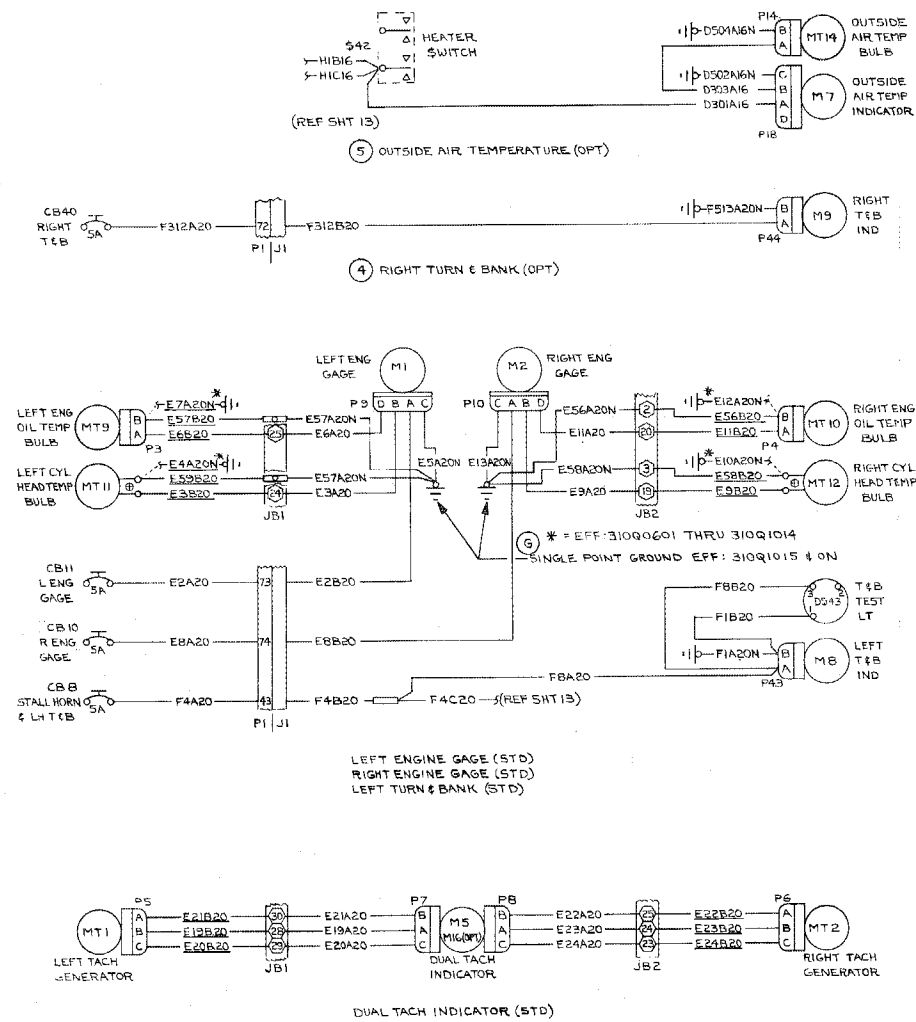


WHEN OPTIONAL ENGINE IS INSTALLED (T310)
 SPLICE Q1D1B TO Q3A1B IN JB1 &
 JUMPER Q2D1B TO Q4A1B IN JB2



LEFT WING LOCKER FUEL TRANSFER PUMPS (OPT)
 RIGHT WING LOCKER FUEL TRANSFER PUMPS (OPT)

		AIRCRAFT CO. P. O. BOX 1877 MILITARY & TRAVEL DIVISION WICHITA, KANSAS 67201	
TITLE: FUEL PUMPS ~ STD & OPT			
SIZE: D	CODE IDENT. NO.: 71379	DRAWING NO.: 0808080	
SCALE: NONE	REF: G	SHEET: 9 OF 16	



Cessna AIRCRAFT CO. P. O. BOX 1877 MILITARY & TWIN DESIGN WICHITA, KANSAS 67201

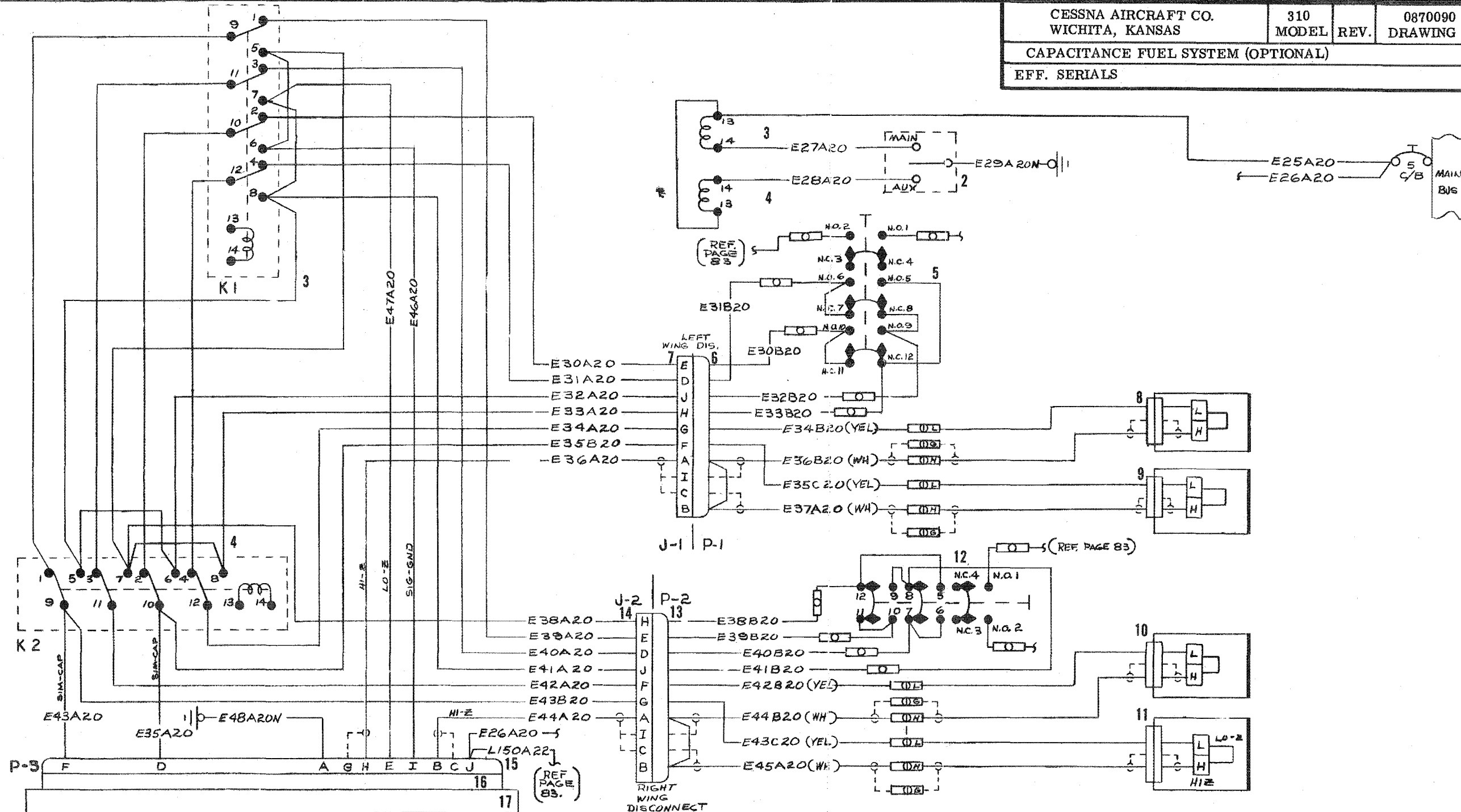
TITLE: TACH GEN, ENGINE INSTR, T&B(LH&RH), O.A.T, ELECT TRIM & ELECT SEAT (LH&RH)

SIZE: D CODE IDENT. NO.: 71379 DRAWING NO.: 0808080

SCALE: NONE REF: C SHEET 8 OF 16

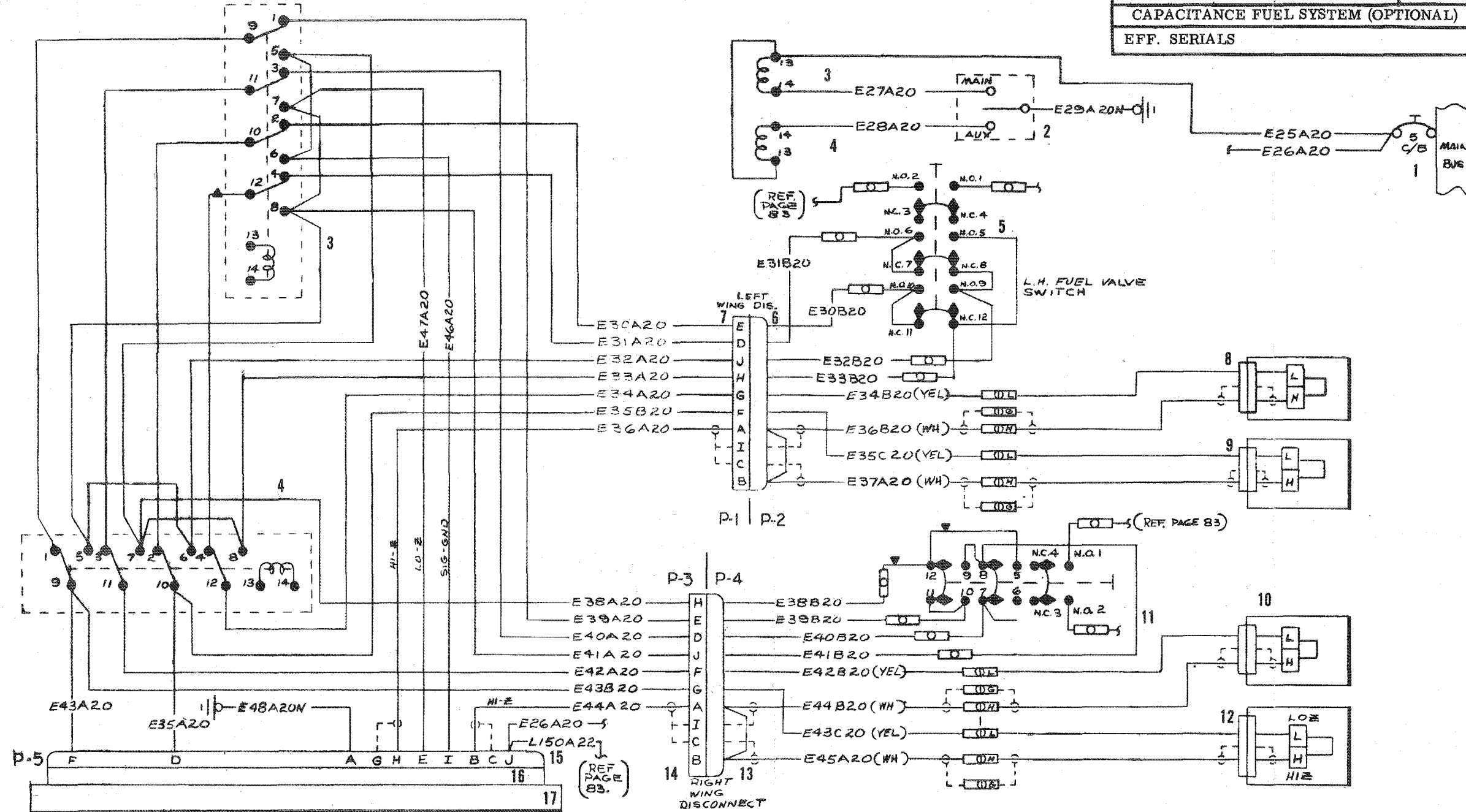
CAPACITANCE FUEL SYSTEM (OPTIONAL)

EFF. SERIALS

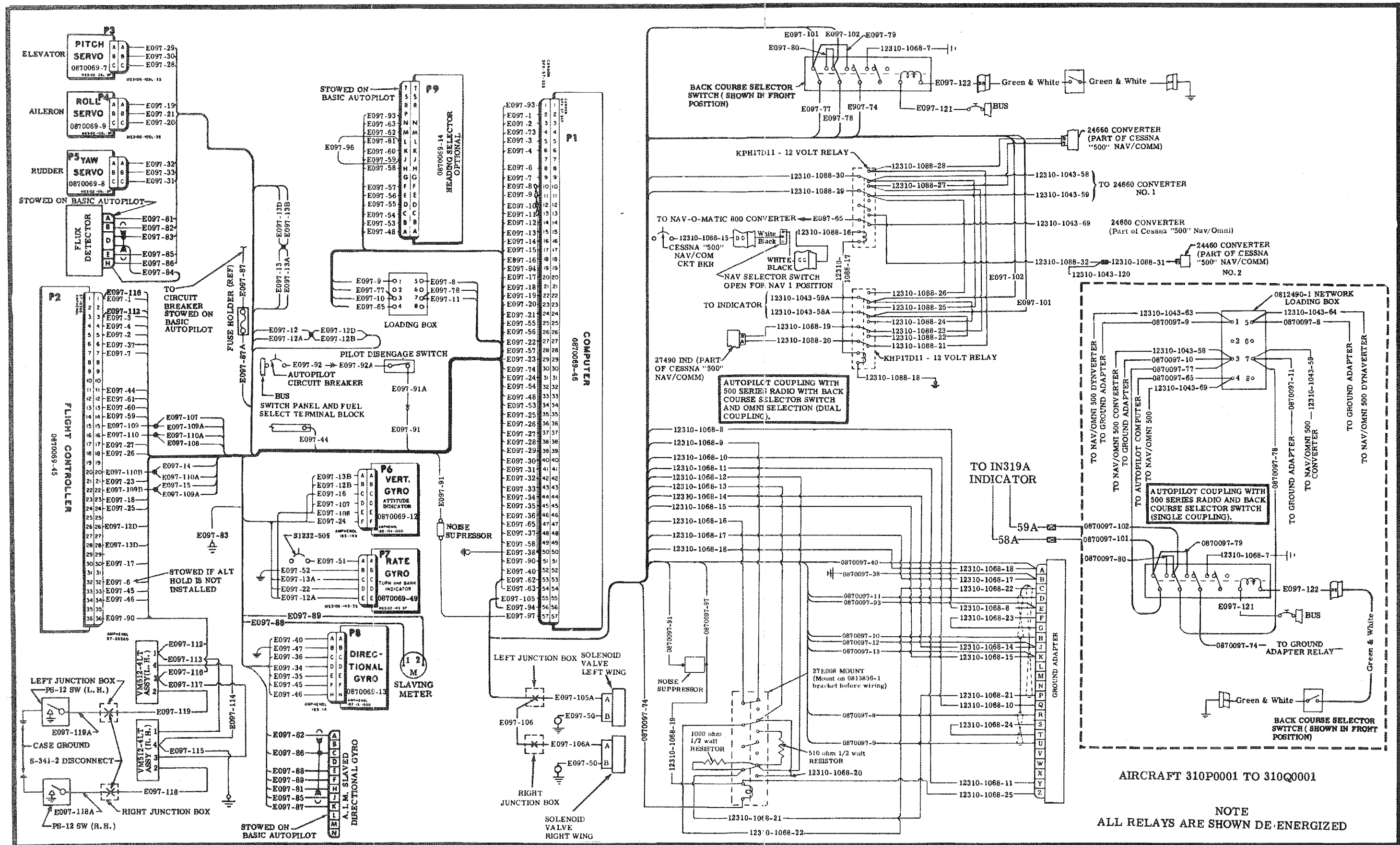


Unit	Part Number	Nomenclature	Unit	Part Number	Nomenclature
1	S1232-505	Circuit Breaker	10	PAA700-3	(RH) Main Tank Unit
2	S1695-4	Selector Switch	11	PAA700-2	(RH) Aux Tank Unit
3	281XDX-24VDC	Relay	12	36-604	(RH) Selector Valve Switch
4	281XDX-24VDC	Relay	13	MS3106R-18-1P	Plug
5	36-604	(LH) Selector Valve Switch	14	MS3100R-18-1S	Receptacle
6	MS3106R-18-1P	Plug	15	MS3106E-18-1S	Receptacle
7	MS3100R-18-1S	Receptacle	16	MS3106E-18-1P	Plug
8	PAA700-3	(LH) Main Tank Unit	17	C662013-0101	Indicator
9	PAA700-2	(LH) Aux Tank Unit			

CAPACITANCE FUEL SYSTEM (OPTIONAL)
EFF. SERIALS



Item	Part Number	Nomenclature	Item	Part Number	Nomenclature
1	S1232-505	Circuit Breaker	10	PAA700-3	(RH) Main Tank Unit
2	S1695-4	Selector Switch	11	36-604	(RH) Selector Valve Switch
3	281XDX-24VDC	Relay	12	PAA700-2	(RH) Aux Tank Unit
4	281XDX-24VDC	Relay	13	MS3106R-18-1P	Plug
5	36-604	(LH) Selector Valve Switch	14	MS3100R-18-1S	Receptacle
6	MS3106R-18-1P	Plug	15	MS3106E-18-1S	Receptacle
7	MS3100R-18-1S	Receptacle	16	MS3100E-18-1P	Plug
8	PAA700-3	(LH) Main Tank Unit	17	C662013-0101	Indicator
9	PAA700-2	(LH) Aux Tank Unit			



AIRCRAFT 310P0001 TO 310Q0001

NOTE
ALL RELAYS ARE SHOWN DE-ENERGIZED

Figure 15-22. Nav-O-Matic 800 Basic Wiring Circuit (Sheet 3 of 3)

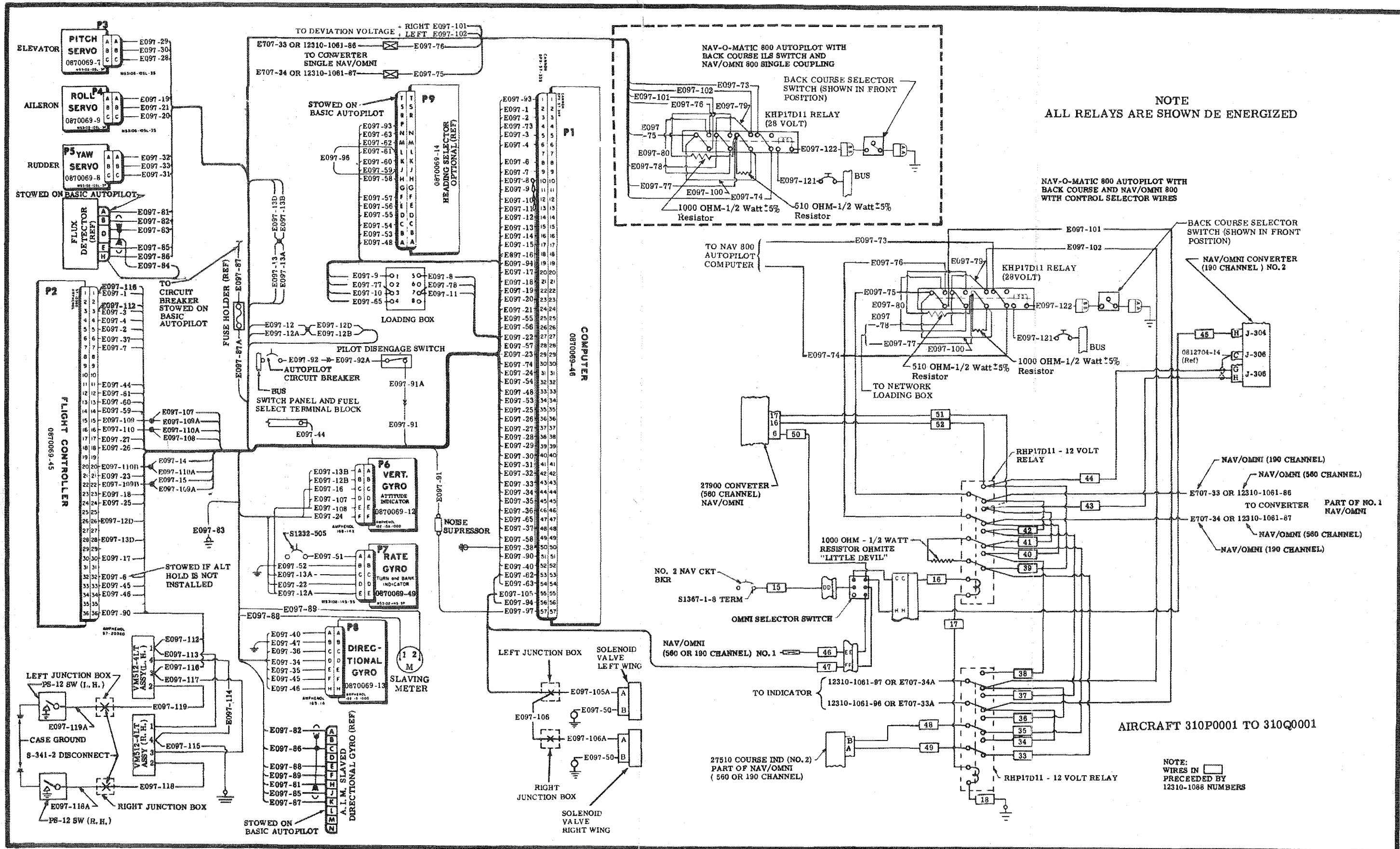


Figure 15-22. Nav-O-Matic 800 Basic Wiring Circuit (Sheet 2 of 3)

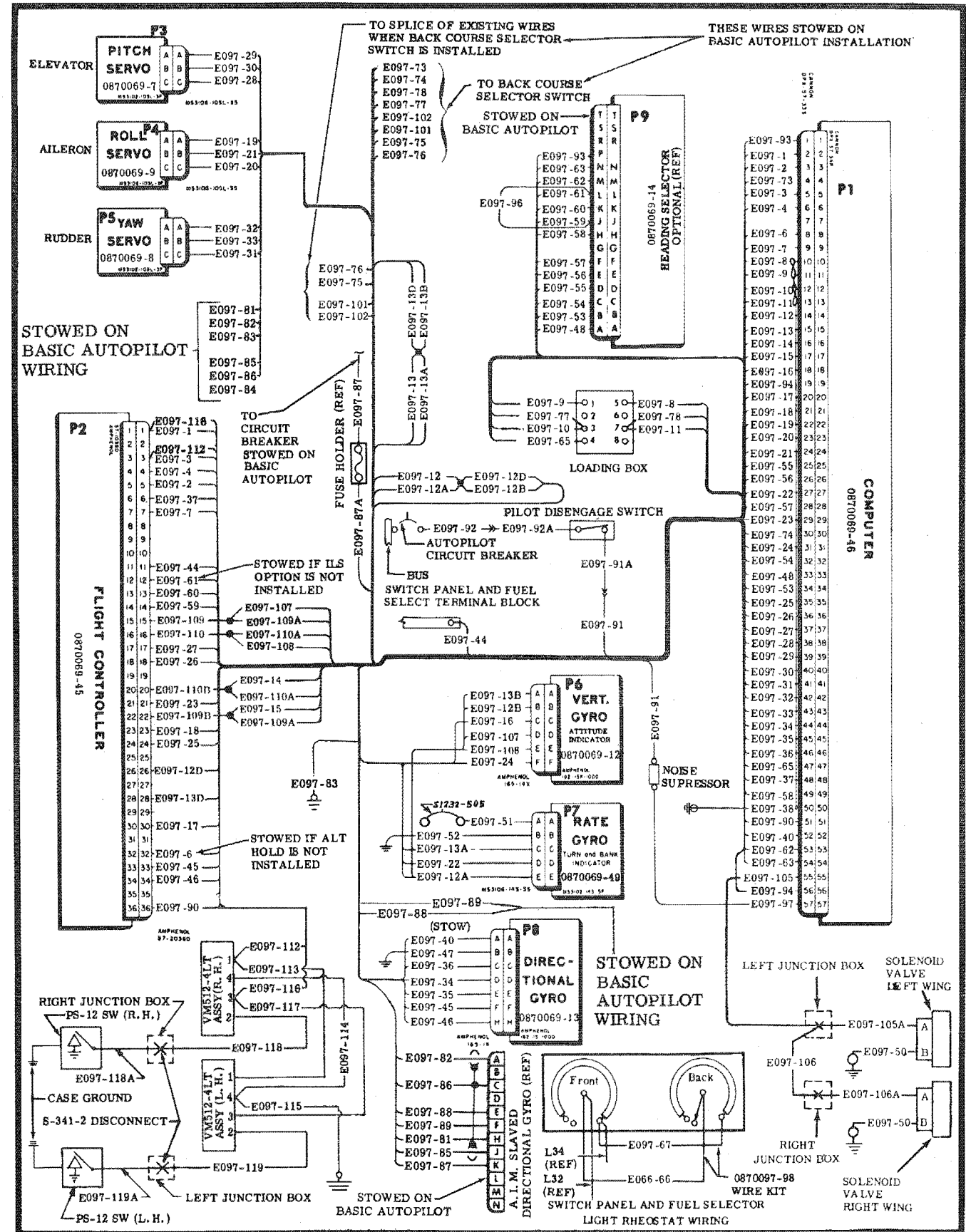


Figure 15-22. Nav-O-Matic 800 Basic Wiring Circuit (Sheet 1 of 3)

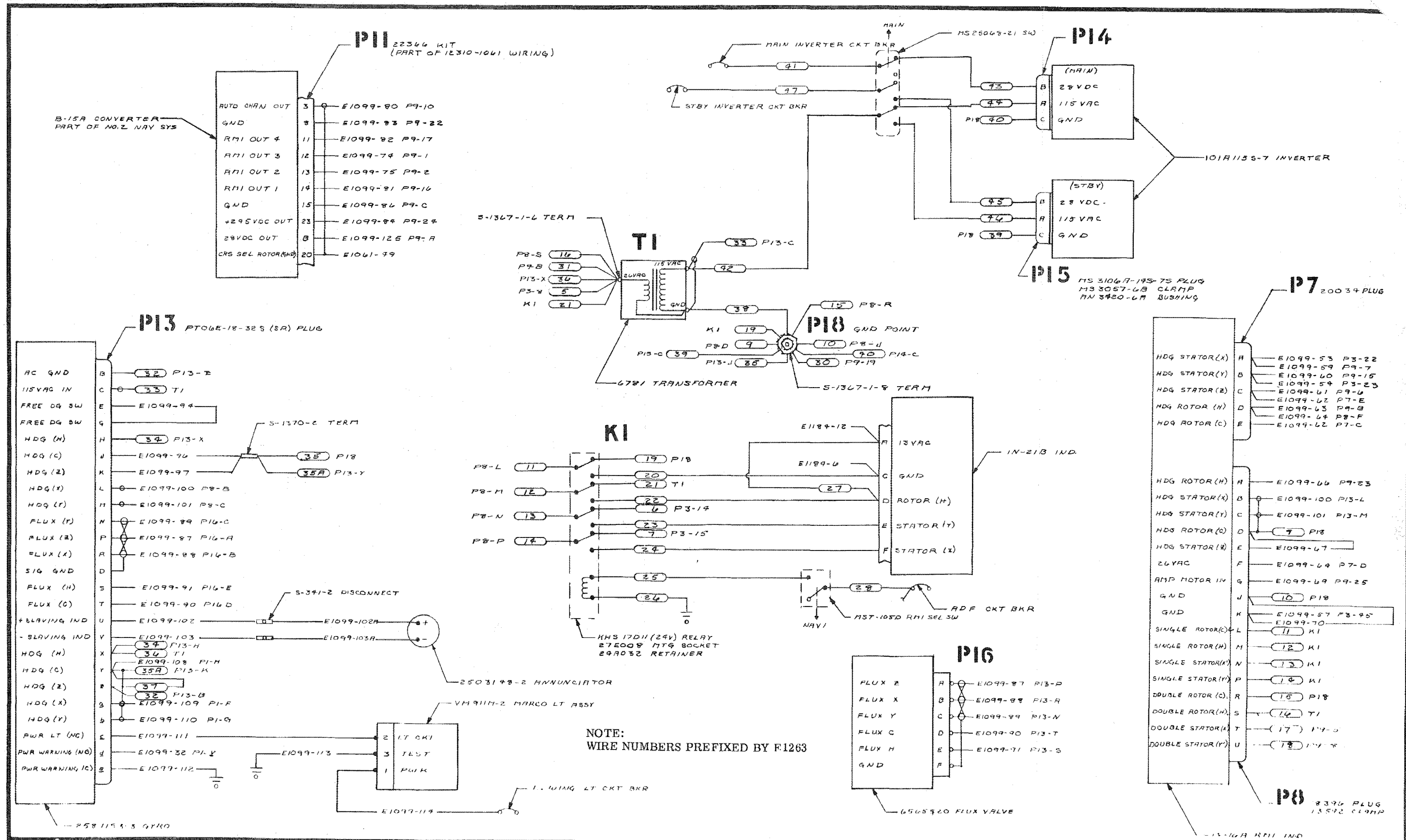
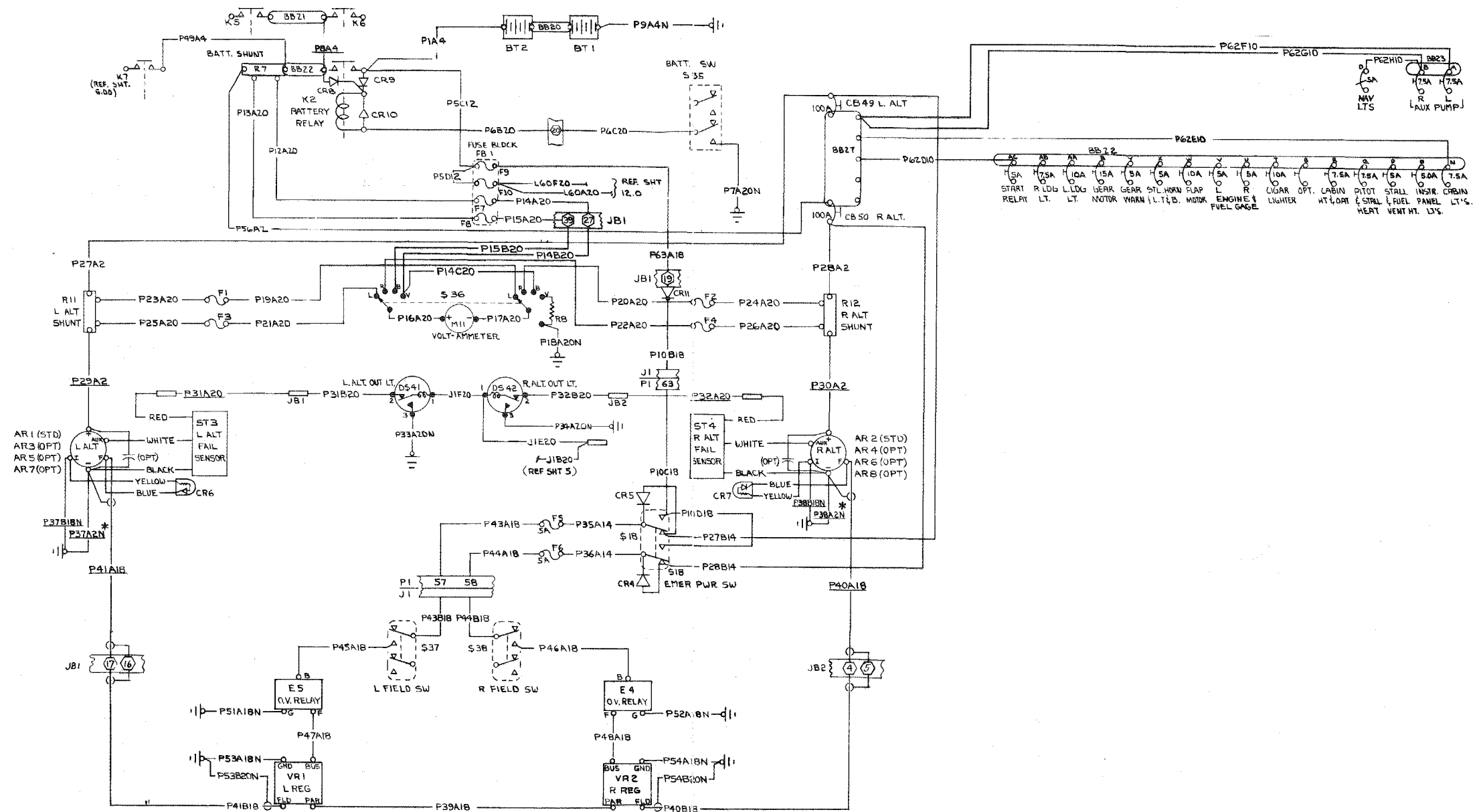


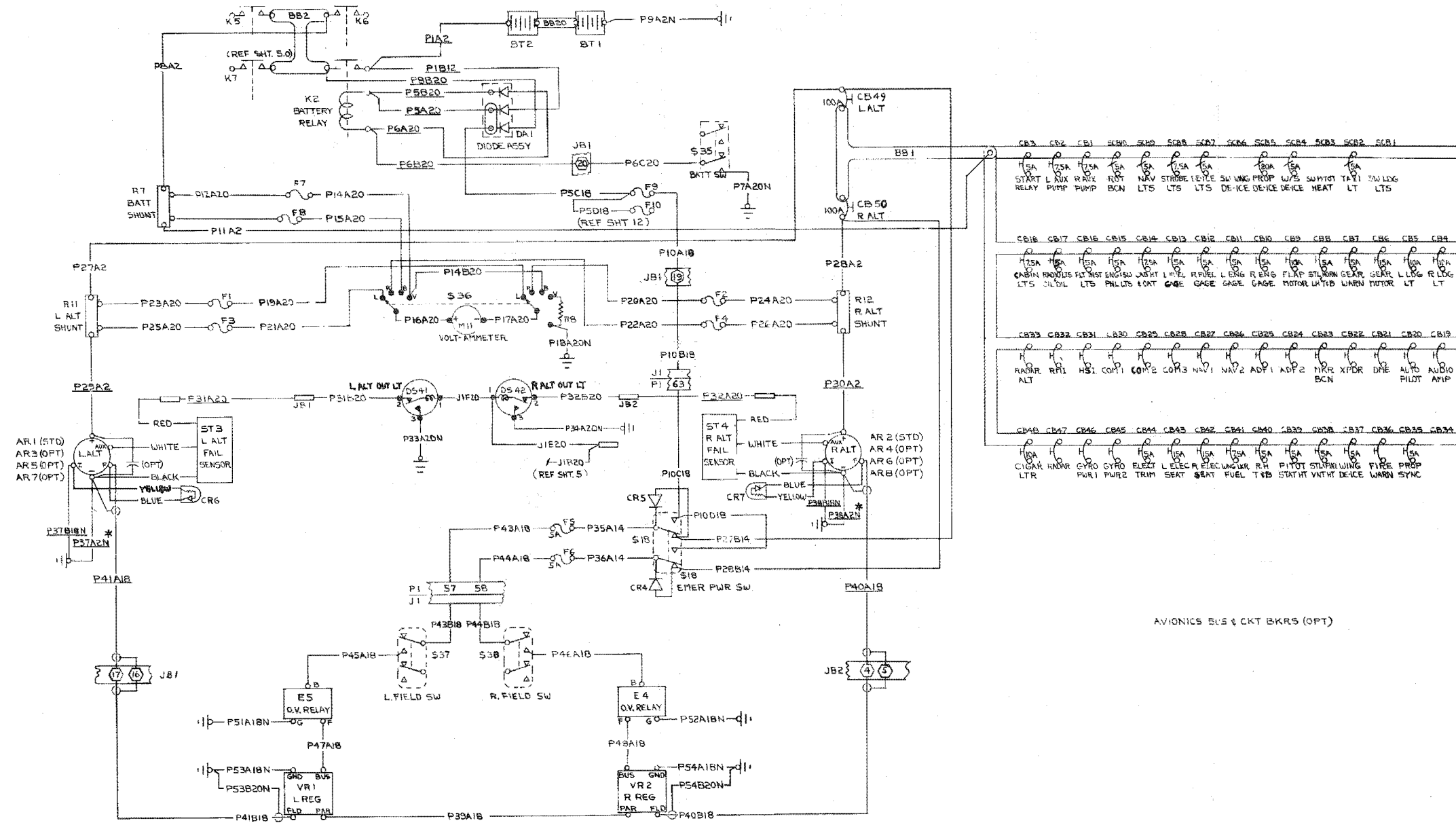
Figure 15-21. PN101 NAV System Coupled with C-14 Gyro and Cessna 800 RMI (Sheet 2 of 2)



EFF. 1 31000901:04

*P37A2N AND P38A2N USED ON 100 AMP OPTION ONLY

Cessna AIRCRAFT CO. P. O. BOX 1827 WICHITA, KANSAS 67201	
TITLE: POWER DISTRIBUTION (STD.)	
SIZE: D	DRAWING NO.: 0808080
SCALE: NONE	SHEET 6 OF 16



*P37A2N AND P38A2N USED ON 100 AMP OPTION ONLY

EFF. 31002601 THRU 31000900

AVIONICS BUS & CKT BKRS (OPT)

Cessna AIRCRAFT CO. P. O. BOX 1877 MILWAUKEE & TRIN DIVISION WICHITA, KANSAS 67201	
TITLE: POWER DISTRIBUTION	
SIZE: D	CODE IDENT. NO.: 71379
DRAWING NO.: 0808080	
SCALE: NONE	SHEET 6 OF 16

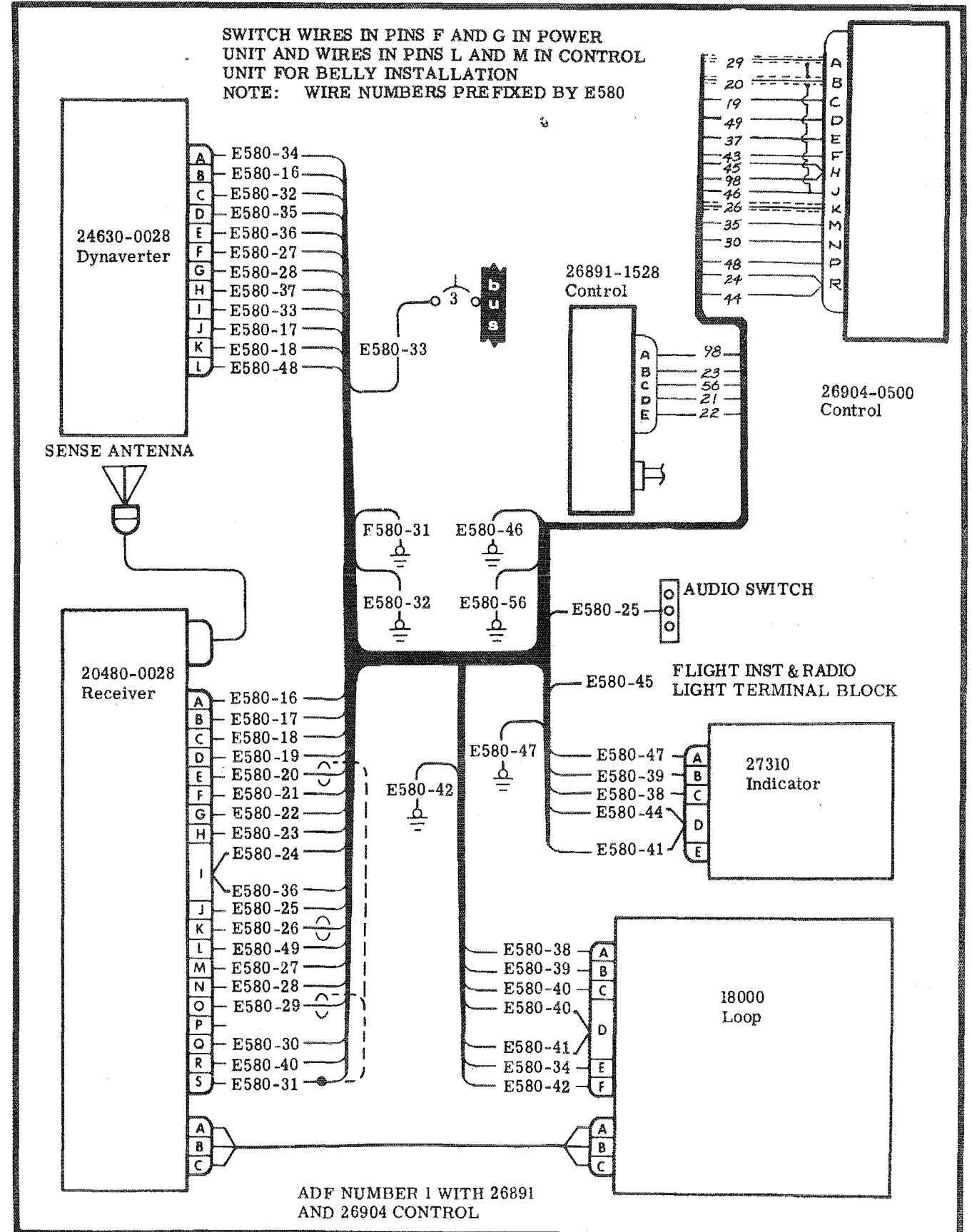
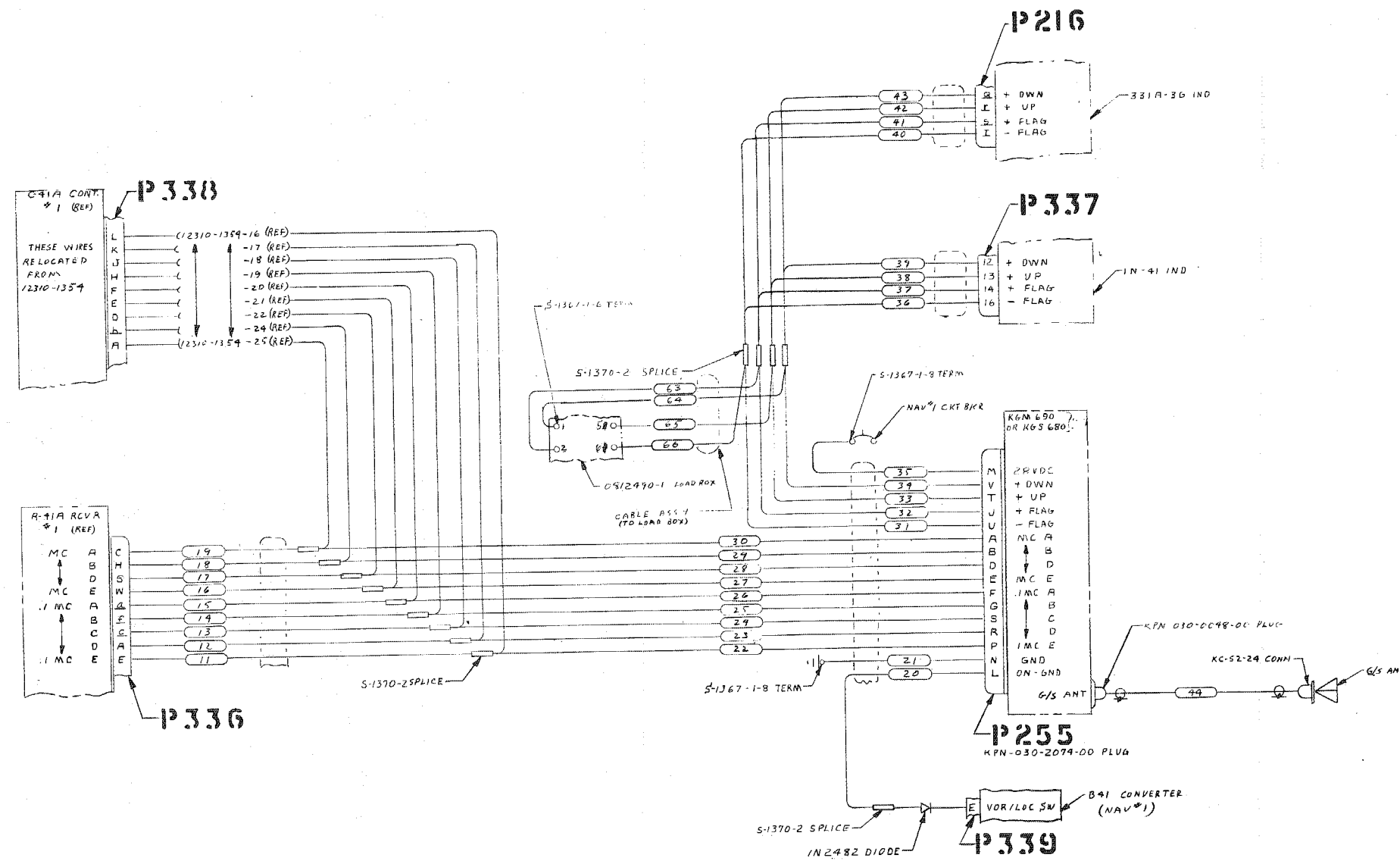


Figure 15-11. Cessna ADF 800 Rotatable Loop (Sheet 1 of 2)

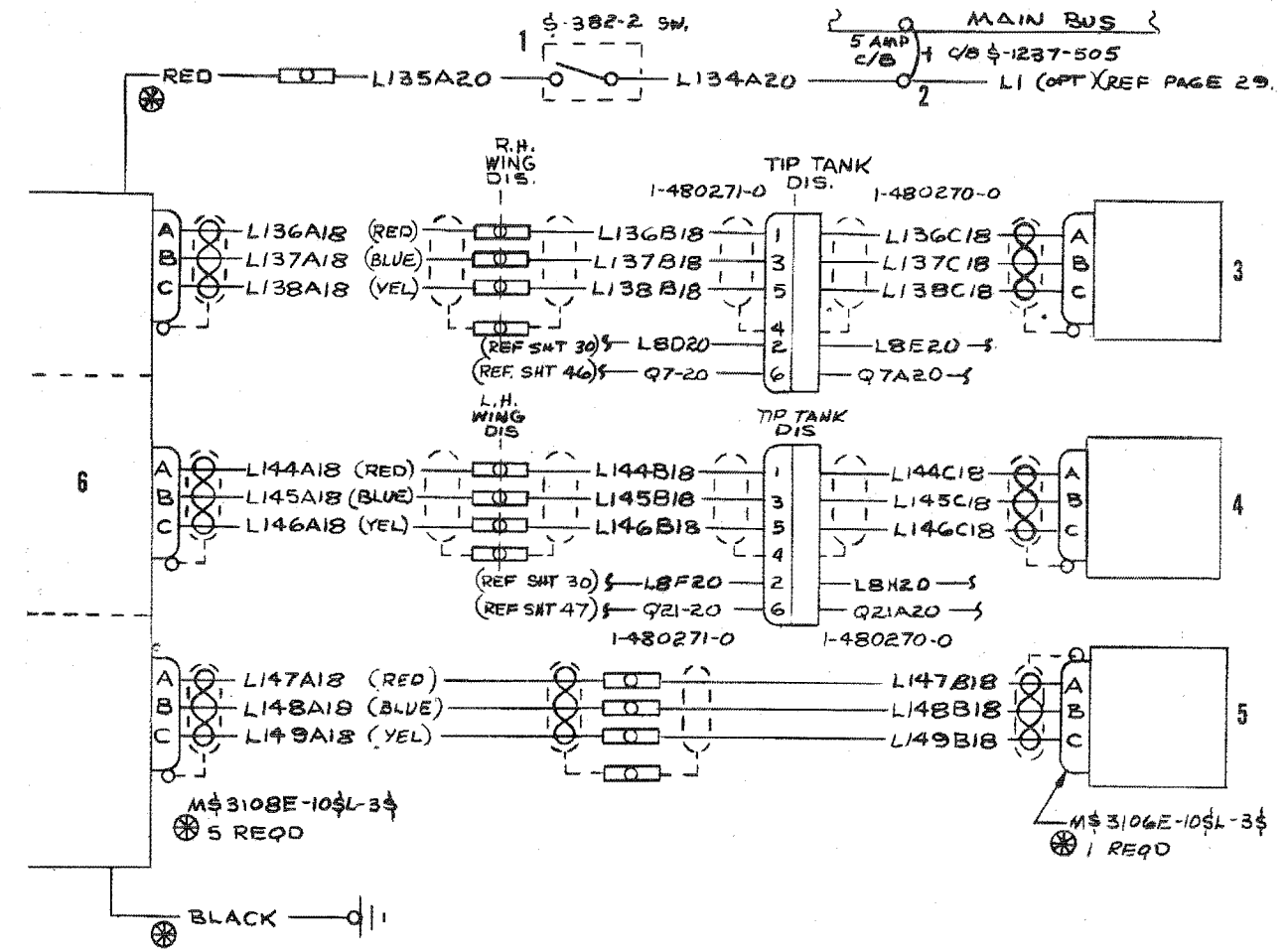


310P0015 AND ON
TURBO-SYSTEM 310P0015 AND ON

Figure 15-10. Cessna 800 Glide Slope

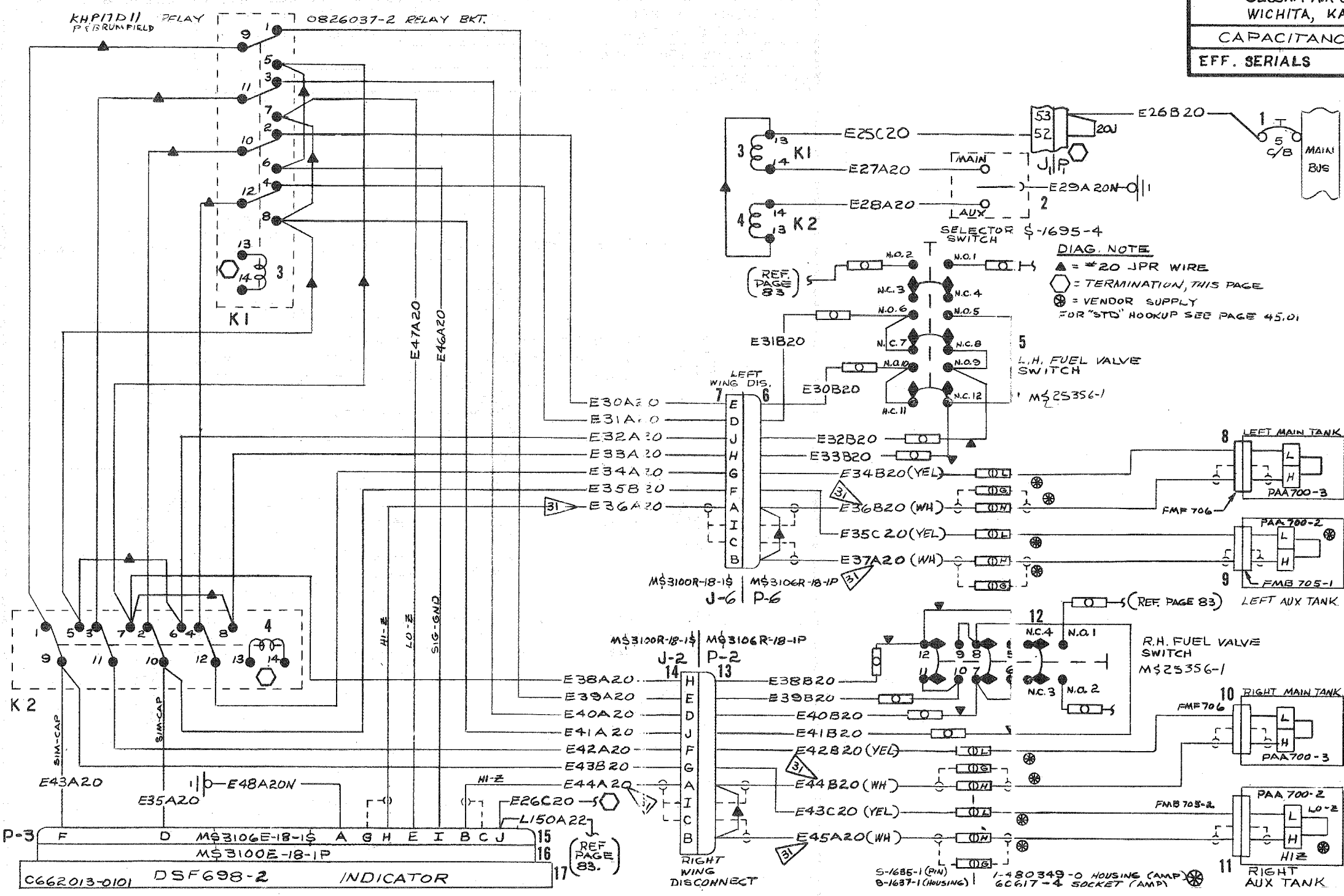
STROBE LIGHT SYSTEM (OPT)

EFF. SERIALS

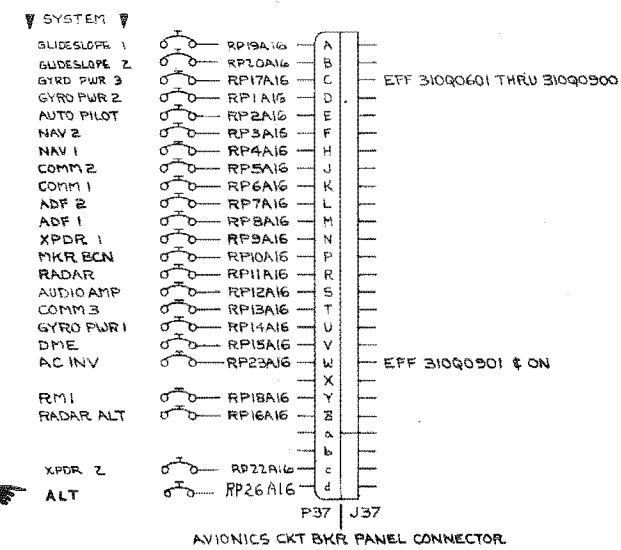
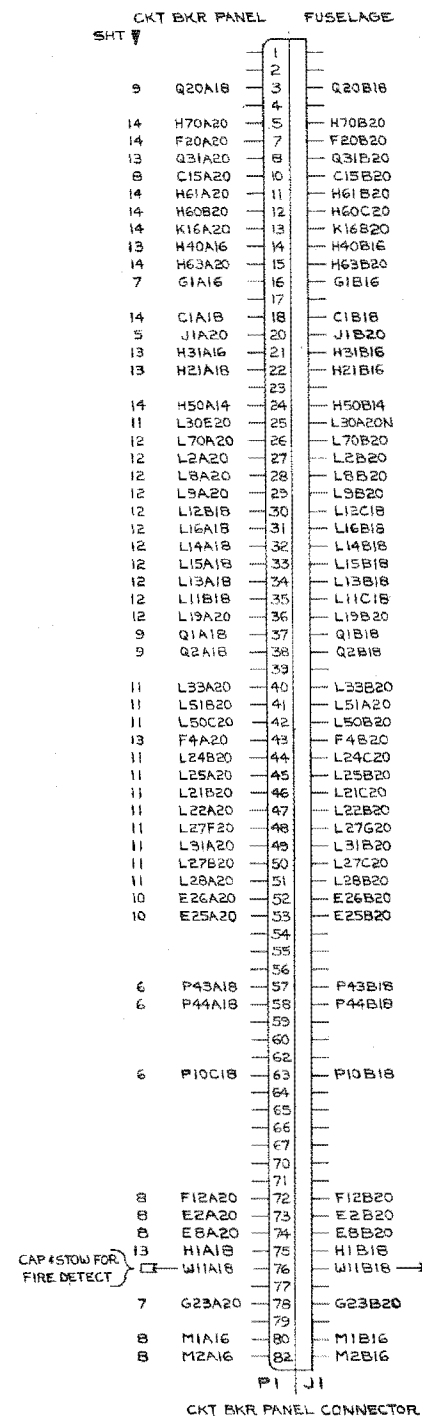


⊗ VENDOR FURNISHED

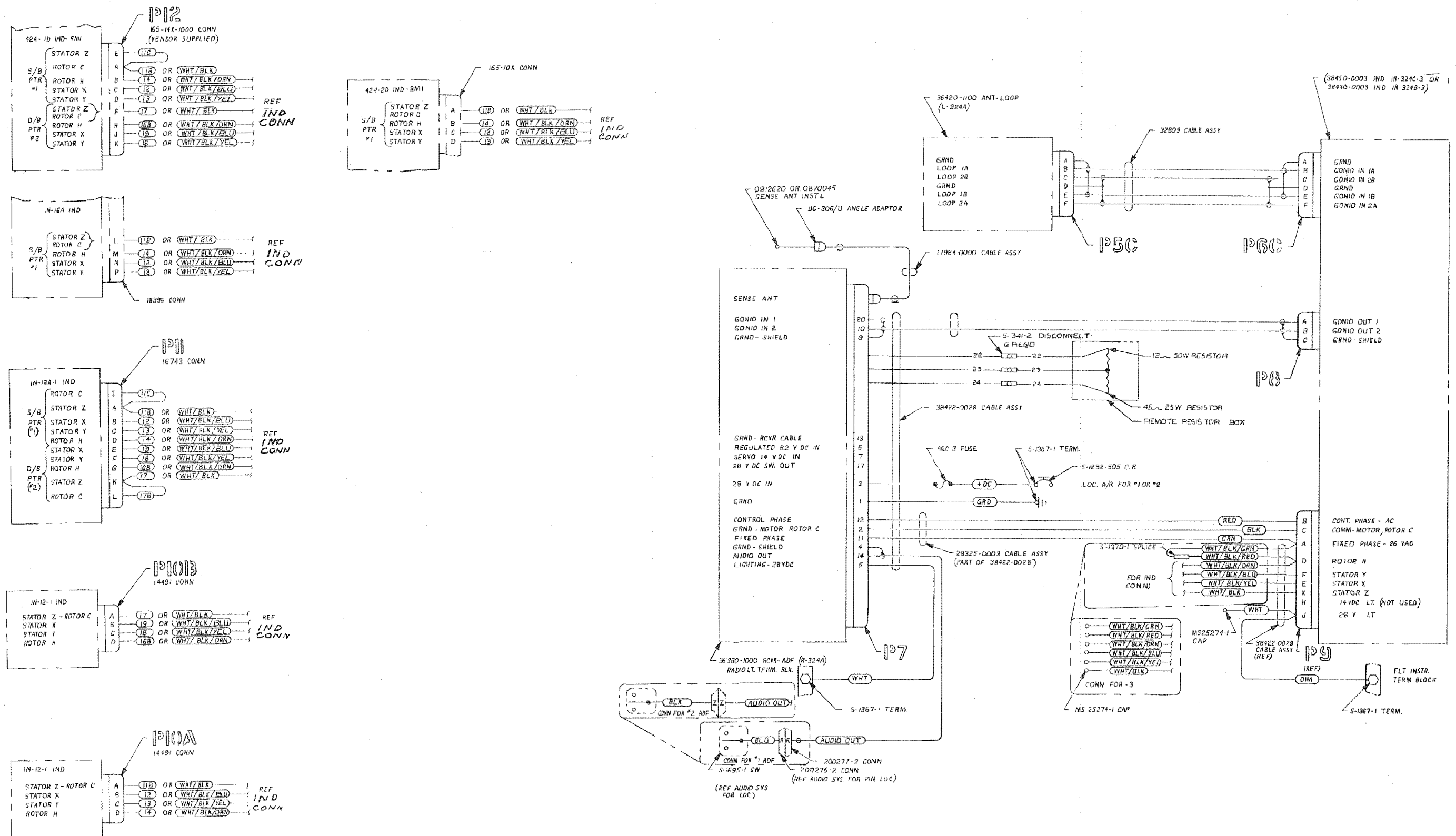
ITEM	PART NUMBER	NOMENCLATURE
1	S382-2	Switch
2	S1232-510	Circuit Breaker
3	30-0199-3LH	Light Assembly
4	30-0199-3RH	Light Assembly
5	30-0329-3	Light-Stinger
6	60-1038	Power Supply



ITEM	PART NUMBER	NOMENCLATURE
1	S1232-505	Circuit Breaker
2	S1695-4	Selector Switch
3	281XDX-24VDC	Relay
4	281XDX-24VDC	Relay
5	36-604	Selector Valve Switch LH
6	MS3106R-19-1P	Plug
7	MS3100R-18-1S	Receptacle
8	PAA700-3	Main Tank Unit LH
9	PAA700-2	Aux Tank Unit LH
10	PAA700-3	Main Tank Unit RH
11	PAA700-2	Aux Tank Unit RH
12	36-604	Selector Valve Switch RH
13	MS3106R-18-1P	Plug
14	MS3100R-18-1S	Receptacle
15	MS3106E-18-1S	Receptacle
16	MS3106E-18-1P	Plug
17	C662013-0101	Indicator

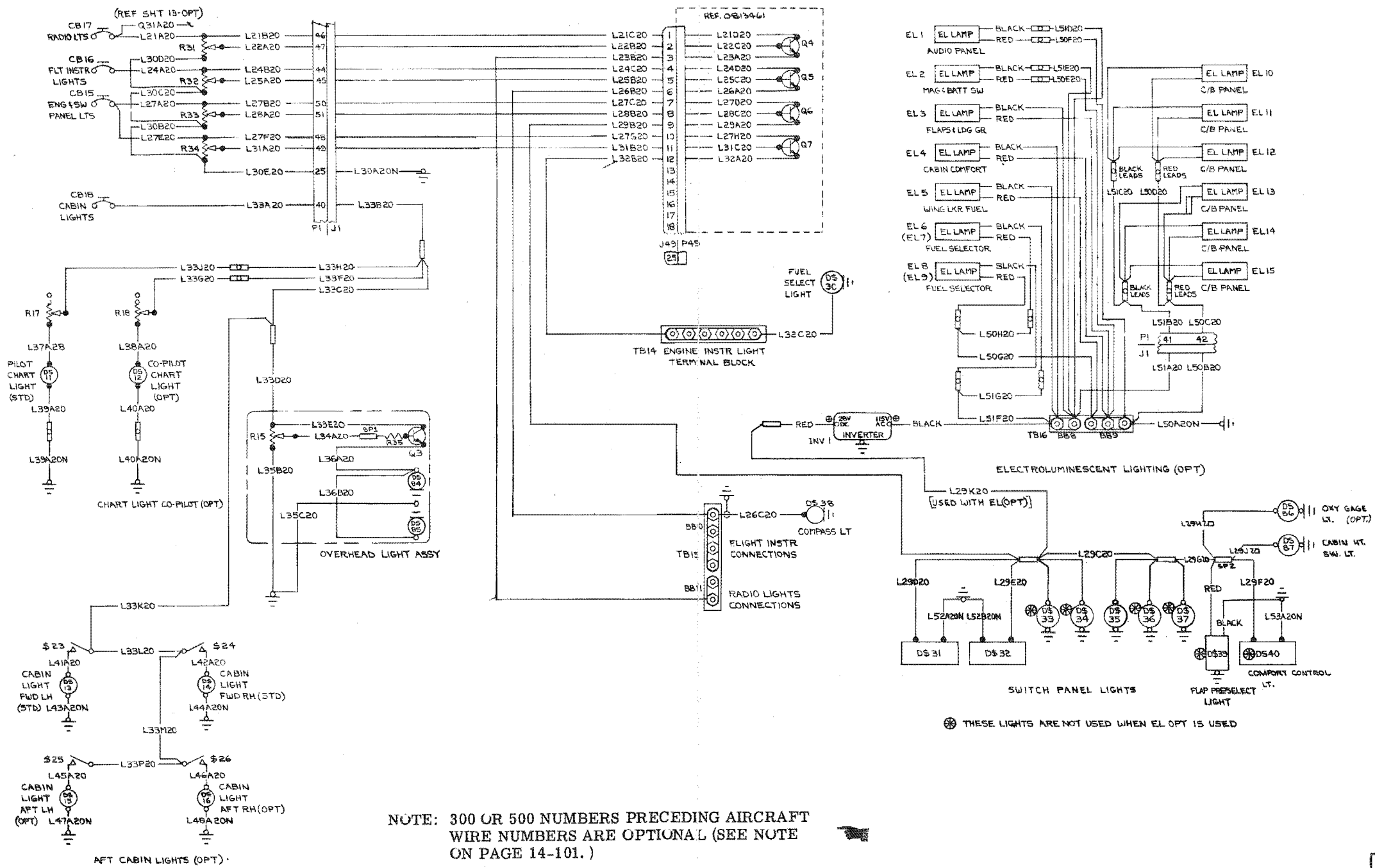


		P. O. BOX 1877 MILITARY & TWIN ENGINE DIVISION WICHITA, KANSAS 67201
TITLE: CONNECTORS		
SIZE: D	CODE IDENT NO: 71379	DRAWING NO: 0808080
SCALE: NONE	REF: G	SHEET: 3 OF 16

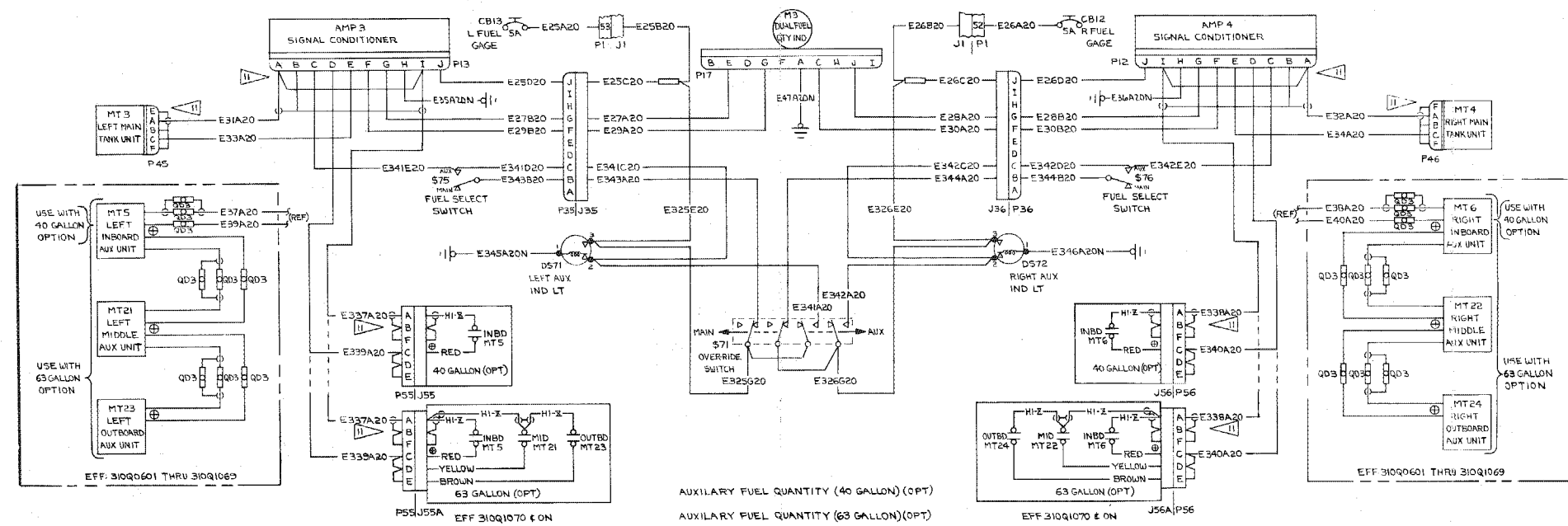
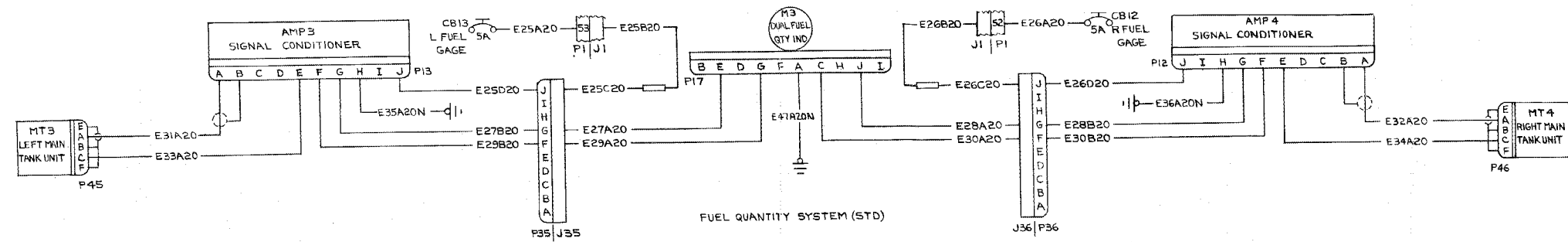


310P0015 AND ON
TURBO-SYSTEM 310P0015 AND ON

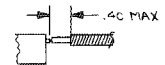
Figure 15-25. Cessna 400 ADF (Sheet 1 of 2)



Cessna		AIRCRAFT CO.		P.O. BOX 1877 MILWAUKEE & TRUCK DIVISION WICHITA, KANSAS 67201	
TITLE INTERIOR LIGHTING (OPT)					
SIZE D	CODE IDENT. NO. 71379	DRAWING NO. 0808080			
SCALE NONE	REF G	SHEET 1 OF 16			



ALL HI-Z SHIELDS SHOULD TERMINATE AT LESS THAN .40 IN FROM AFT END OF HI-Z PIN (AS SHOWN). ALL HI-Z JUMPER WIRES SHOULD REMAIN INSIDE THE AFT PLUG HOUSING, BUT THE SHIELD JUMPER Wires MAY BE LOOped EXTERNALLY



		AIRCRAFT CO. P.O. BOX 1977 MILITARY & TWIN DIVISION WICHITA, KANSAS 67201	
TITLE: FUEL QUANTITY SYSTEM - STD & OPT (CAPACITANCE)			
SIZE:	CODE IDENT. NO.:	DRAWING NO.:	
D	71379	0808080	
SCALE:	NONE	REP:	G
		SHEET 10 OF 16	

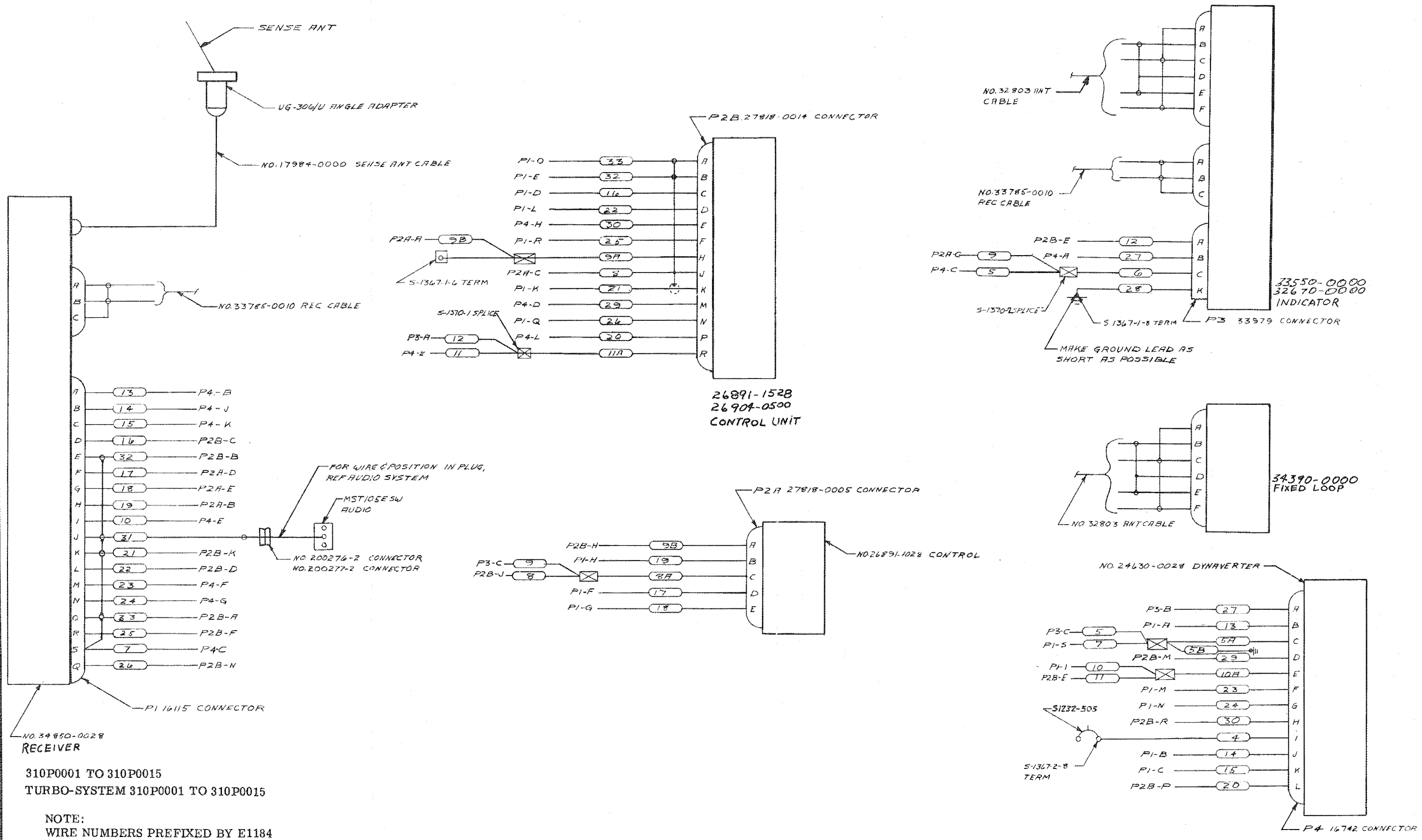


Figure 15-12. Cessna AD F 800

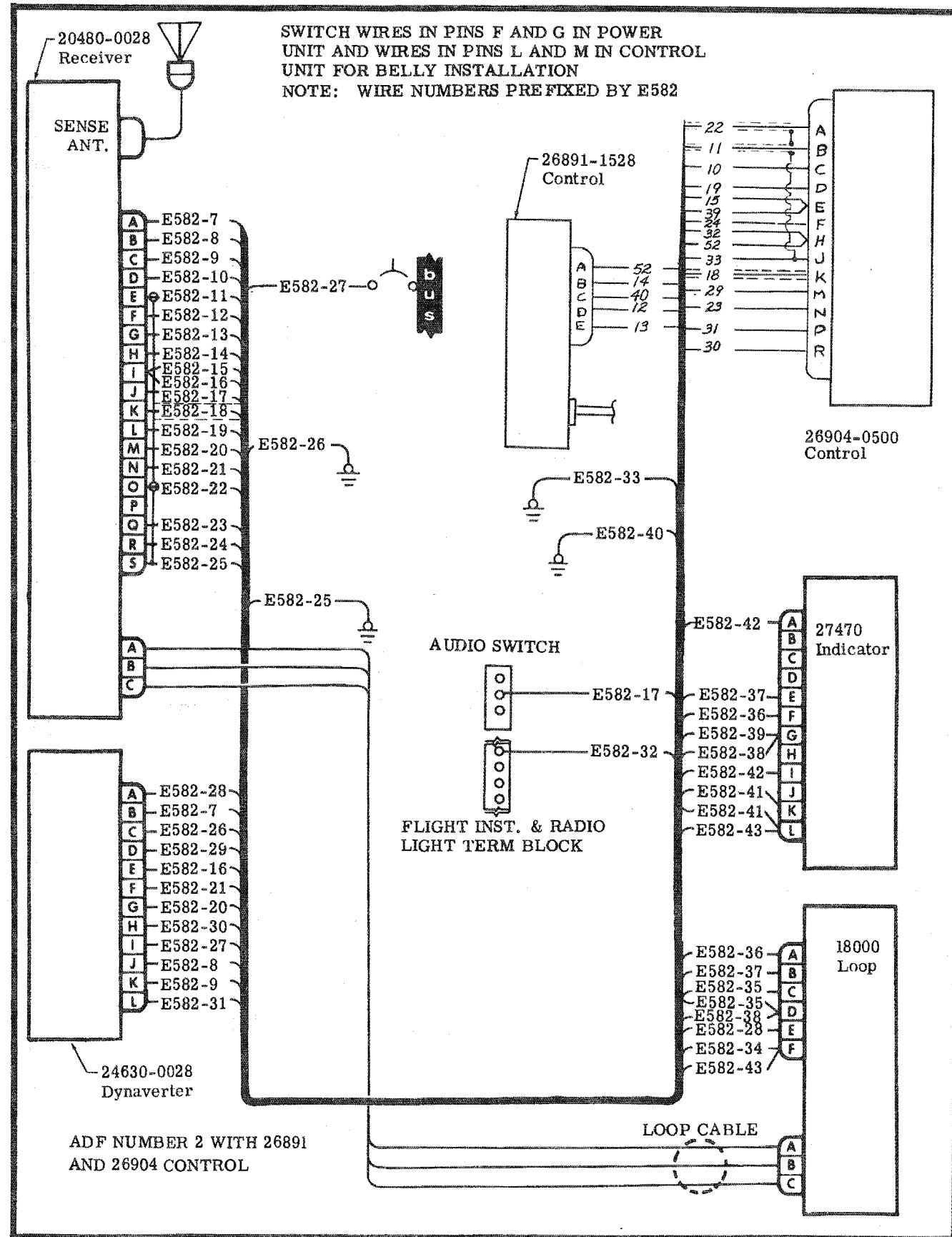


Figure 15-11. Cessna ADF 800 Rotatable Loop (Sheet 2 of 2)

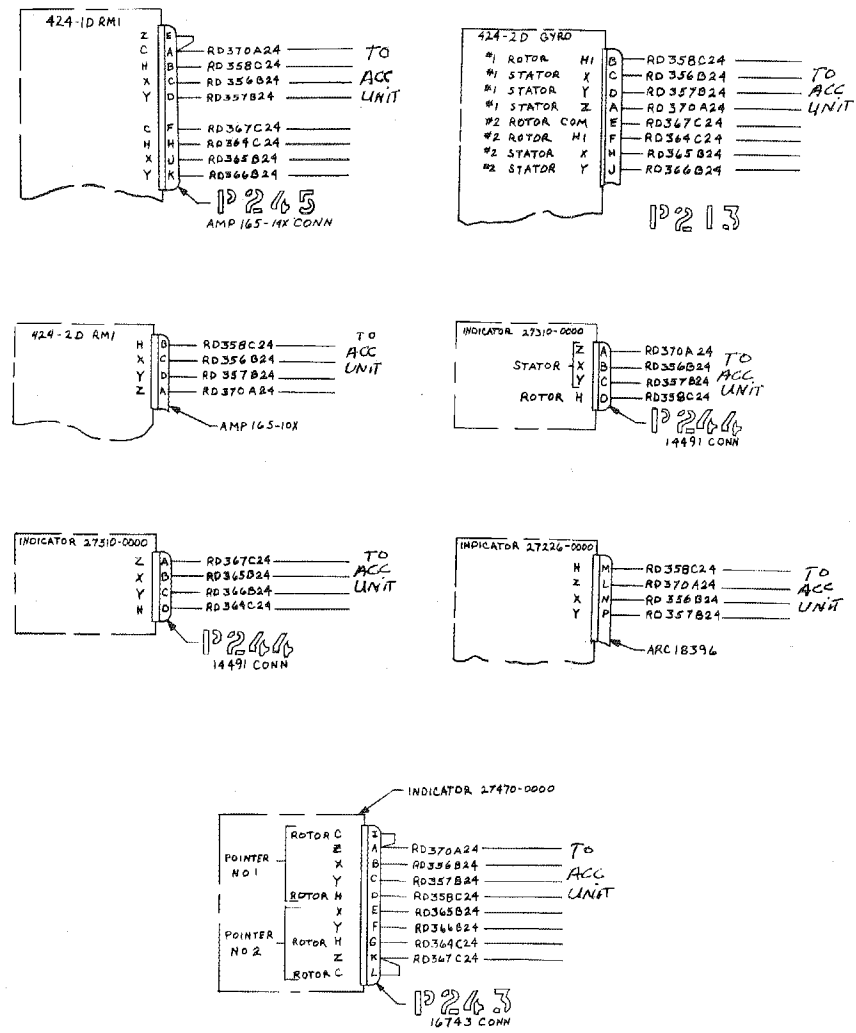
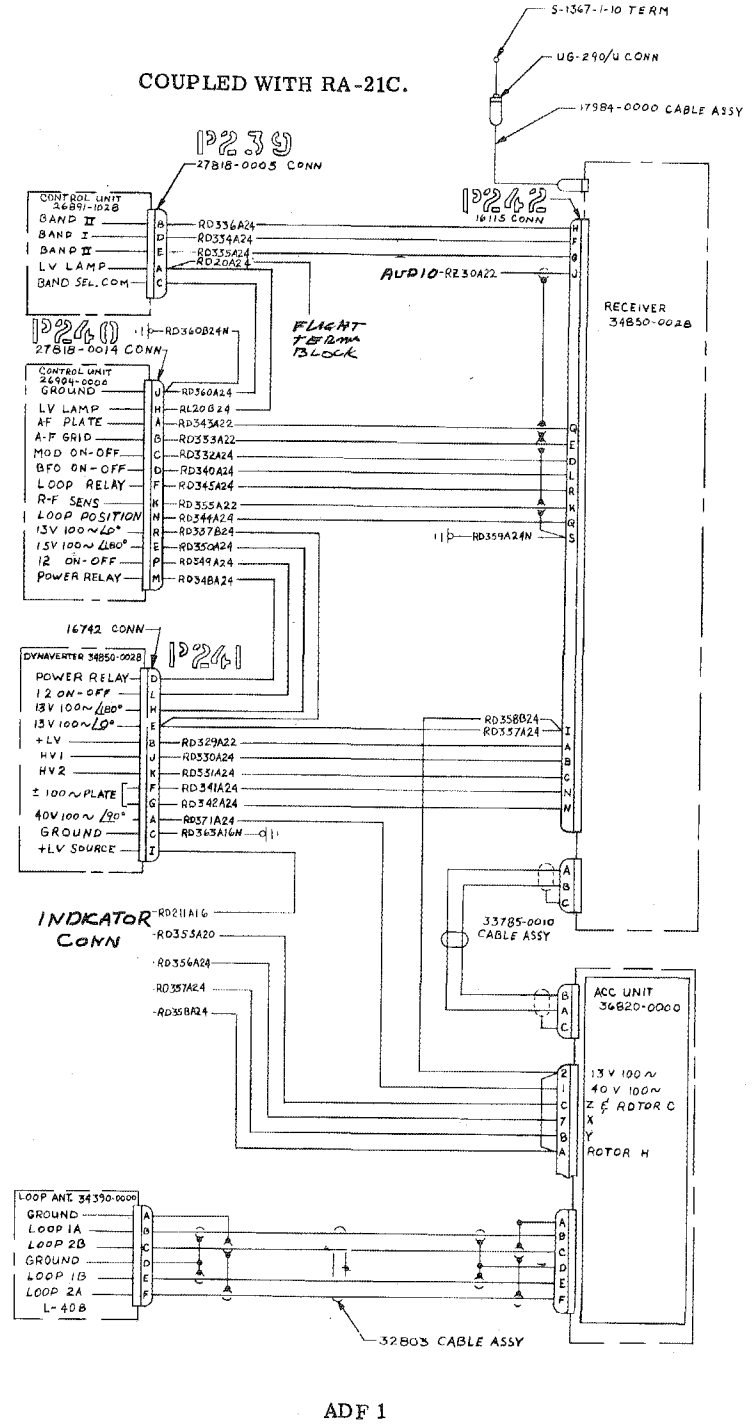
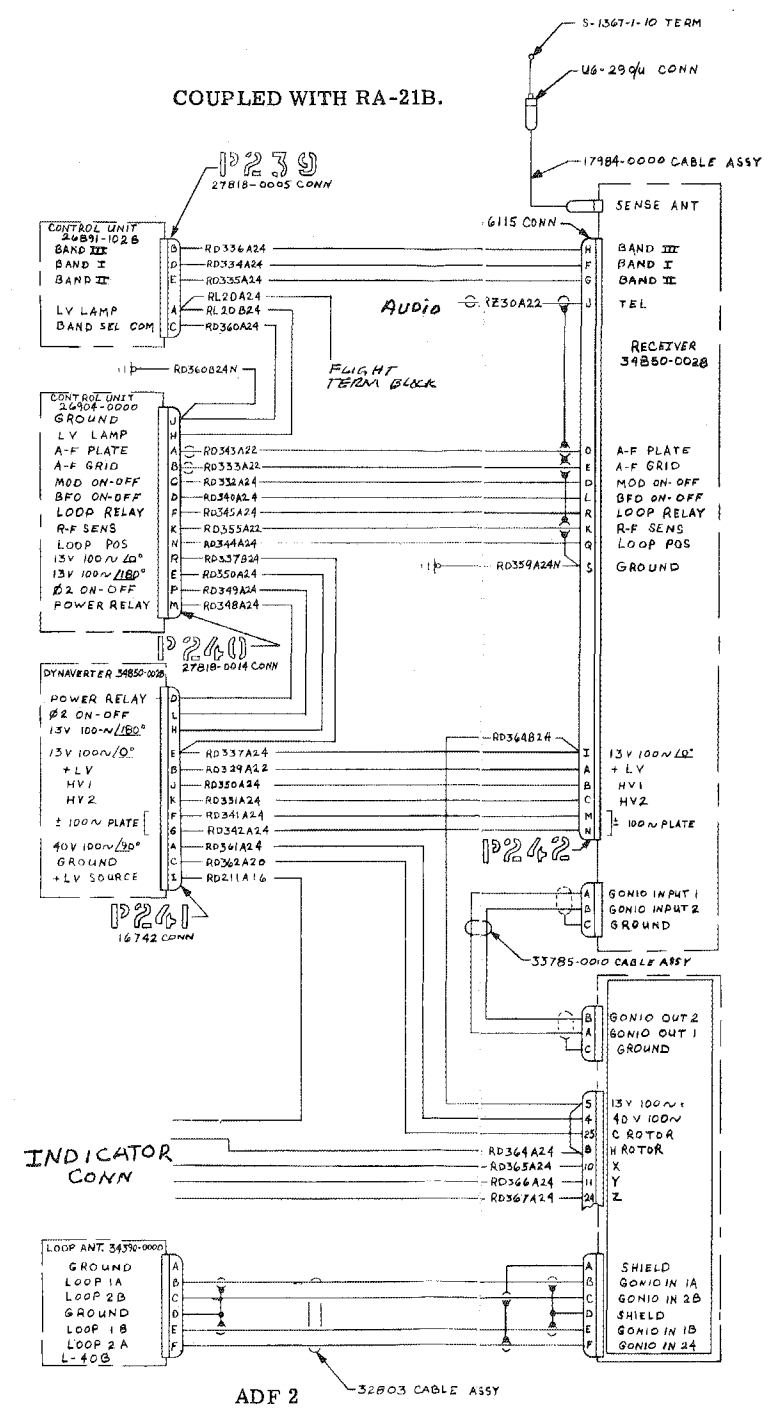


TABLE I TERM. CONNS. FOR ADF 1

WIRE COLOR	TOP LOOP	BOTTOM LOOP	
		TOP	BOT
BODY	TCR	SENSE	SENSE
GY	BLK	18	19
SRN	BLK	19	18
RED	BLK	22	22
BRN	BLK	23	23

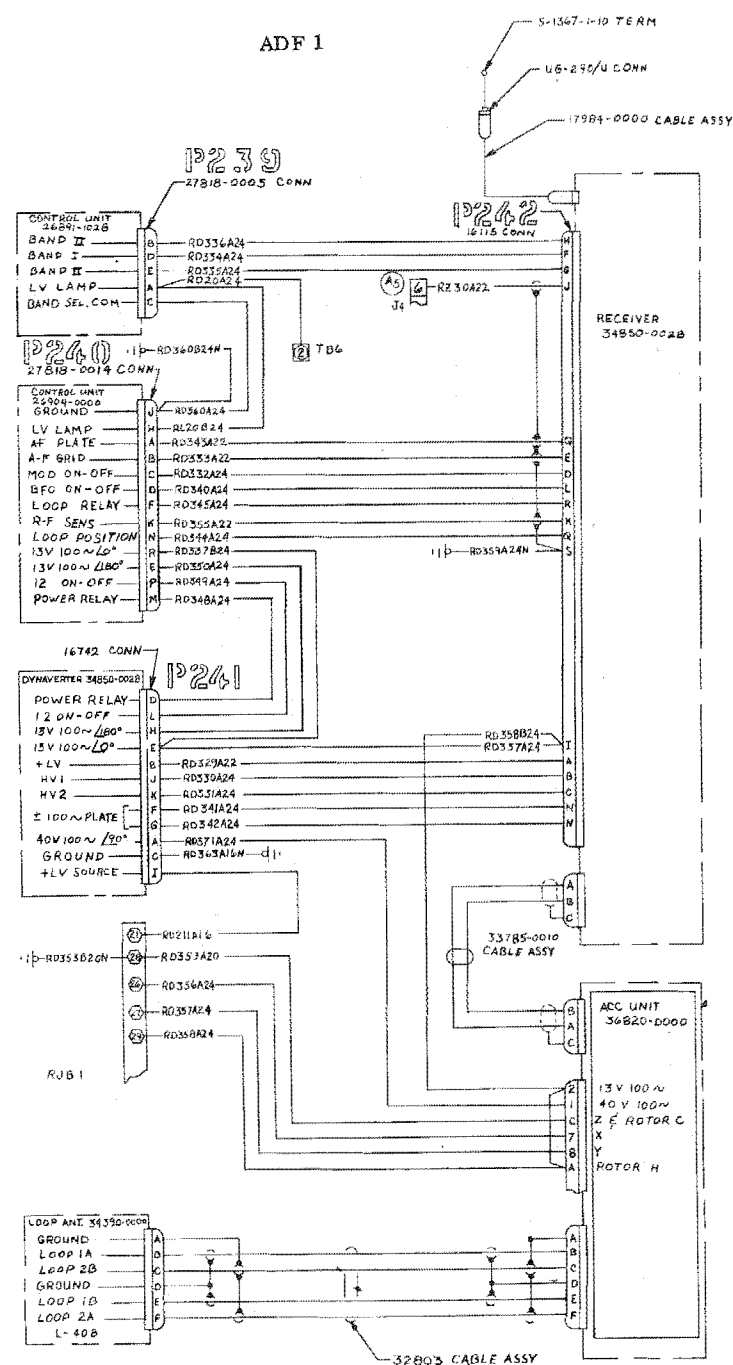
TABLE II TERM. CONNS. FOR ADF 2

WIRE COLOR	TOP LOOP	BOTTOM LOOP	
		TOP	BOT
BODY	TCR	SENSE	SENSE
GRN	BLK	16	17
EY	BLK	17	17
RED	BLK	12	13
BRN	BLK	13	12

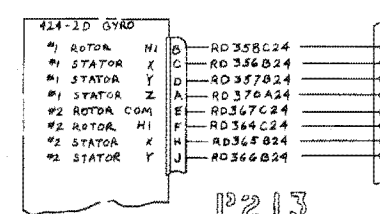


310P0015 AND ON
TURBO-SYSTEM 310P0015 AND ON

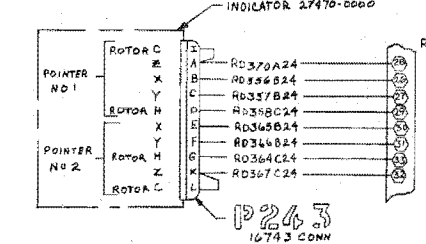
Figure 15-14. Cessna 800 Dual ADF



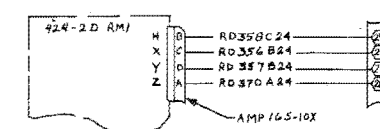
COUPLED WITH RA-21C.



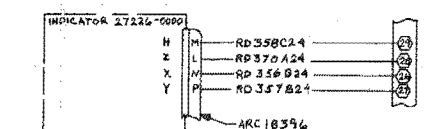
RUB1



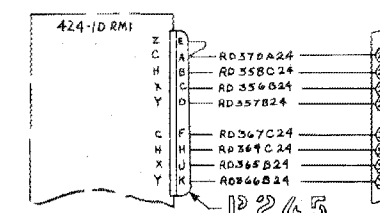
RUB1



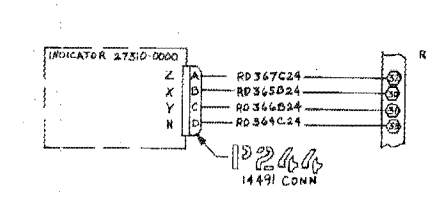
RUB1



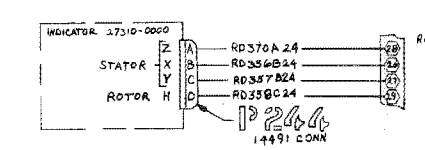
RUB1



RUB1



RUB1



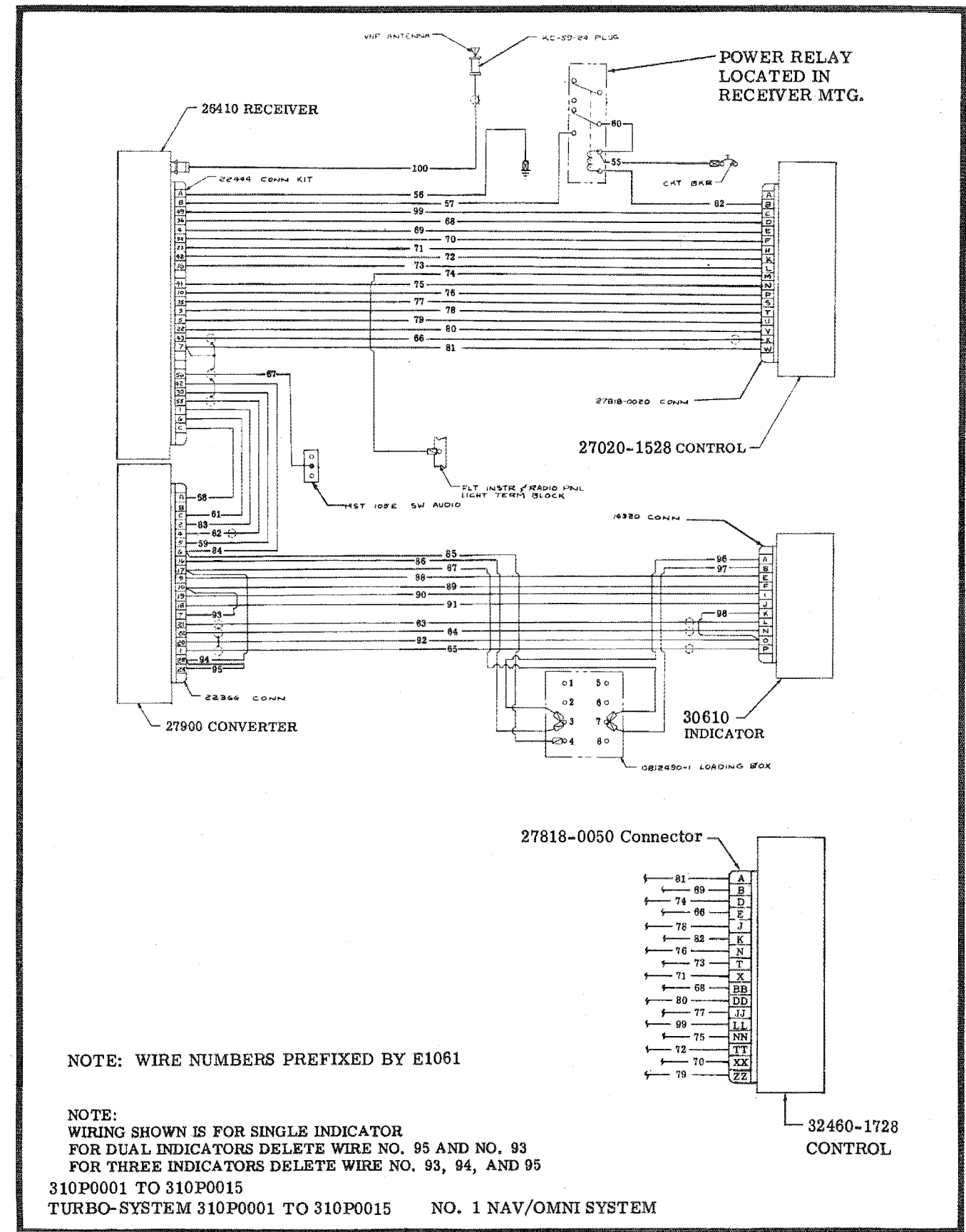
RUB1

TABLE I TERM CONNS FOR ADF1

WIRE COLOR	TOP LOOP		BOTTOM LOOP	
	TOP	SENSE	TOP	SENSE
GY	18	18	19	19
GRN	19	19	18	18
RED	22	23	23	22
BRN	23	22	22	23

310P0015 AND ON
TURBO-SYSTEM 310P0015 AND ON

Figure 15-13. Cessna 800 ADF

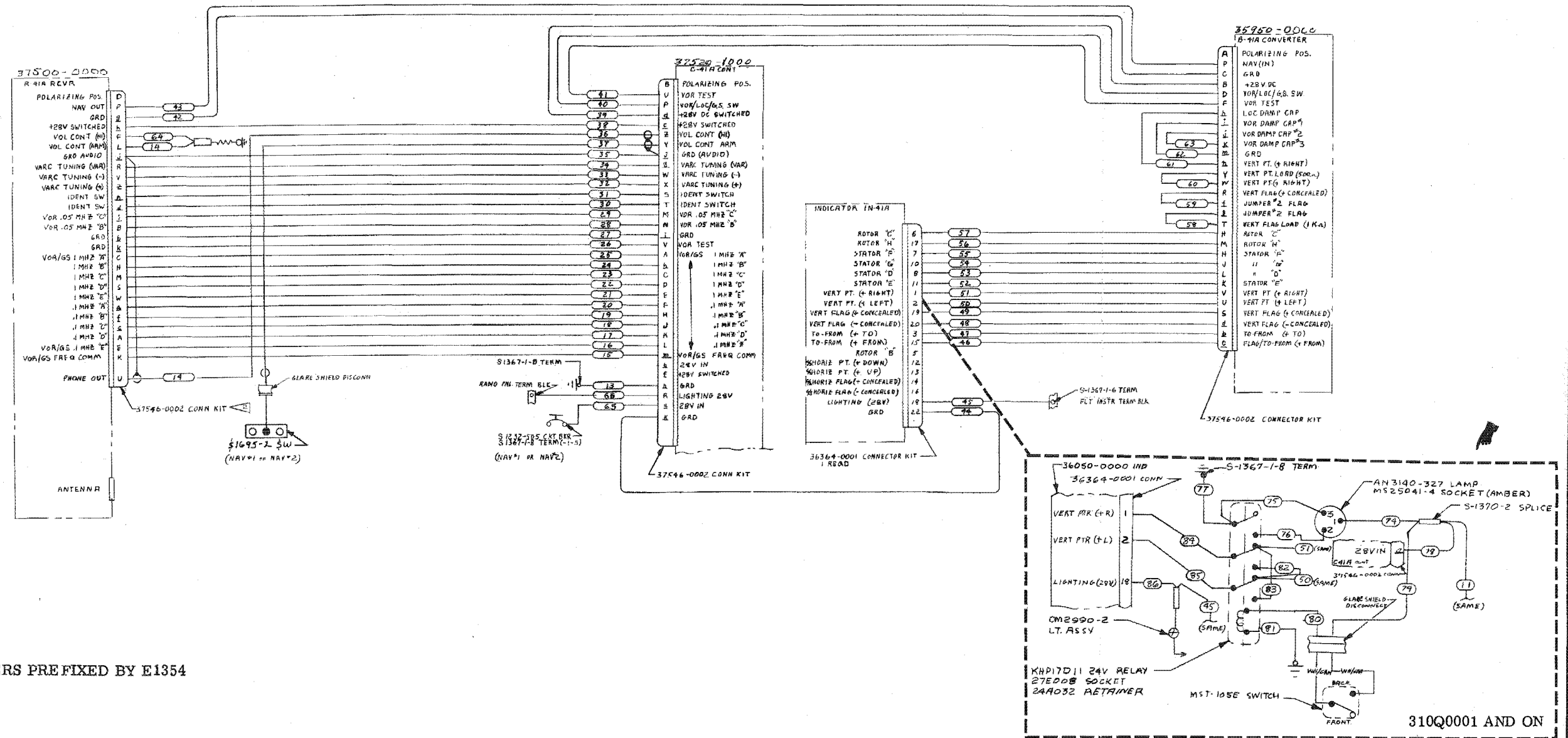


NOTE: WIRE NUMBERS PREFIXED BY E1061

NOTE:
 WIRING SHOWN IS FOR SINGLE INDICATOR
 FOR DUAL INDICATORS DELETE WIRE NO. 95 AND NO. 93
 FOR THREE INDICATORS DELETE WIRE NO. 93, 94, AND 95

310P0001 TO 310P0015
 TURBO-SYSTEM 310P0001 TO 310P0015 NO. 1 NAV/OMNI SYSTEM

Figure 15-8. Cessna Nav/Omni 800 (560 Channel) (Sheet 1 of 2)

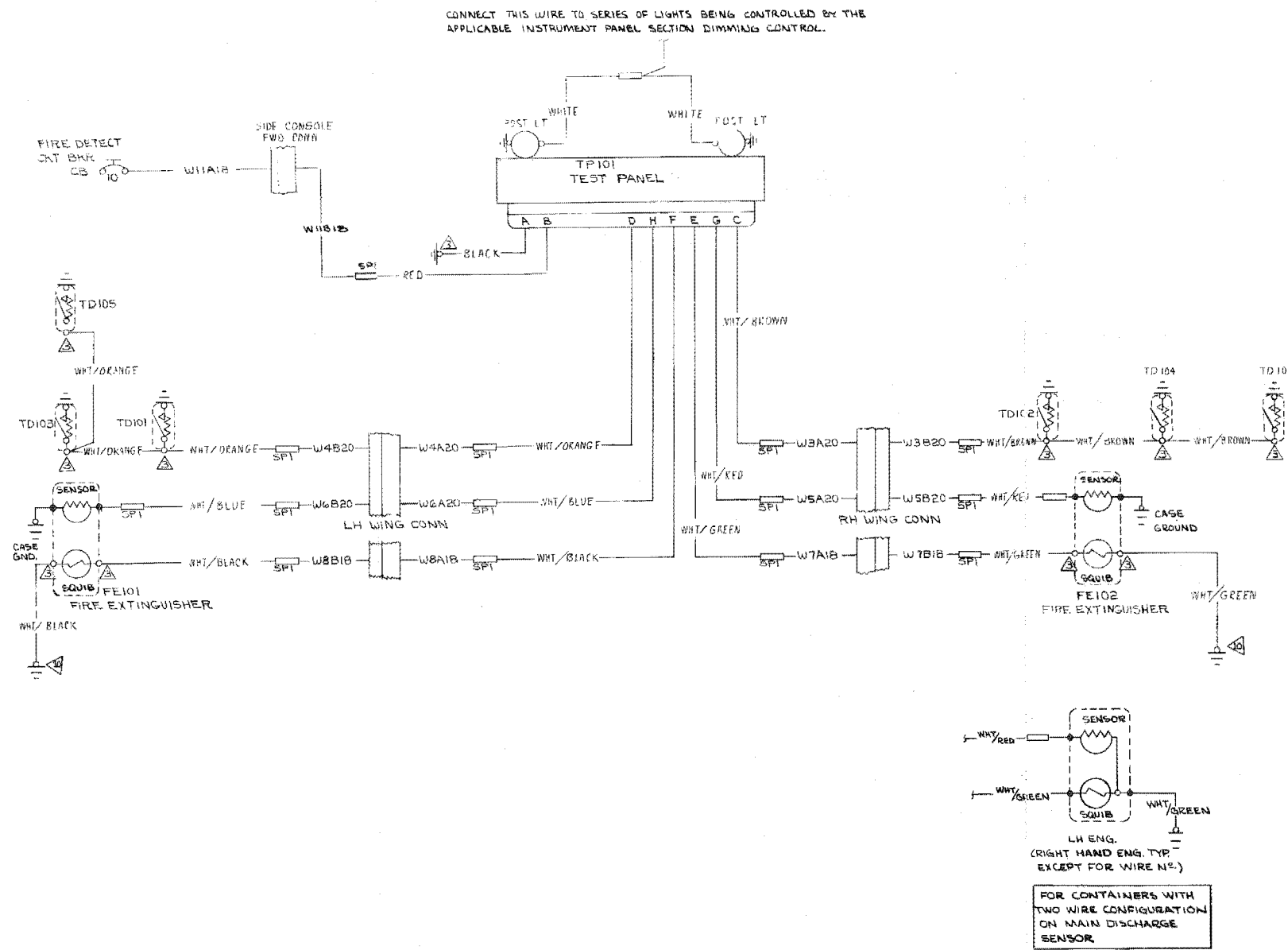


NOTE: WIRE NUMBERS PRE FIXED BY E1354

310P0015 AND ON
TURBO-SYSTEM 310P0015 AND ON

310Q0001 AND ON

Figure 15-7. Cessna Nav 800



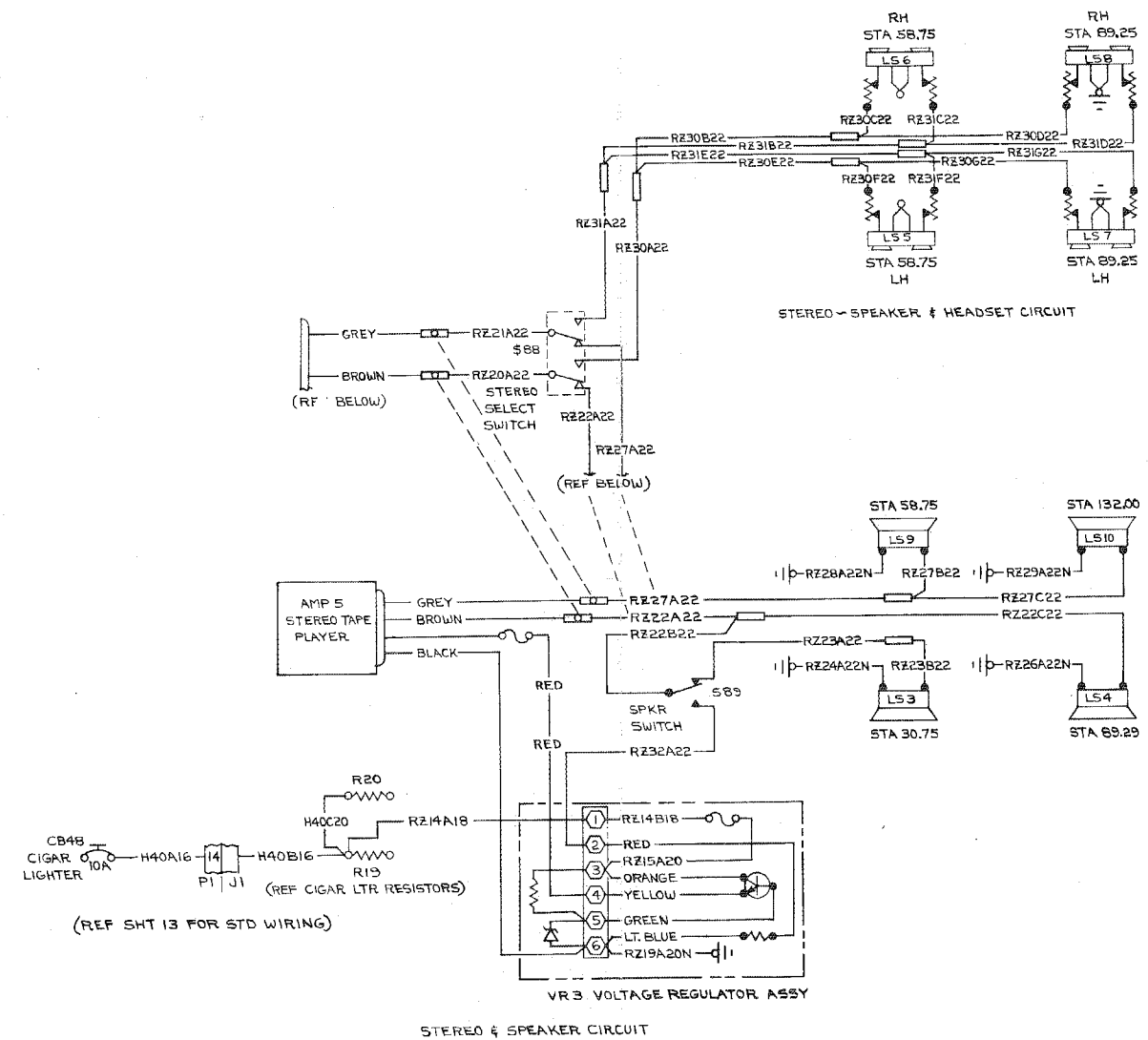
PARTS LIST		
SYMBOL	PART NO.	PART NAME
CB	7277-2-10	CIRCUIT BREAKER
FE101	3010101	FIRE EXTINGUISHER
FE102	3010101	FIRE EXTINGUISHER
SP1	54379-1	SPLICE
TD101	107-700	THERMAL DETECTOR
TD102	107-700	THERMAL DETECTOR
TD103	107-700	THERMAL DETECTOR
TD104	107-700	THERMAL DETECTOR
TD105	107-700	THERMAL DETECTOR
TD106	107-700	THERMAL DETECTOR
TP101	33370500	TEST PANEL

Cessna AIRCRAFT CO. P. O. BOX 1877
MILWAUKEE 8, WISCONSIN

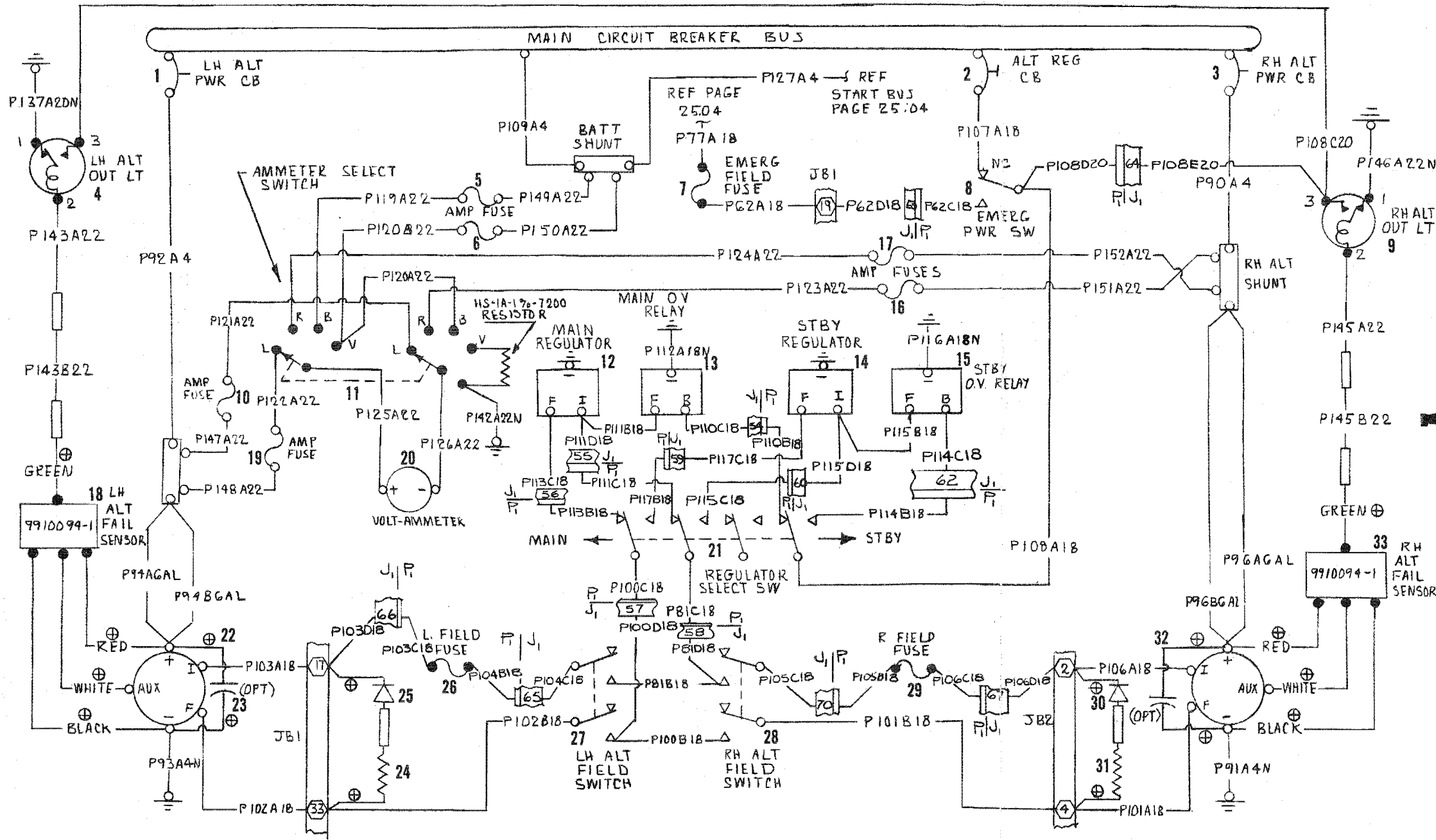
TITLE: FIRE DETECT SYSTEM WIRING ASSY

SIZE: D CODE IDENT NO. 5118461
REF: 33370525

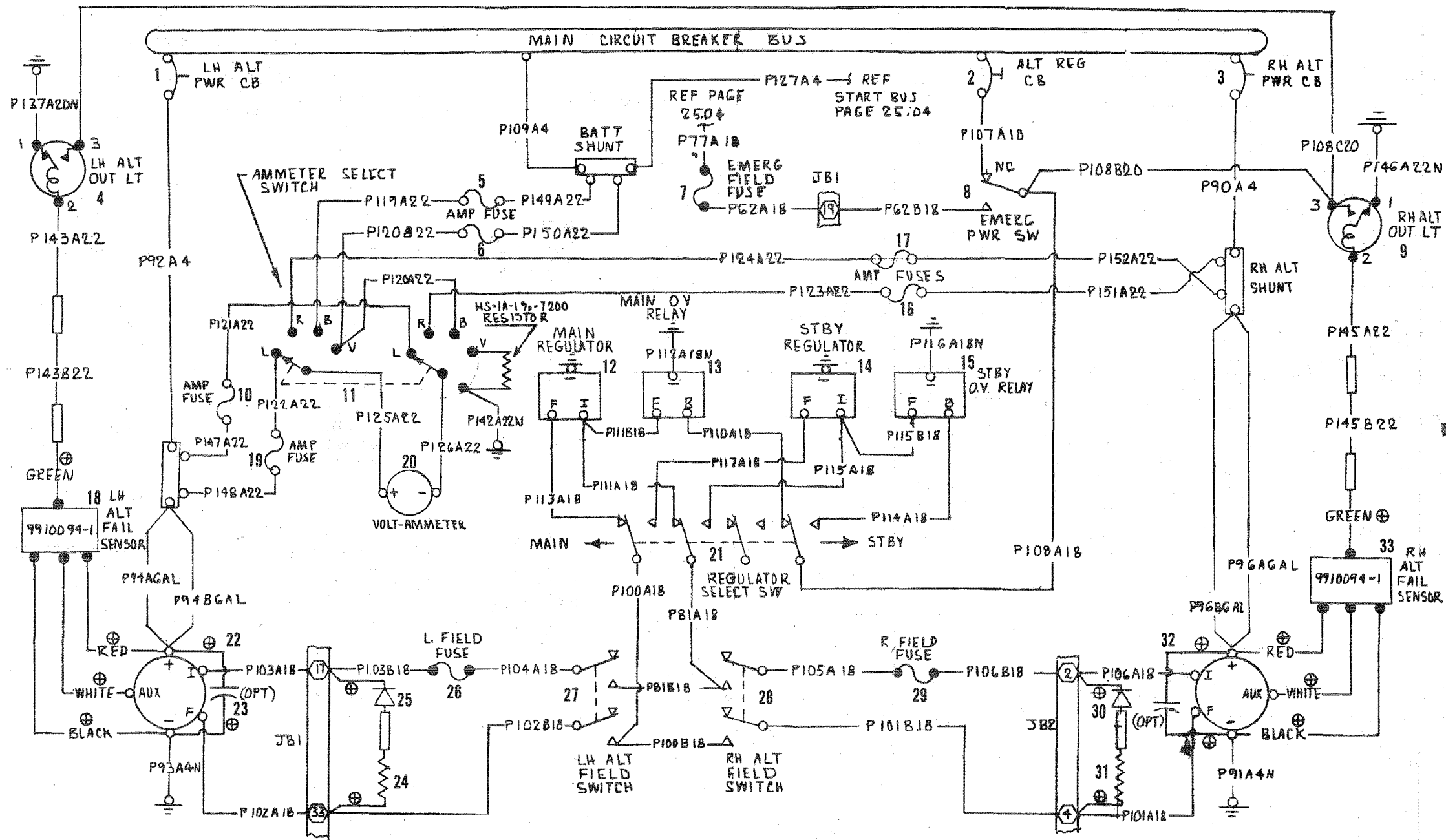
SCALE: NONE SHEET 1 OF 2



		AIRCRAFT CO.	P. O. BOX 1877 MILITARY & TWIN DIVISION WICHITA, KANSAS 67201
TITLE: STEREO - SPEAKERS & HEADSETS			
SIZE: D	CODE IDENT. NO. 71379	DRAWING NO. 0808080	
SCALE: NONE	REF:	SHEET 16 OF 16	



ITEM	PART NUMBER	NOMENCLATURE
1	45-2-S-N2 100 AMP	Circuit Breaker
2	S1232-510	Circuit Breaker
3	45-2-S-N2 100 AMP	Circuit Breaker
4	VM911M4	Alternator Out Lt LH
5	311005	Fuse
6	311005	Fuse
7	MTH-5	Fuse
8	S-392-3	Sw - Emer Power
9	VM911M4	Alternator Out Lt RH
10	311005	Fuse
11	0870176-1	Sw - Ammeter Select
12	635837	Voltage Regulator - Mn
13	138-3	Mn Overvoltage Relay
14	635837	Voltage Regulator - Stby
15	138-3	Stby Overvoltage Relay
16	311005	Fuse
17	311005	Fuse
18	991004-1	Alternate Fail Sensor LH
19	311005	Fuse
20	CM2664	Volt Ammeter
21	MS25127E3	Sw - Regulator Select
22	634692 634445 (Turbo)	Alternator LH
23		Capacitor
24	1K-1W	Resistor
25	IN4720	Diode
26	AGC2.0	Fuse LH Field
27	MS35059-22	Sw - LH Alternator Field
28	MS35059-22	Sw - RH Alternator Field
29	AGC2.0	Fuse - RH Field
30	IN4720	Diode
31	1K-1W	Resistor
32	63492 634445 (Turbo)	Alternator RH
33	991004-1	Alternator Fail Sensor RH



ITEM	PART NUMBER	NOMENCLATURE
1	45-2-S-N2 100 AMP	Circuit Breaker
2	S1232-510	Circuit Breaker
3	45-2-S-N2 100 AMP	Circuit Breaker
4	VM911M4	Alternator Out Lt LH
5	311005	Fuse
6	311005	Fuse
7	MTH-5	Fuse
8	S-392-3	Sw - Emer Power
9	VM911M4	Alternator Out Lt RH
10	311005	Fuse
11	0870176-1	Sw - Ammeter Select
12	635837	Voltage Regulator - Mn
13	138-3	Mn Overvoltage Relay
14	635837	Voltage Regulator Stby
15	138-3	Stby Overvoltage Relay
16	311005	Fuse
17	311005	Fuse
18	991004-1	Alternate Fail Sensor LH
19	311005	Fuse
20	CM2664	Volt Ammeter
21	MS25127E3	Sw - Regulator Select
22	634692 634445 (Turbo)	Alternator LH
23		Capacitor
24	1K-1W	Resistor
25	IN4720	Diode
26	AGC 2.0	Fuse LH Field
27	MS35059-22	Sw - LH Alternator Field
28	MS35059-22	Sw - RH Alternator Field
29	AGC 2.0	Fuse - RH Field
30	IN4720	Diode
31	1K-1W	Resistor
32	63492 634445 (Turbo)	Alternator RH
33	991004-1	Alternator Fail Sensor RH

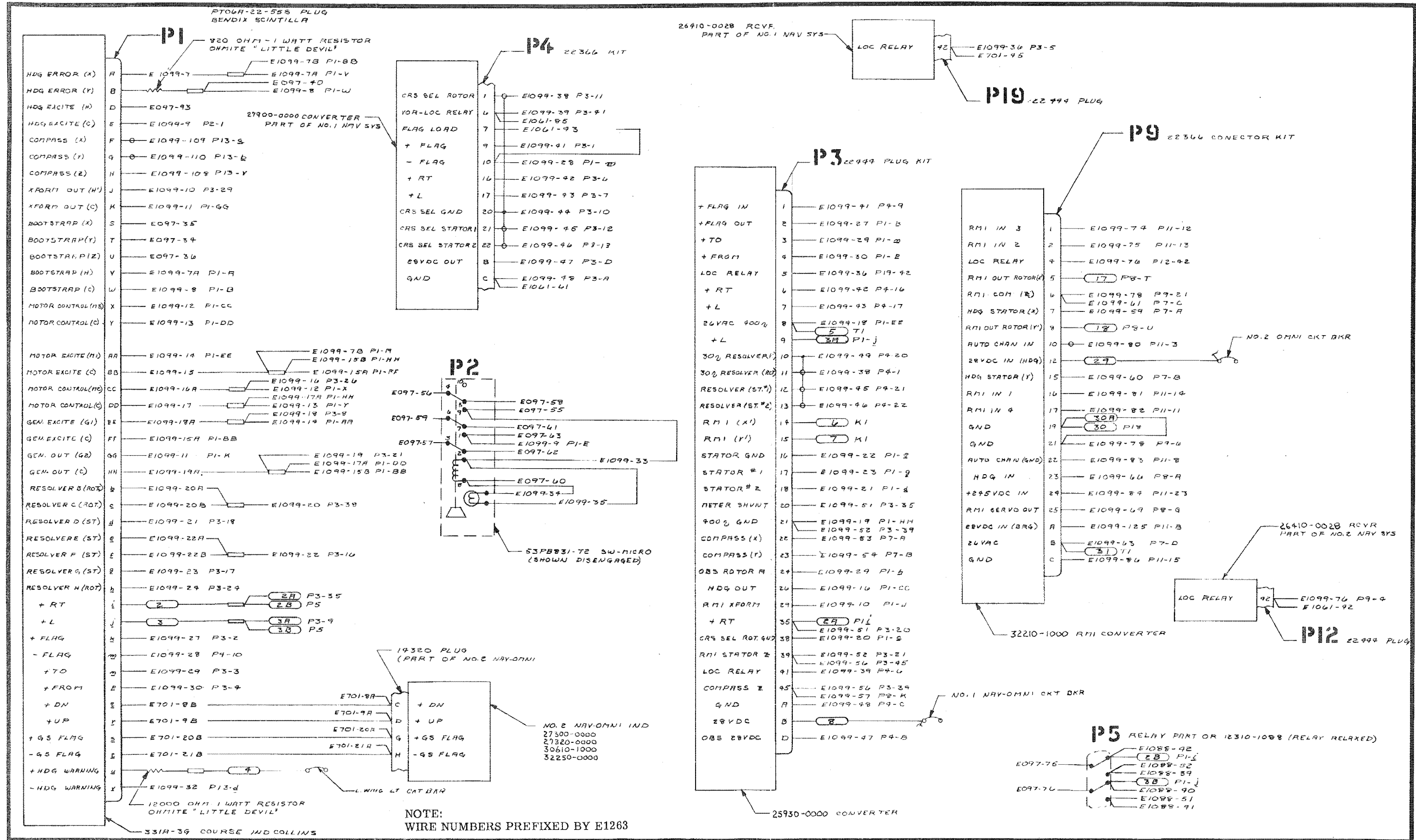
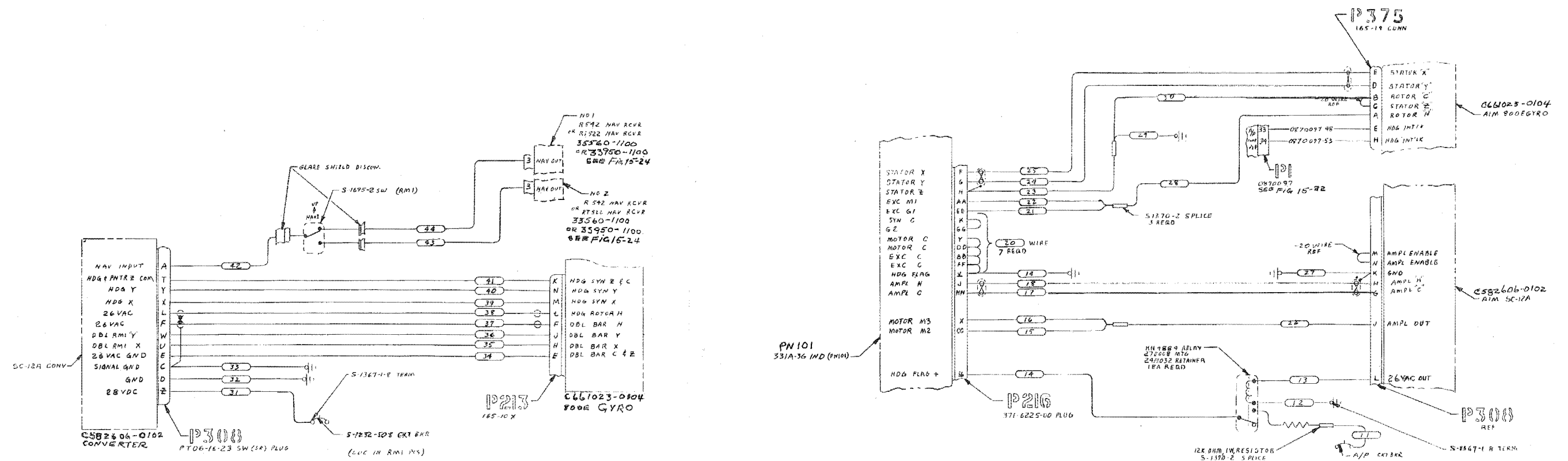
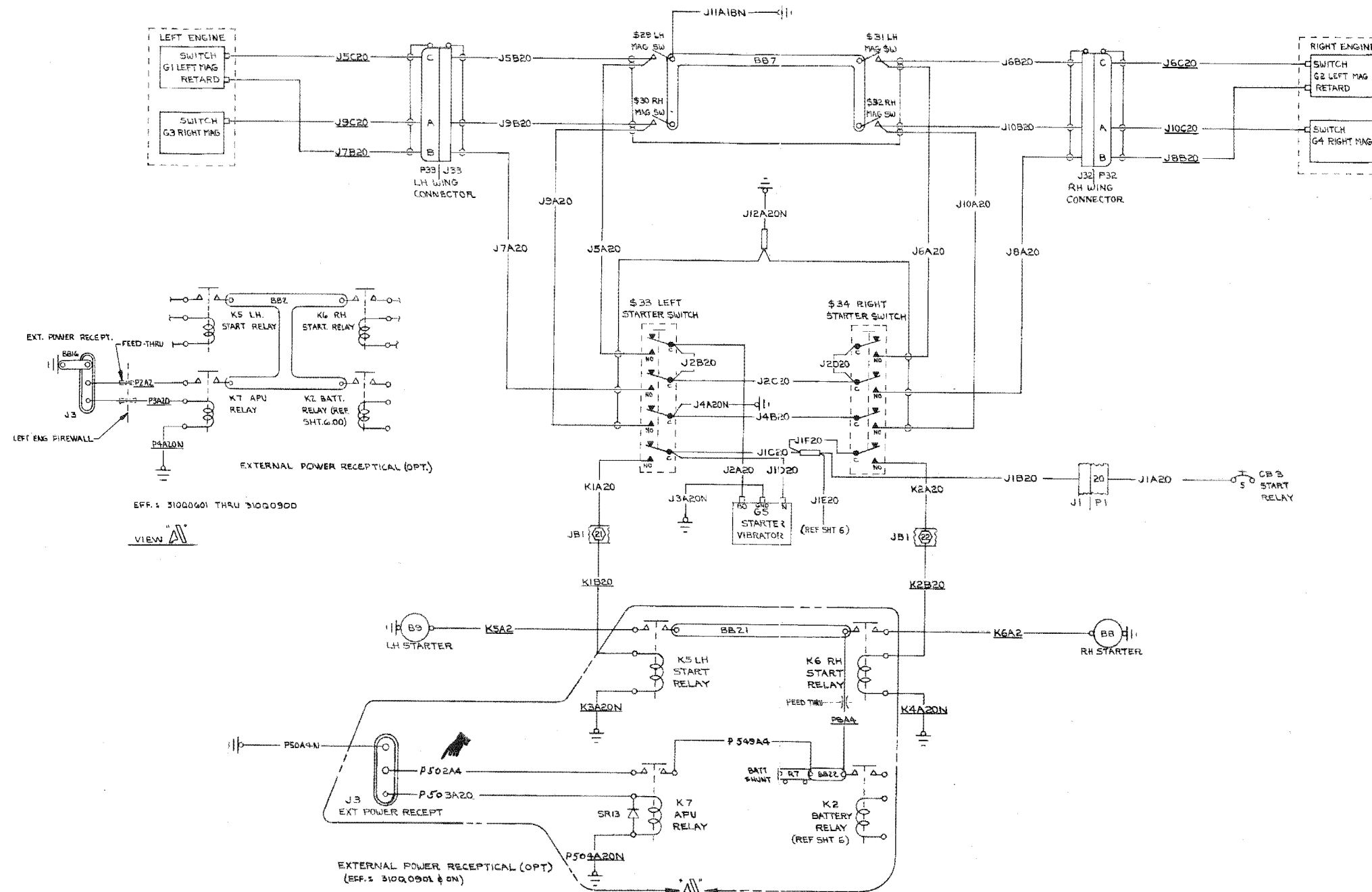


Figure 15-21. PN101 NAV System Coupled with C-14 Gyro and Cessna 800 RMI (Sheet 1 of 2)

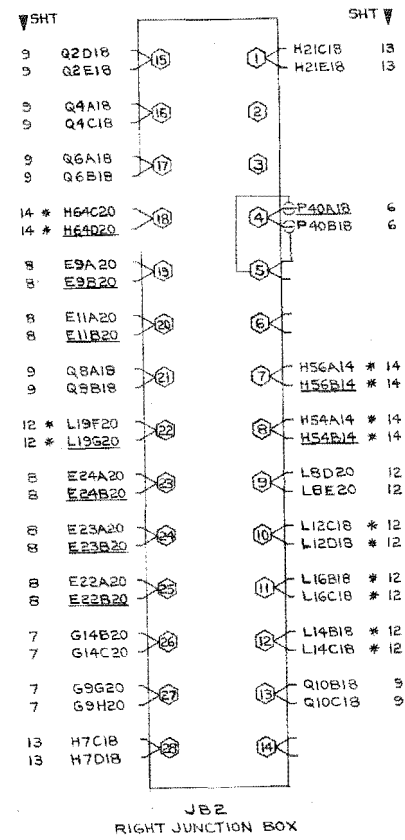
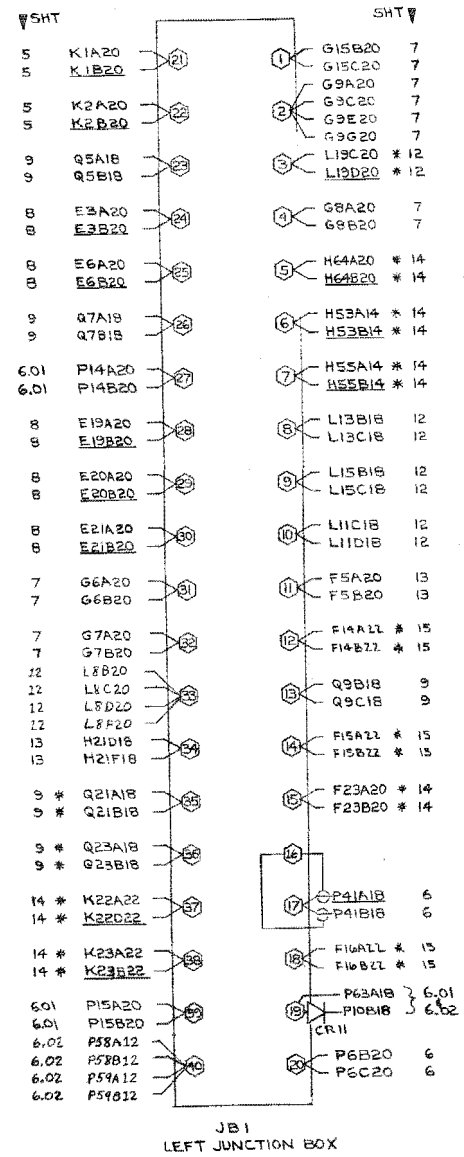


NOTE: WIRE NUMBERS PREFIXED BY E1360

Figure 15-20. Cessna 800 RMI

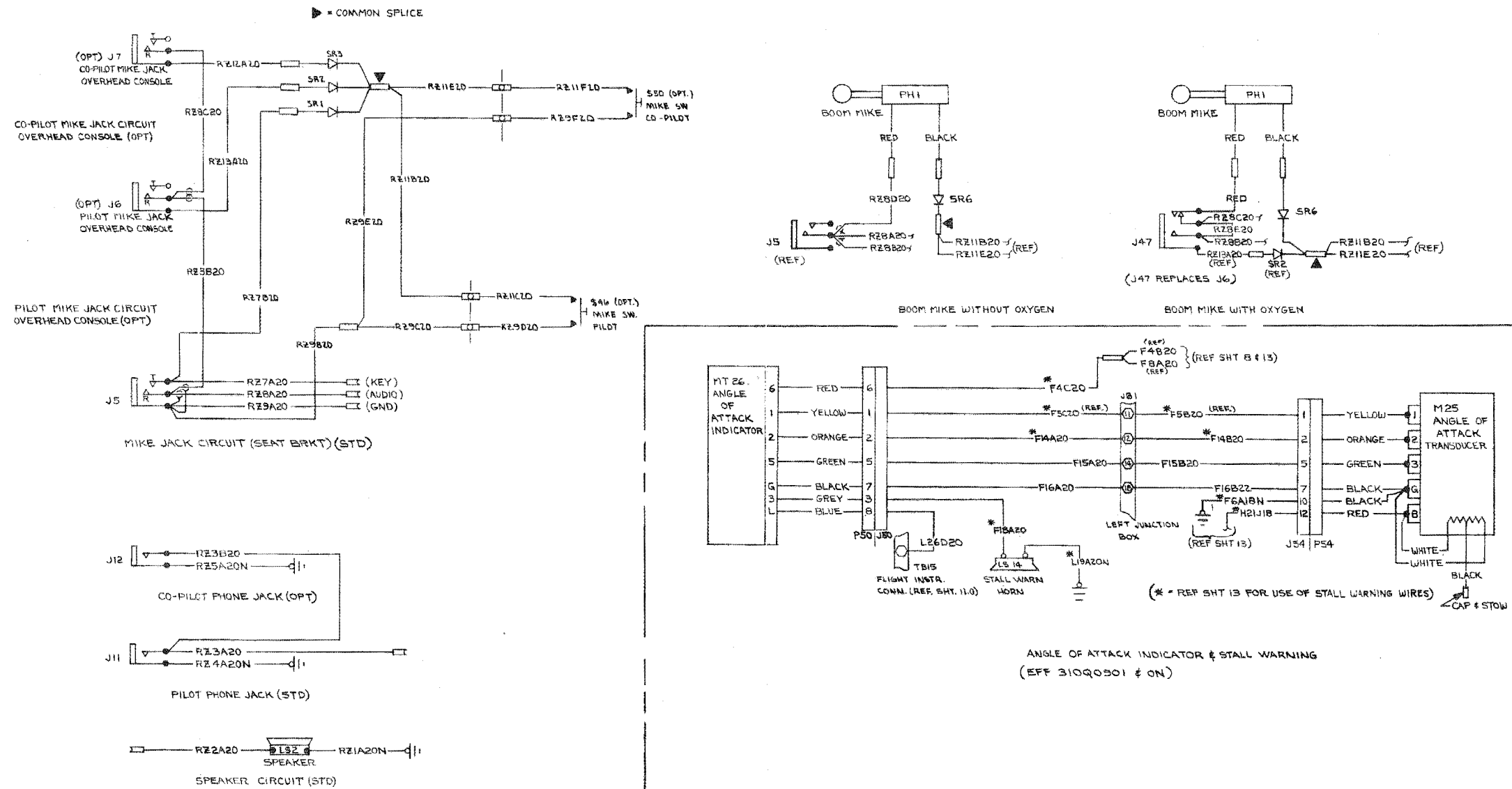


Cessna AIRCRAFT CO. P. O. BOX 1977 MILITARY & TWIN DIVISION WICHITA, KANSAS 67201	
TITLE STARTER, IGNITION & APU (OPT)	
SIZE	DRAWING NO.
D	0808080
SCALE: NONE	SHEET 5 OF 16



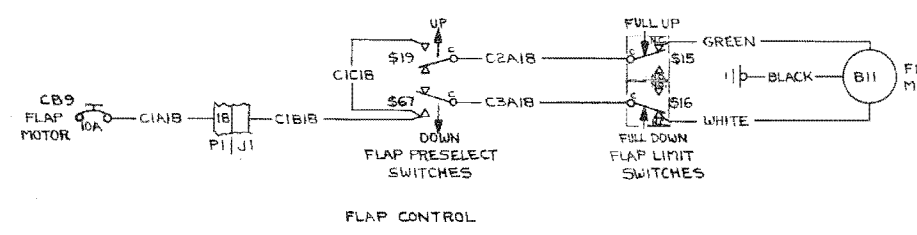
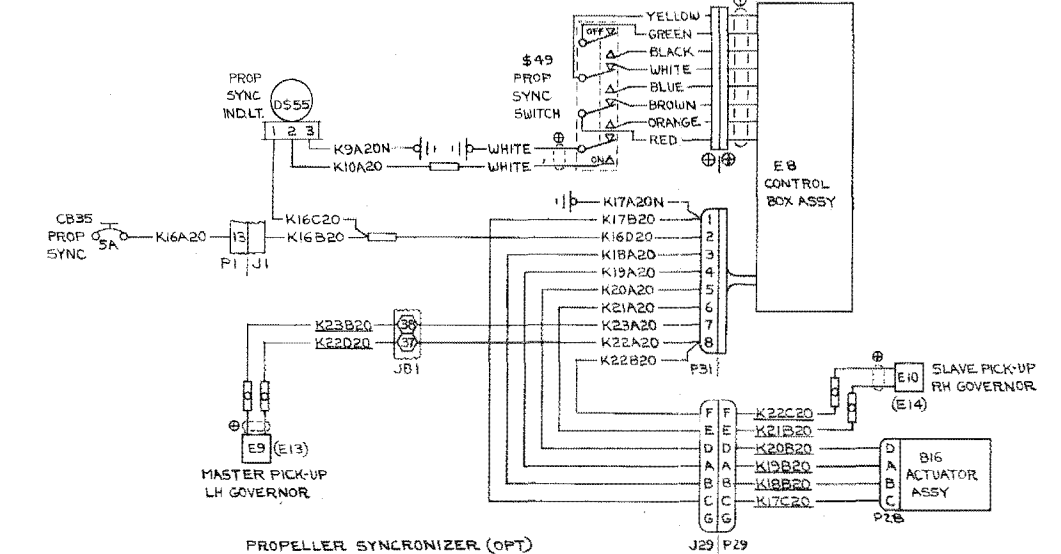
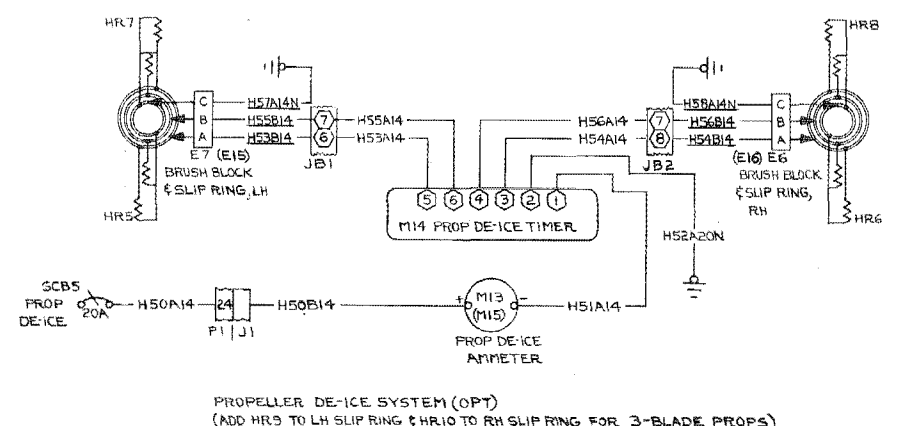
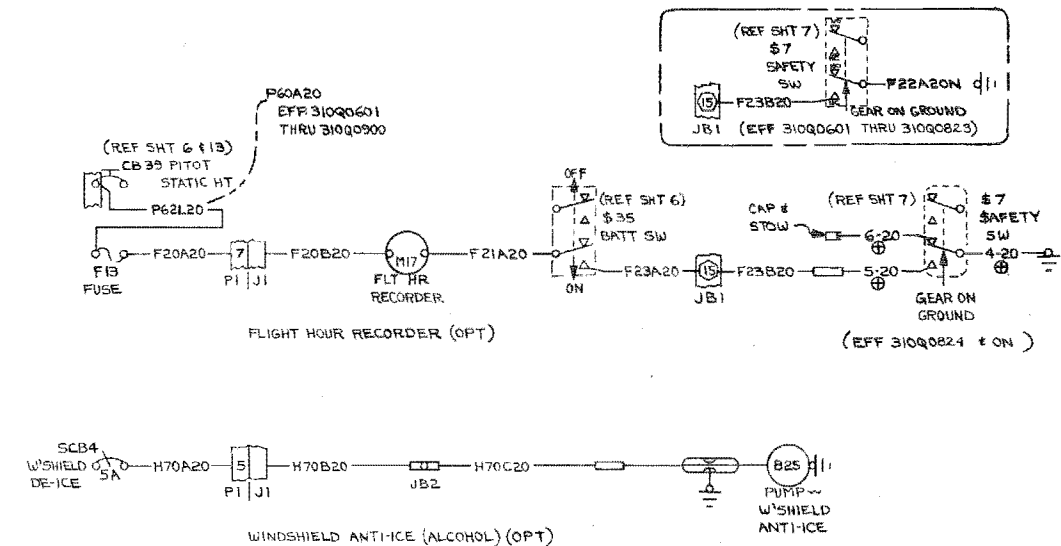
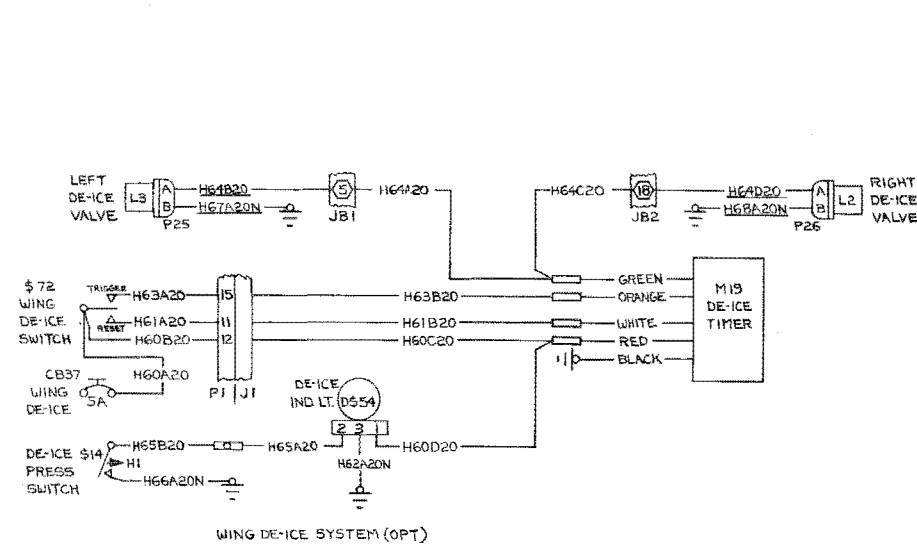
* DENOTES OPTIONAL CIRCUIT WIRES

		AIRCRAFT CO.	P. O. BOX 1877 WICHITA, KANSAS 67201
TITLE: TERMINAL BOARDS (JUNCTION BOX CONNECTIONS)			
SIZE D	CODE IDENT. NO. 71379	DRAWING NO. 0808080	
SCALE NONE	REF F	SHEET 4 OF 16	



NOTE: 300 OR 500 NUMBERS PRECEDING AIRCRAFT WIRE NUMBERS ARE OPTIONAL (SEE NOTE ON PAGE 14-101.)

		AIRCRAFT CO. P. O. BOX 1077 MILITARY & TWIN DIVISION WICHITA, KANSAS 67201	
TITLE SPEAKER, PHONE JACKS, MIKE JACKS, BOOM MIKES & ANGLE OF ATTACK			
SIZE D	CODE IDENT. NO. 71379	DRAWING NO. 0808080	
SCALE NONE	REF G	SHEET 15 OF 16	



NOTE: 300 OR 500 NUMBERS PRECEDING AIRCRAFT WIRE NUMBERS ARE OPTIONAL (SEE NOTE ON PAGE 14-101.)

Cessna		AIRCRAFT CO.		P. O. BOX 1977 MILITARY & TWIN DIVISION WICHITA, KANSAS 67201	
TITLE: FLAP CONTROL, PROP SYNC, PROP DE-ICE, WING DE-ICE, W'SHIELD ANTI-ICE, FLT. HR. RCDR.					
SIZE	CODE IDENT. NO.	DRAWING NO.			
D	71379	0808080			
SCALE	NONE	REF	G	SHEET 14 OF 16	

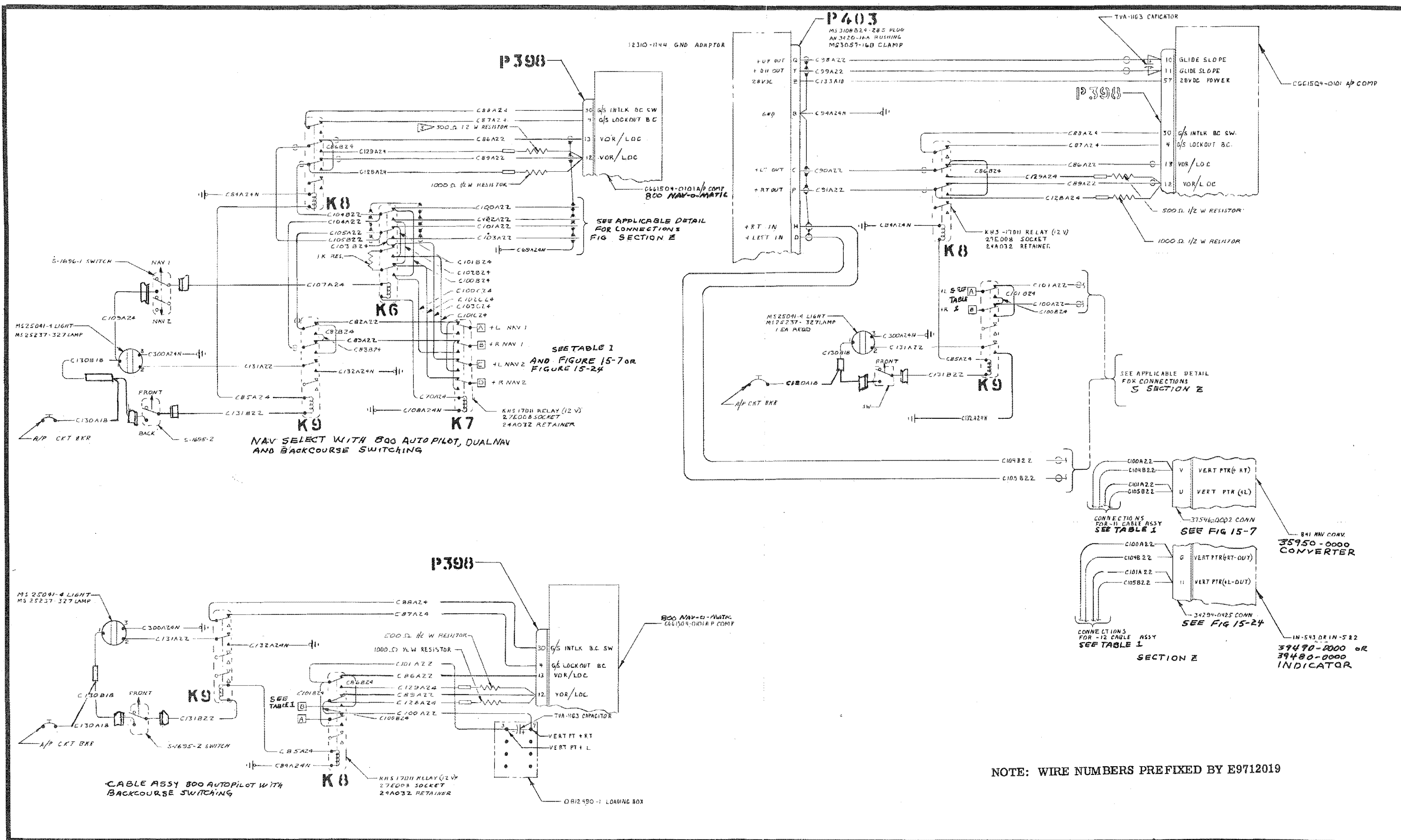


Figure 15-22A. Cessna 800 Nav-O-Matic with Nav Selection and Backcourse Switching (Sheet 2 of 2)

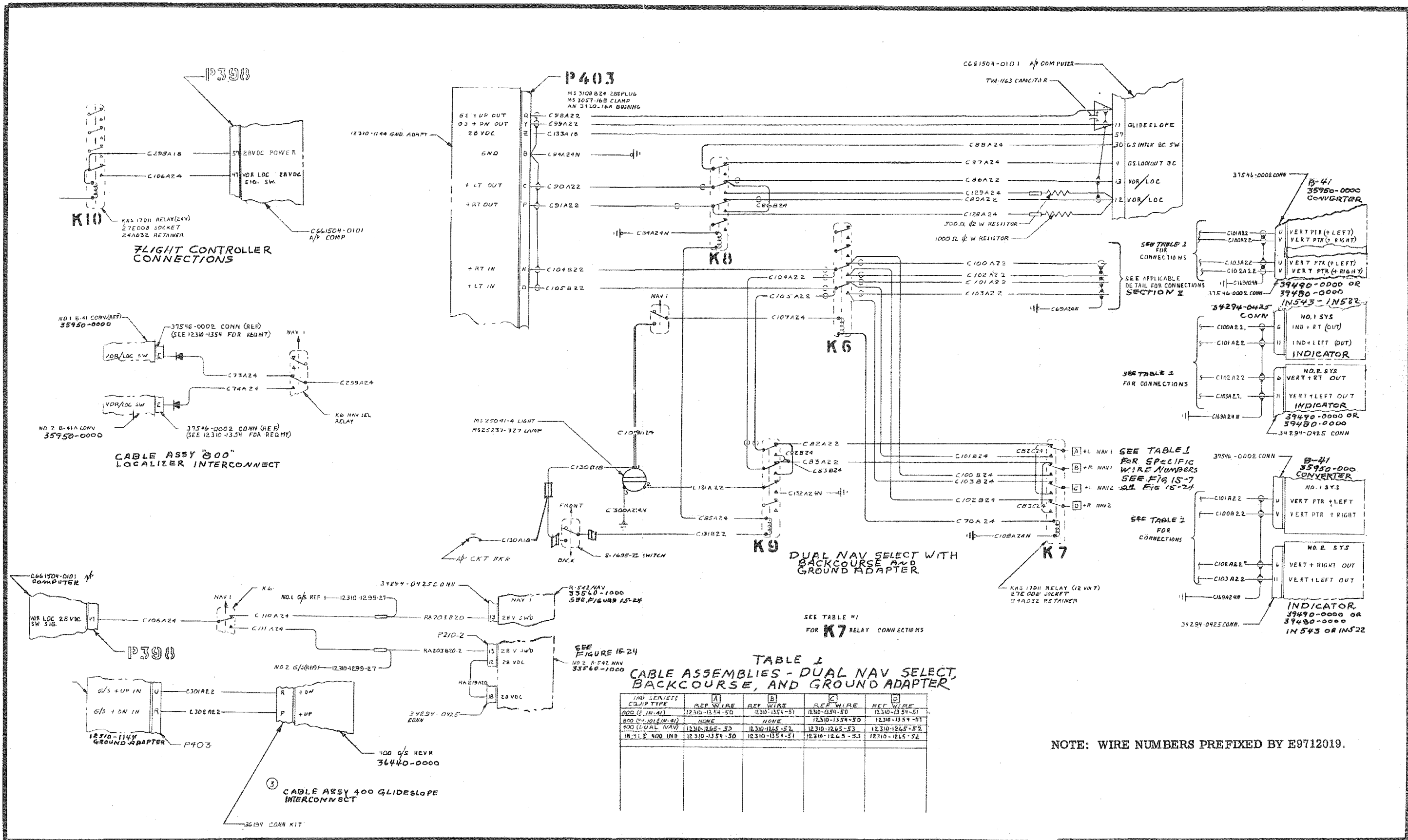
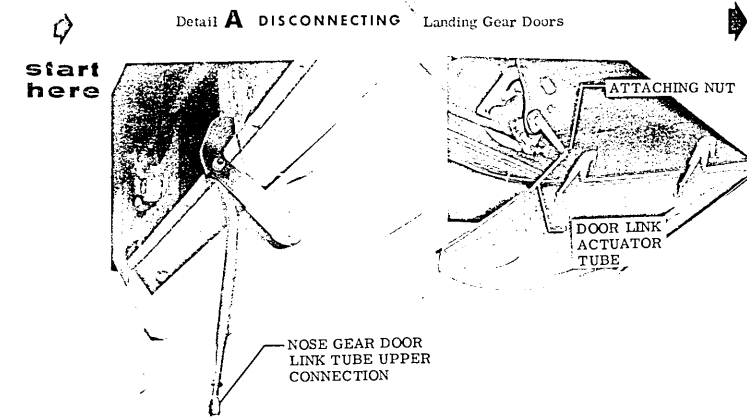


TABLE 1
CABLE ASSEMBLIES - DUAL NAV SELECT,
BACKCOURSE, AND GROUND ADAPTER

IND SERIES/COIL TYPE	A WIRE REF WIRE	B WIRE REF WIRE	C WIRE REF WIRE	D WIRE REF WIRE
800 (2 IN-4)	12310-1254-50	12310-1354-51	12310-1265-50	12310-1354-51
800 (1/2 IN-4)	NONE	NONE	12310-1354-50	12310-1354-51
400 (DUAL NAV)	12310-1265-52	12310-1265-52	12310-1265-53	12310-1265-52
18-11, 8 400 IND	12310-1354-50	12310-1354-51	12310-1265-53	12310-1265-52

NOTE: WIRE NUMBERS PREFIXED BY E9712019.

Figure 15-22A. Cessna 800 Nav-O-Matic with Nav Selection and Backcourse Switching (Sheet 1 of 2)



1. Disconnect nose and main landing gear doors.

CAUTION

When disconnecting the landing gear doors, always run the landing gear up approximately 20 to 30 degrees and disconnect main gear door by removing attaching nut from actuator arm. On the nose gear doors always disconnect the door link tube from the upper connection to prevent the possibility of connecting lower connector to the wrong side of the bellcrank.

---READ THIS--- BEFORE STARTING INSPECTION

NOSE AND MAIN LANDING GEAR RIGGING INSPECTION.

The nose and main landing gear rigging inspection should be performed indoors with proper jacks, 28-volt power supply, 0 to 150 pound spring scale and an 0880001 actuator arm tension tool available.

The aircraft should be placed on jacks, the necessary access plates, seats, cabin divider, carpets, floorboards removed, and the landing gears cleaned with a suitable solvent prior to inspection. Step by step procedures are presented and each step must be completed before performing the next step.

NOTE

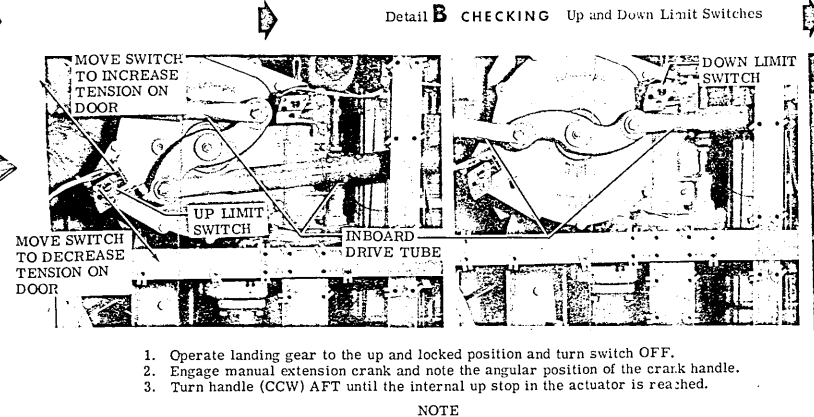
The operational checks and tension measurements requirements of this inspection will require the services of two people.

CAUTION

- When operating the landing gear always be prepared to stop to prevent damage to the system.
- After removal of retraction linkage, assist springs or component parts for checking, they must be reinstalled before proceeding to the next step.

The Landing Gear Rigging Inspection is given in alphabetical and/or alphanumerical sequence. The alphabetical details are the items to be checked. The alphanumerical detail is the related adjustment performed only when adjustment is necessary. The following table lists the details and the related adjustment to be checked.

Check	Adjustment If Required
Detail A	---
Detail B	---
Detail C	---
Detail D	D-1
Detail E	E-1
Detail F	---
Detail G	G-1
Detail H	H-1
Detail J	J-1
Detail K	K-1
Detail L	L-1
Detail M	---
Detail N	N-1 - N-2
Detail P	---
Detail Q	Q-1



1. Operate landing gear to the up and locked position and turn switch OFF.

2. Engage manual extension crank and note the angular position of the crank handle.

3. Turn handle (CCW) AFT until the internal stop is reached.

NOTE

The internal stop should be reached in approximately 3/4 to 1-1/2 turns.

4. If the internal stop is not reached in 3/4 to 1-1/2 turns, run landing gear down and disconnect inboard main drive tube (D-1). Adjust up limit switch until the correct number of turns are obtained.

NOTE

Each time the actuator switches are adjusted the landing gear must be operated approximately half way down then back up before noting the number of turns required to reach the internal stop.

5. Operate the landing gear to the down and locked position.

6. Engage manual extension crank and note the angular position of the crank handle.

7. Turn handle (CW) FWD until the internal stop is reached.

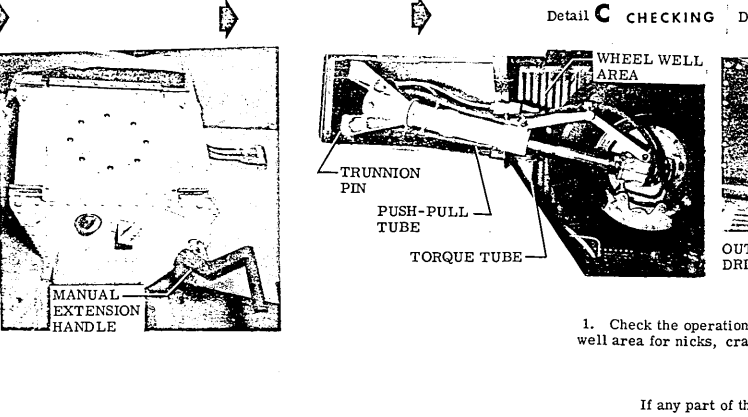
NOTE

The internal stop should be reached in approximately 1 to 2 turns.

8. If the internal stop is not reached in 1 to 2 turns the landing gear actuator down limit switch must be adjusted until the proper number of turns are obtained.

CAUTION

During manual extension of the landing gear, never release the manual extension crank, damage could result to personnel and the skirt of the pilot's seat.



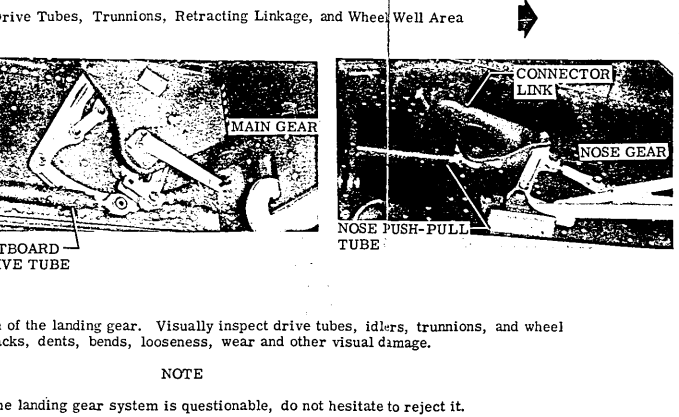
1. Check the operation of the landing gear. Visually inspect drive tubes, idlers, trunnions, and wheel well area for sticks, cracks, dents, bends, looseness, wear and other visual damage.

NOTE

If any part of the landing gear system is questionable, do not hesitate to reject it.

2. Inspect the landing gear retraction linkage for excessive wear, looseness, dents, cracks, bends, and deep scratches.

3. Check main gear trunnion roll pin in pivot shaft for looseness.



1. Check main gear doors for tension as shown 25 ± 10 pounds.

2. Shortening rod end will increase door tension (CCW).

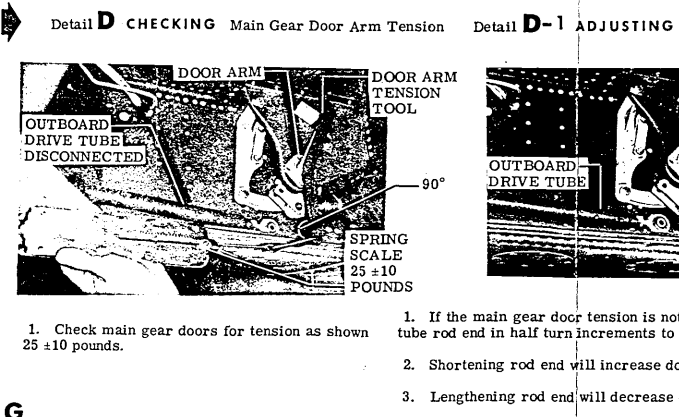
3. Lengthening rod end will decrease door tension.

4. After checking door tension with the gear down, run the landing gear up and check the door tension.

5. The tension should be 25 ± 10 pounds in the up position and a maximum of 10 pounds difference from the down position. Adjust up limit switch (see Detail B) to obtain proper tension.

NOTE

Check hand crank for number of turns to internal stop after obtaining tension, 3/4 to 1-1/2 turns.



1. Check main gear door free fall by breaking main lock links as shown, raise gear approximately six (6) inches by hand, then release.

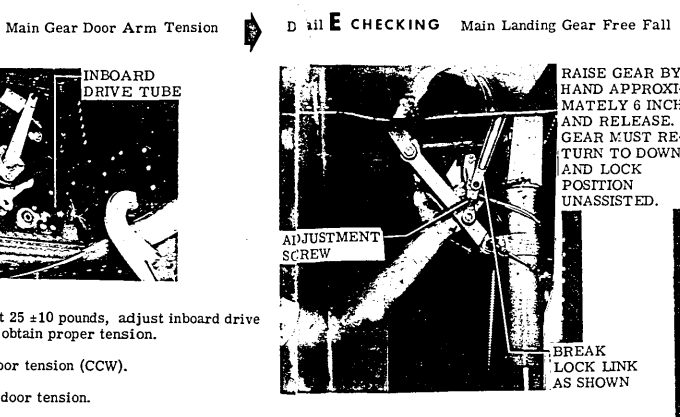
2. If the gear does free fall down and locked, disconnect end fitting and lengthen 1/2 turn, reconnect and check free fall.

NOTE

When checking main gear for free fall the outboard drive tube must be disconnected as shown in Detail D.

3. If the gear does not free fall to a down and lock position, visually check the following:

- Drive tubes for bends, breaks, binding and damage.
- Trunnion bolts for seizing, binding, alignment and lubrication.
- Lock link brace for alignment, overcenter travel, bending and breaks.
- Side braces for proper overcenter engagement, bolts for proper torque, (refer to Torque Chart, Section 1).
- Bolts in wheel well area for binding and interference.



1. Check main gear free fall by breaking main lock links as shown, raise gear approximately six (6) inches by hand, then release.

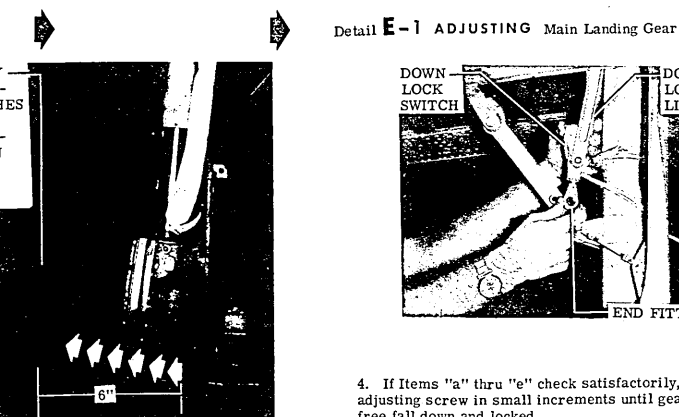
2. If the gear does free fall down and locked, disconnect end fitting and lengthen 1/2 turn, reconnect and check free fall.

NOTE

When checking main gear for free fall the outboard drive tube must be disconnected as shown in Detail D.

3. If the gear does not free fall to a down and lock position, visually check the following:

- Drive tubes for bends, breaks, binding and damage.
- Trunnion bolts for seizing, binding, alignment and lubrication.
- Lock link brace for alignment, overcenter travel, bending and breaks.
- Side braces for proper overcenter engagement, bolts for proper torque, (refer to Torque Chart, Section 1).
- Bolts in wheel well area for binding and interference.



1. Check main gear free fall by breaking main lock links as shown, raise gear approximately six (6) inches by hand, then release.

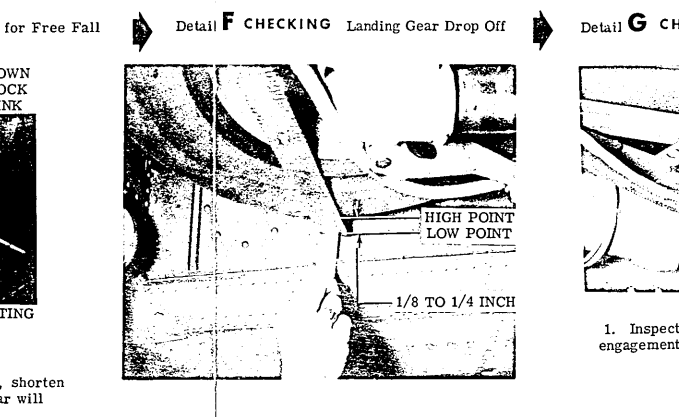
2. If the gear does free fall down and locked, disconnect end fitting and lengthen 1/2 turn, reconnect and check free fall.

NOTE

When checking main gear for free fall the outboard drive tube must be disconnected as shown in Detail D.

3. If the gear does not free fall to a down and lock position, visually check the following:

- Drive tubes for bends, breaks, binding and damage.
- Trunnion bolts for seizing, binding, alignment and lubrication.
- Lock link brace for alignment, overcenter travel, bending and breaks.
- Side braces for proper overcenter engagement, bolts for proper torque, (refer to Torque Chart, Section 1).
- Bolts in wheel well area for binding and interference.



1. (See Detail G-1.) Disconnect uplock push-pull tube.

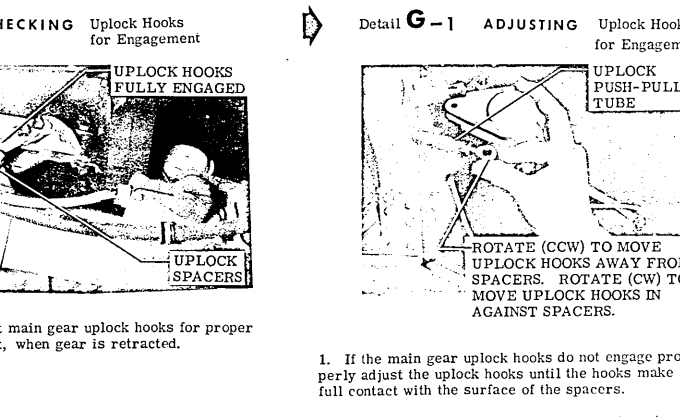
2. Operate landing gear up and measure drop off as shown.

3. If drop off is not 1/8 to 1/4 inches refer to Detail D-1 and adjust outboard drive tube.

NOTE

Lengthen the outboard drive tube to decrease the amount of drop off. Shorten the outboard drive tube to increase drop off.

4. Reconnect uplock hooks.

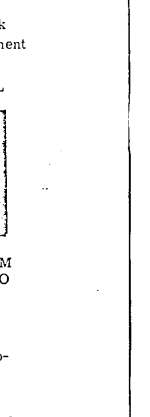


1. Inspect main gear uplock hooks for proper engagement, when gear is retracted.

2. Lengthening the uplock push-pull tube (CCW) will pull the uplock hooks away from the spacers.

3. Shortening the uplock push-pull tube will pull the uplock hooks closer to the spacers.

4. After adjustment the uplock hooks must engage and disengage freely with no binding.

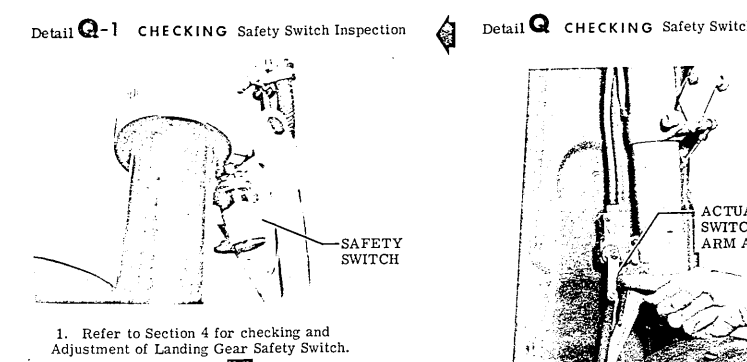
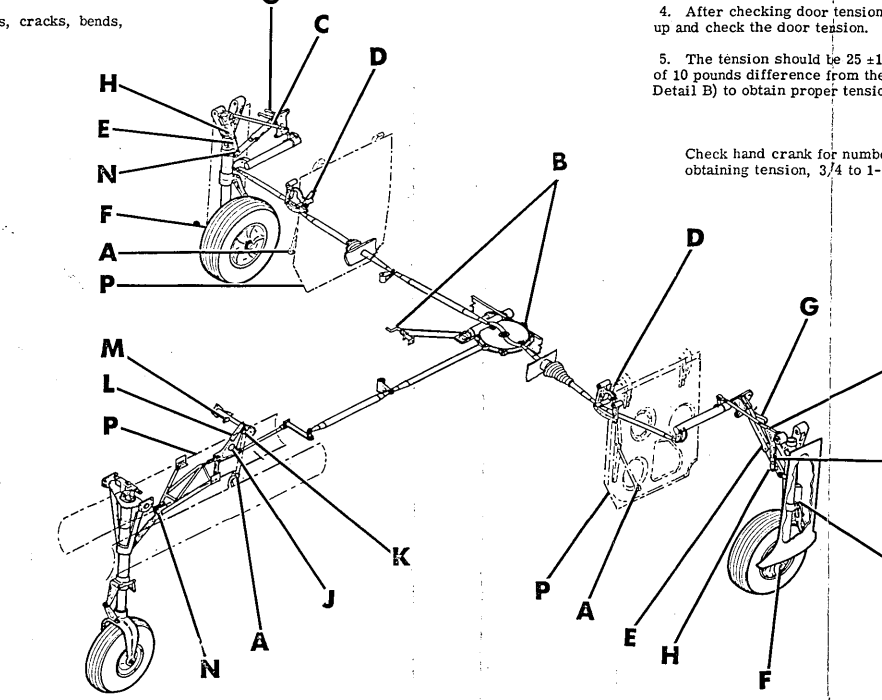


1. If the main gear uplock hooks do not engage properly adjust the uplock hooks until the hooks make full contact with the surface of the spacers.

2. Lengthening the uplock push-pull tube (CCW) will pull the uplock hooks away from the spacers.

3. Shortening the uplock push-pull tube will pull the uplock hooks closer to the spacers.

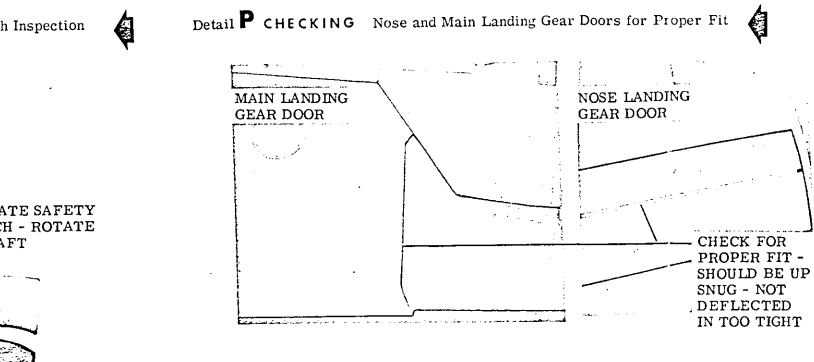
4. After adjustment the uplock hooks must engage and disengage freely with no binding.



1. Refer to Section 4 for checking and Adjustment of Landing Gear Safety Switch.

NOTE

If adjustments were made, make sure all bolts have been torqued, cotter pins installed and doors connected, before installing access plates, floorboards, seats, carpets and cabin divider and removing aircraft from jacks.



1. Turn battery master switch ON.

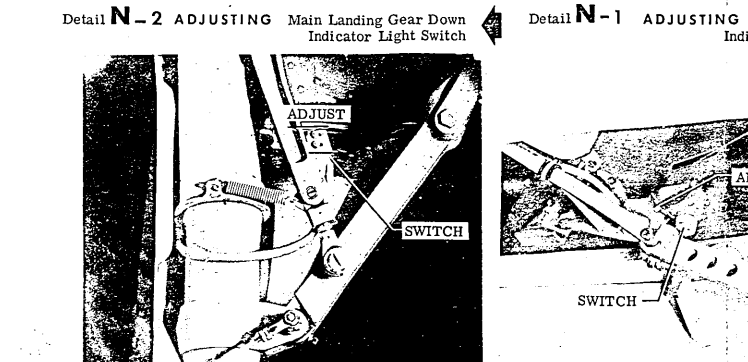
2. Close throttle and actuate landing gear safety switch by hand as shown.

3. Place landing gear switch handle in the UP position. The landing gear should remain down and locked and the horn should sound.

NOTE

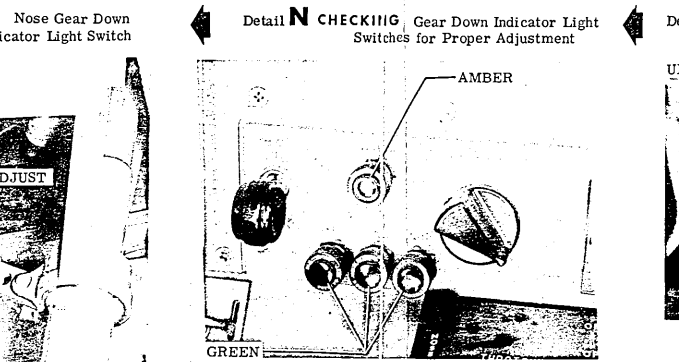
If the horn does not sound, refer to Section 4, Troubleshooting. If the landing gear does not remain down and locked, the safety switch is defective and must be replaced.

4. If the landing gear remains down and locked return landing gear switch handle to DOWN position. Release the safety switch.



1. Operate the landing gear through one complete cycle and visually inspect nose and main landing gear doors for operation, proper fit and other damage.

2. Operate the landing gear through one complete cycle and check the gear indicator lights and warning horn. Check for operation with gear extended and retracted.



1. If the indicator lights do not illuminate within the required number of turns or the overcenter linkage is not overcenter when the lights illuminate make the following adjustments.

2. Adjust the nose gear down and locked indicator light switch by adjusting the switch actuating bolt. Turn bolt CCW to actuate switch farther from the internal stop inside the actuator. Turn bolt CW to actuate switch closer to the internal stop inside the actuator.

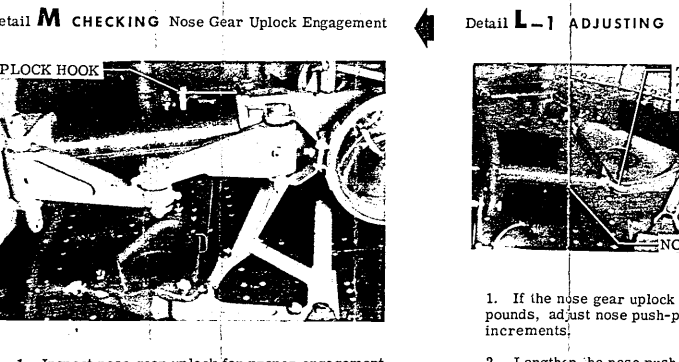
3. Check applicable gear of illuminated light for being down and locked with the overcenter linkage overcenter.

4. Resume cranking toward the down position noting the number of turns required to reach the internal stop in the actuator.

5. The number of turns required to reach the internal stop should not be less than 8 or more than 14 for the nose gear. Not less than 4 or more than 8 on the main gear.

NOTE

After adjusting indicator switches check to see that gear indicator lights do not illuminate before gear is down and locked by overcenter linkage.



1. Retract landing gear approximately halfway.

2. Engage manual extensor handle (see Detail B). Crank toward the down position and stop when green light comes on. Note the angular position of the manual extension handle.

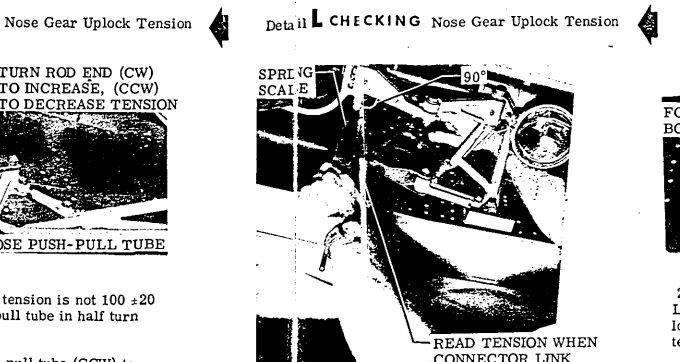
3. Check applicable gear of illuminated light for being down and locked with the overcenter linkage overcenter.

4. Resume cranking toward the down position noting the number of turns required to reach the internal stop in the actuator.

5. The number of turns required to reach the internal stop should not be less than 8 or more than 14 for the nose gear. Not less than 4 or more than 8 on the main gear.

NOTE

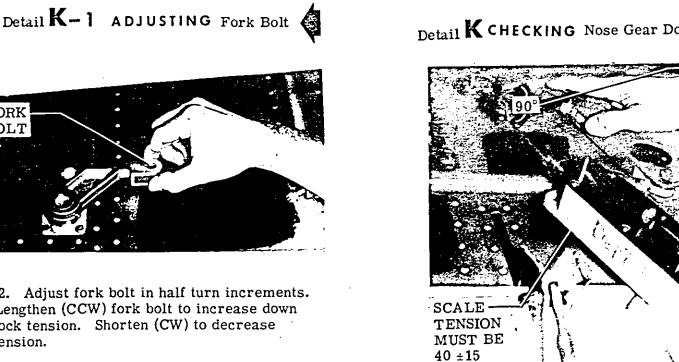
After adjusting indicator switches check to see that gear indicator lights do not illuminate before gear is down and locked by overcenter linkage.



1. Inspect nose gear uplock for proper engagement.

2. Uplock hook must be fully engaged with hook against the nose strut bolt and spacer engages and disengages freely with no binding.

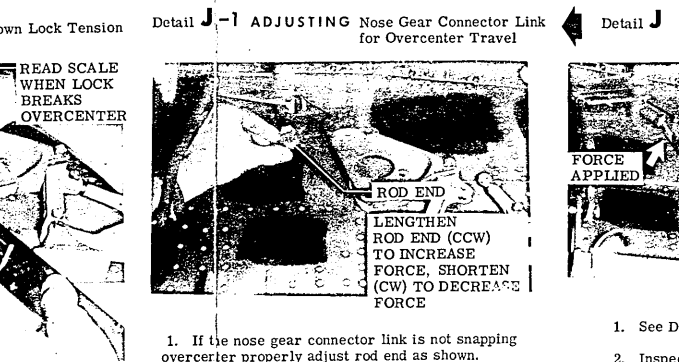
3. Uplock hook must be engaged with .003 to .004 inches clearance between the spacer on the nose strut and surface of the hook.



1. If the nose gear uplock tension is not 100 ± 20 pounds, adjust nose push-pull tube in half turn increments.

2. Lengthen the nose push-pull tube (CCW) to decrease the uplock tension.

3. Shorten nose push-pull tube to increase (CW) uplock tension.



1. Retract landing gear approximately halfway.

2. Engage manual extensor handle (see Detail B). Crank toward the down position and stop when green light comes on. Note the angular position of the manual extension handle.

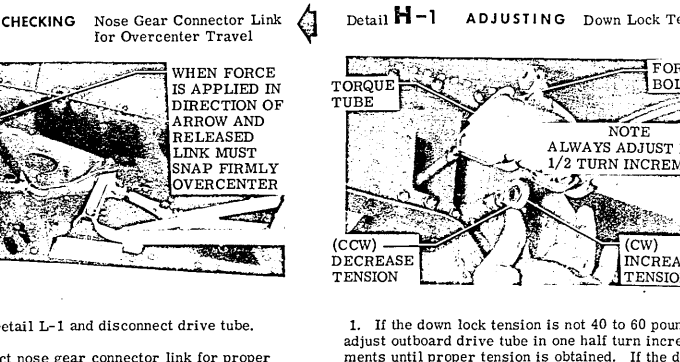
3. Check applicable gear of illuminated light for being down and locked with the overcenter linkage overcenter.

4. Resume cranking toward the down position noting the number of turns required to reach the internal stop in the actuator.

5. The number of turns required to reach the internal stop should not be less than 8 or more than 14 for the nose gear. Not less than 4 or more than 8 on the main gear.

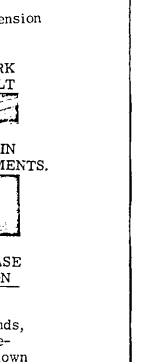
NOTE

After adjusting indicator switches check to see that gear indicator lights do not illuminate before gear is down and locked by overcenter linkage.



1. Inspect nose gear uplock tension (100 ± 20 pounds).

2. Adjust fork bolt in half turn increments. Lengthen (CCW) fork bolt to increase down lock tension. Shorten (CW) to decrease tension.



1. Connect nose push-pull tube. (See Detail L-1.)

2. Inspect nose gear down lock tension (40 ± 15 pounds).

3. If the nose gear down lock tension is not 40 ± 15 pounds, make sure the nose gear fork bolt is properly adjusted (K-1).

1. If the nose gear connector link is not snapping overcenter properly adjust rod end as shown.

2. Inspect nose gear connector link for proper overcenter adjustment.

1. See Detail L-1 and disconnect drive tube.

2. Inspect nose gear connector link for proper overcenter adjustment.

1. If the down lock tension is not 40 to 60 pounds, adjust outboard drive tube in one half turn increments until proper tension is obtained. If the down lock tension is in excess of 60 pounds the down lock tension should be rigged 40 to 60 pounds. Lengthening the outboard drive tube (CCW) decreases and shortening the outboard drive tube (CW) increases tension.

NOTE

When shortening or lengthening the outboard push-pull tube the fork bolt must be lengthened or shortened a corresponding amount of turns so that the combined length of the two parts does not change.

Figure 2-8A. Nose and Main Landing Gear Rigging Inspection

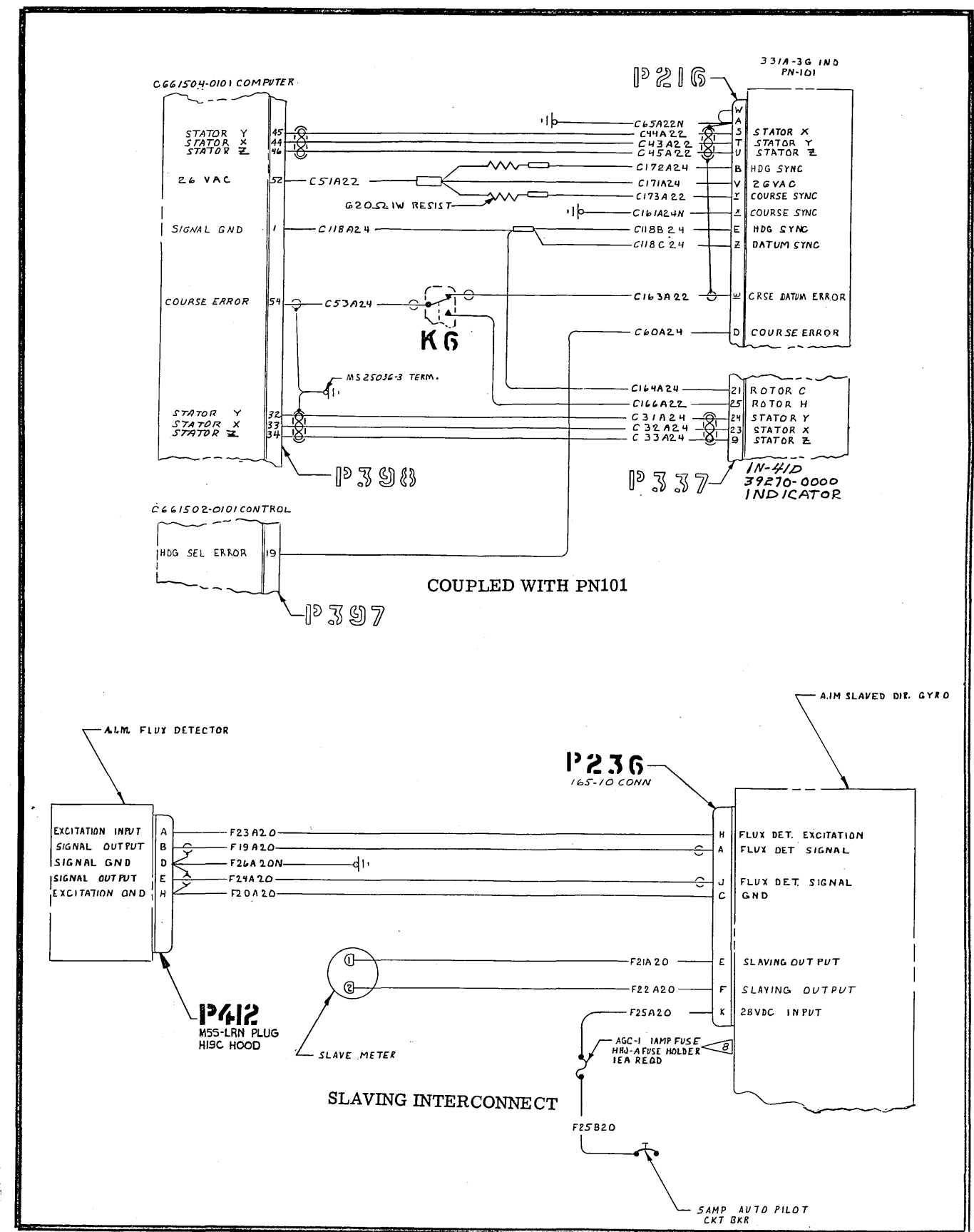


Figure 15-22B. Cessna 800 Nav-O-Matic Autopilot (Sheet 2 of 2)

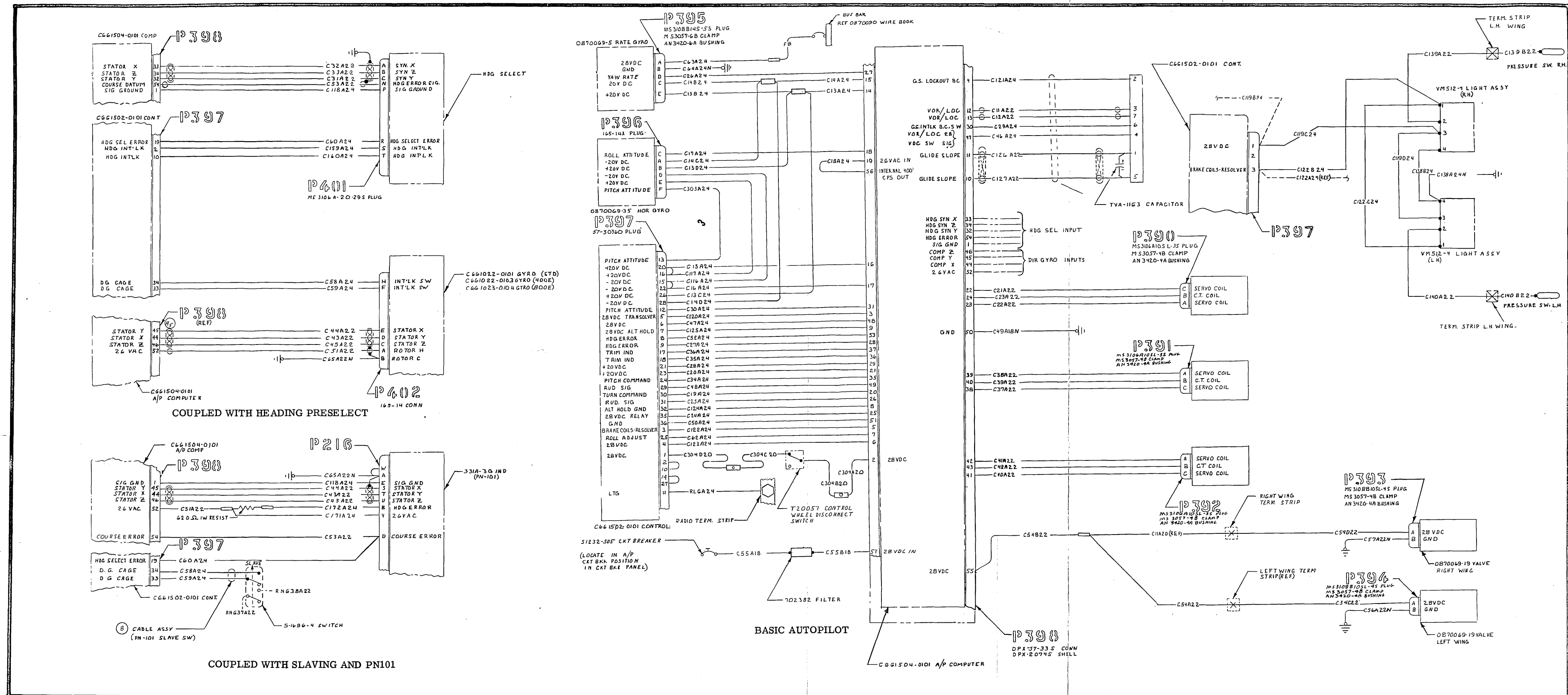
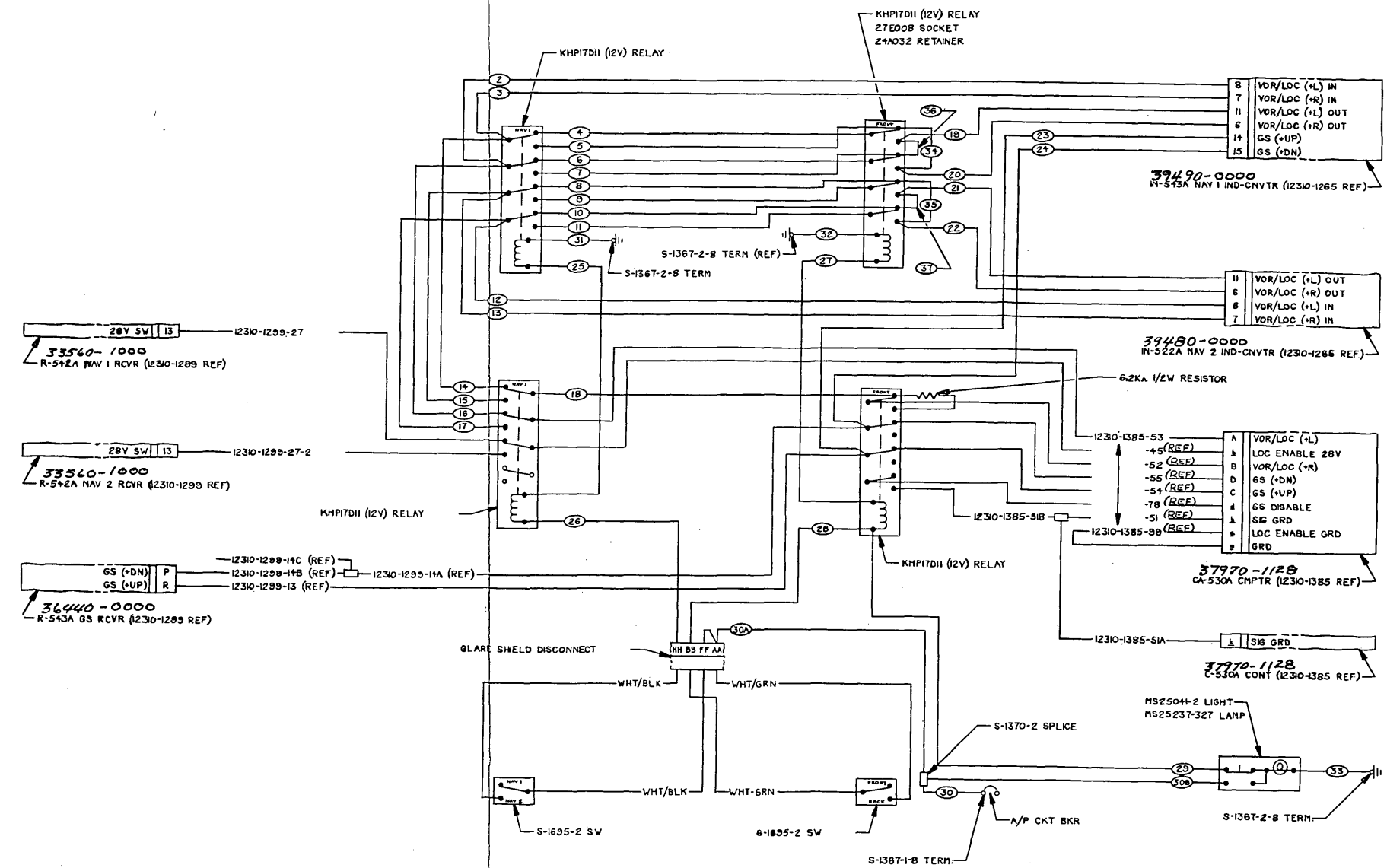


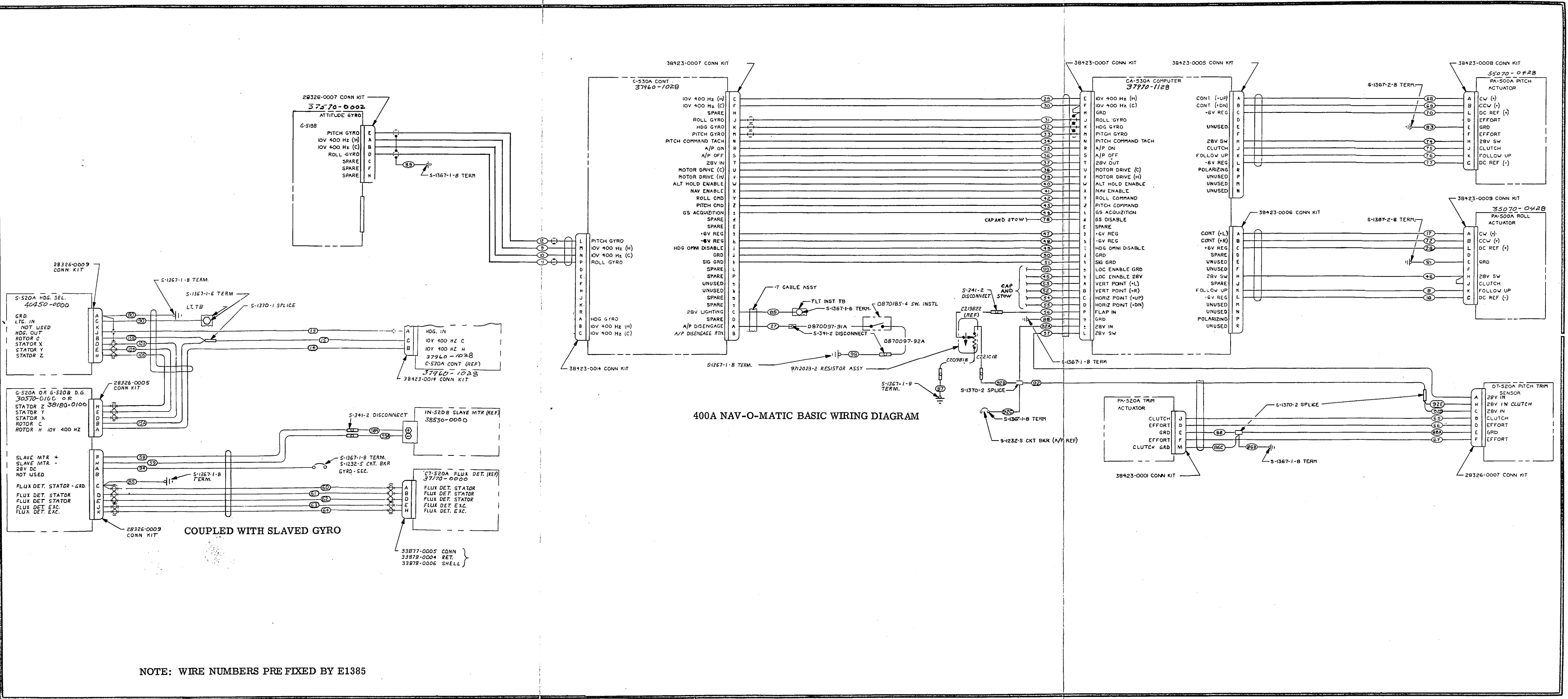
Figure 15-22B. Cessna 800 Nav-O-Matic Autopilot (Sheet 1 of 2)



NOTE: WIRE NUMBERS PREFIXED BY E1386

310Q0001 AND ON

Figure 15-32B. Cessna 400A Nav Select with Backcourse Switching



400A NAV-O-MATIC BASIC WIRING DIAGRAM

NOTE: WIRE NUMBERS PRE FIXED BY E1385

Figure 15-32A. Cessna 400A Nav-O-Matic Autopilot